

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

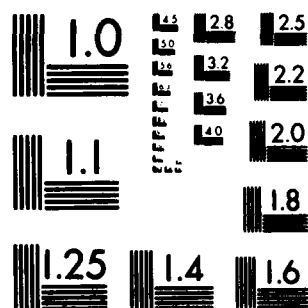
AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

AD-A110 673 BOM CORP MONTEREY CA F/O 20/4  
PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (W--ETC(U)  
AUG 80 M ZEBNER N00014-78-C-0204  
UNCLASSIFIED NPS-67-80-001CR NL

1 of 2  
2000-0000

1 of 2  
2000-0000



MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A.

AD A110673

LEVEL II

②

NPS67-80-001CR

# NAVAL POSTGRADUATE SCHOOL

Monterey, California



DTIC  
ELECTE  
FEB 9 1982  
S E D

## CONTRACTOR REPORT

PROCEDURE AND COMPUTER PROGRAM  
FOR THE APPROXIMATION OF DATA  
(WITH APPLICATION TO MULTIPLE SENSOR PROBES)

H. ZERNER  
BDM CORPORATION  
P.O. Box 2019  
Monterey, CA 93940

August 1980

Interim Report for Period June 1980 - August 1980

Approved for public release; distribution unlimited

Prepared for:  
Naval Postgraduate School  
Monterey, California 93940

Copy available to DTIC does not  
permit fully legible reproduction

88 02 08 186

FILE COPY

NAVAL POSTGRADUATE SCHOOL

Monterey, California

Rear Admiral J. J. Ekelund  
Superintendent

D. A. Schradz  
Acting Provost

The work reported herein was carried out by the BDM Corporation under Work Order 426, Turbopropulsion Laboratory Support, under Contract Number N00014-78-C-0204. The work was part of a program entitled Transonic Compressor Investigations, funded in part by Naval Air Systems Command under Program Element 61153N, Contract Number N00019-80-WR-01199. The program was under the cognizance of Dr. H. J. Mueller, Code 310.

The author is a candidate for the degree D. Ing under Prof. H. Gallus at Rhein,-Westf. Techn. Hochschule, 51 Aachen, W. Germany.

This report was prepared by:

*Hans M. Platen*

BDM Corporation  
P.O. Box 2019  
Monterey, CA 93940

Publication of the report does not constitute approval of the sponsor for the findings or conclusions. It is published for information and for the exchange and stimulation of ideas.

Reviewed by:

*R. P. Shreave*

R. P. SHREAVE, Director  
Turbopropulsion Laboratory

*M. F. Platzer*

M. F. PLATZER, Chairman  
Department of Aeronautics

Released by:

*William M. Tolles*

W. M. TOLLES  
Dean of Research

## **DISCLAIMER NOTICE**

**THIS DOCUMENT IS BEST QUALITY  
PRACTICABLE. THE COPY FURNISHED  
TO DTIC CONTAINED A SIGNIFICANT  
NUMBER OF PAGES WHICH DO NOT  
REPRODUCE LEGIBLY.**

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER NPS67-80-001CR	2. GOVT ACCESSION NO. AD-A110 673	3. RECIPIENT'S CATALOG NUMBER
4. TITLE (and Subtitle) PROCEDURE AND COMPUTER PROGRAM FOR THE APPROXIMATION OF DATA (WITH APPLICATION TO MULTIPLE SENSOR PROBES).		5. TYPE OF REPORT & PERIOD COVERED CONTRACTOR REPORT June 1980 - August 1980
		6. PERFORMING ORG. REPORT NUMBER
7. AUTHOR(s) H. Zebner		8. CONTRACT OR GRANT NUMBER(s) N00014-78-C-0204
9. PERFORMING ORGANIZATION NAME AND ADDRESS BDM CORPORATION P.O. BOX 2019 MONTEREY, CALIFORNIA 93940		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS Work Order 426
11. CONTROLLING OFFICE NAME AND ADDRESS NAVAL POSTGRADUATE SCHOOL MONTEREY, CALIFORNIA 93940		12. REPORT DATE August 1980
		13. NUMBER OF PAGES 186
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office)		15. SECURITY CLASS. (of this report) Unclassified
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report)  Approved for Public Release; distribution unlimited		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES Interim Report. Principal Investigator, Dr. R. P. Shreeve, Director Turbopropulsion Laboratory, Naval Postgraduate School, Monterey, California 93940		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Data Approximation Probe Calibration Flow Measurements		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) A procedure to approximate data given at arbitrary intervals in two-, three- and four dimensions using polynomial expressions is described. The programming of the problem is explained in each case and a user manual is given for software implemented on the TPL Hewlett-Packard computer system. The method, which is general, was derived and has been applied to represent the calibration of flow probes which have multiple sensors.		

DD FORM 1 JAN 73 1473

EDITION OF 1 NOV 65 IS OBSOLETE  
S/N 0102-010-0001

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Data Entered)

4/1/81

## TABLE OF CONTENTS

<u>SECTION</u>		<u>PAGE NO.</u>
ABSTRACT		3
1.	INTRODUCTION	4
2.	TWO DIMENSIONAL APPROXIMATION	5
2.1.	Problem	5
2.2.	Approach	6
2.3.	Solution	7
2.4.	Equation System	9
2.4.1.	System Matrix A	9
2.4.2.	Right Hand Side Vector B	10
2.5.	Software	12
2.6.	Sample User Program	15
2.6.1.	Listing	15
2.6.2.	Load Map	18
2.6.3.	Results	19
3.	THREE DIMENSIONAL APPROXIMATION	21
3.1.	Problem	21
3.2.	Approach	22
3.3.	Solution	24
3.4.	Equation System	27
3.4.1.	System Matrix A	27
3.4.2.	Right Hand Side Vector B	33
3.5.	Software	35
3.6.	Sample User Program	37
3.6.1.	Listing	37
3.6.2.	Load Map	43
3.6.3.	Results	44
4.	FOUR DIMENSIONAL APPROXIMATION	49
4.1.	Problem	49
4.2.	Approach	50
4.3.	Solution	53
4.4.	Structure of Equation System	67
4.4.1.	System Matrix A	67
4.4.2.	Right Hand Side Vector B	76
4.5.	Software	79
4.6.	Sample User Program	81
4.6.1.	Listing	81
4.6.2.	Load Map	83
4.6.3.	Results	84

# TABLE OF CONTENTS (Cont'd)

5.	CONCLUSIONS AND RECOMMENDATIONS	87
6.	LIST OF REFERENCES	88
Appendix A	SOME USEFUL MATRIX CONVENTIONS AND OPERATIONS	89
	A1 Submatrix Notation	89
	A2 Diagonal Lines and Bands 1. Order	89
	A3 Diagonal Lines and Bands 2. Order	90
Appendix B	SOFTWARE DESCRIPTION: FLOW CHARTS	92
Appendix C	SOFTWARE DESCRIPTION: LISTINGS	130
	DISTRIBUTION LIST	180



# ABSTRACT

A procedure to approximate data given at arbitrary intervals in two-, three- and four dimensions using polynomial expressions is described. The programming of the problem is explained in each case and a user manual is given for software implemented on the TPL Hewlett-Packard computer system. The method, which is general, was derived and has been applied to represent the calibration of flow probes which have multiple sensors.

Accession For	
NTIS GRA&I	<input checked="" type="checkbox"/>
DTIC TAB	<input type="checkbox"/>
Unannounced	<input type="checkbox"/>
Justification	
By	
Distribution/	
Availability Codes	
Dist	Avail and/or Special
A	23 CP



## 1. INTRODUCTION

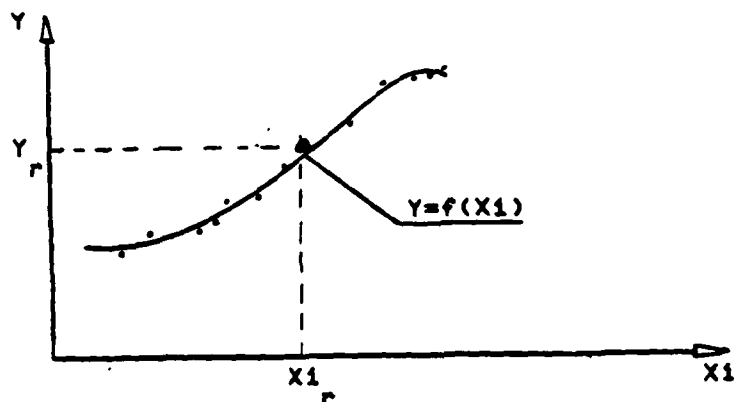
This report describes algorithms to approximate data patterns such as those which occur, for instance, when pneumatic-velocity probes are calibrated. The functional value ( $Y$ ) can depend on either one ( $X_1$ ), two ( $X_1$  and  $X_2$ ) or even three ( $X_1$ ,  $X_2$  and  $X_3$ ) parameters. The approximation  $Y = f(X_1)$  will be referred to as the two dimensional, the approximation  $Y = f(X_1, X_2)$  as the three dimensional and the approximation  $Y = f(X_1, X_2, X_3)$  as the four dimensional approximation. These three options meet the requirements for probes used in the Turbopropulsion Laboratory.

Using the least-squares criterion to obtain the coefficients in assumed polynomials leads to a system of linear equations, which are, in principle easily solved. Numerical problems may arise however, since the software, as described herein and as implemented in the NPS Turbopropulsion Laboratory Hewlett Packard 21MX computer, uses 32-bit real constants.

The author wishes to express his thanks to Professor Ray Shreeve, whose constant and critical interest was a crucial help to solve this mathematical problem and link it to engineering application.

## 2. TWO DIMENSIONAL APPROXIMATION

### 2.1. Problem:



A data set of NPNTS1 data points is given, where  $Y$  depends on parameter  $X_1$ . The data pattern is to be approximated by a function  $Y = f(X_1)$ , so that the error between data points and analytically determined points is lowest.

## 2.2. Approach:

Polynomials are commonly used in order to approximate data patterns.  $Y = f(X_1)$  is a function which approximates the data value,  $V$ , at each value of the variable  $X_1$ , we look for our expression of the form

$$Y = C_1 + C_2 * X_1 + C_3 * X_1^2 + \dots + C_L * X_1^{(L-1)}$$

or

$$Y = \sum_{i=1}^L C_i * X_1^{(i-1)} \quad (2.1.)$$

in which the coefficients  $C_i$  are to be determined by the method of least squares. As Ref. 1 shows the least squares criterion leads to a linear equation system that can easily be solved. We define the error

$$R = \sum_{r=1}^{NPNTS1} [f(X_{1r}) - Y_r]^2 \quad (2.2.)$$

where the index  $r$  denotes the individual data point. Using equation (2.1.)  $R$  becomes

$$R = \sum_{r=1}^{NPNTS1} [C_1 + C_2 * X_{1r} + C_3 * X_{1r}^2 + \dots + C_L * X_{1r}^{(L-1)} - Y_r]^2$$

### 2.3. Solution:

R depends on the selection of the coefficients  $C_1, \dots, C_L$ .

In order to minimize the error, R is differentiated with respect to  $C_1, \dots, C_L$  and the partial derivatives are set to zero. Thus

$$\frac{\partial R}{\partial C_i} = 0 \quad i = 1, \dots, L$$

or

$$\frac{\partial R}{\partial C_i} = \sum_{r=1}^{NPNTS1} 2[C_1 + C_2 * X_{1r} + C_3 * X_{1r}^2 + \dots + C_L * X_{1r}^{(L-1)} - Y_r] * \frac{\partial [C_1 + C_2 * X_{1r} + C_3 * X_{1r}^2 + \dots + C_L * X_{1r}^{(L-1)} - Y_r]}{\partial C_i} = 0$$

Assuming that the summations extend over all the data points,  $\sum$  should be understood to mean  $\sum_{r=1}^{NPNTS1}$ . Performing the differentiations and rearranging the equations, we get

$$\begin{aligned} NPNTS1 * C_1 + \sum X_{1r} * C_2 + \sum X_{1r}^2 * C_3 + \dots + \sum X_{1r}^{(L-1)} * C_L &= \sum Y_r \\ \sum X_{1r} * C_1 + \sum X_{1r}^2 * C_2 + \sum X_{1r}^3 * C_3 + \dots + \sum X_{1r}^L * C_L &= \sum Y_r * X_{1r} \\ \sum X_{1r}^2 * C_1 + \sum X_{1r}^3 * C_2 + \sum X_{1r}^4 * C_3 + \dots + \sum X_{1r}^{(L+1)} * C_L &= \sum Y_r * X_{1r}^2 \\ \dots\dots\dots \\ \sum X_{1r}^{(L-1)} * C_1 + \sum X_{1r}^L * C_2 + \sum X_{1r}^{(L+1)} * C_3 + \dots + \sum X_{1r}^{(2L-2)} * C_L &= \sum Y_r * X_{1r}^{L-1} \end{aligned}$$

In matrix notation

$$\begin{pmatrix}
 \text{NPNTS1} & \Sigma x_1^1 & \Sigma x_1^2 & \dots & \Sigma x_1^{L-1} \\
 \Sigma x_1^1 & \Sigma x_1^2 & \Sigma x_1^3 & \dots & \Sigma x_1^L \\
 \Sigma x_1^2 & \Sigma x_1^3 & \Sigma x_1^4 & \dots & \Sigma x_1^{L+1} \\
 \dots & \dots & \dots & \dots & \dots \\
 \Sigma x_1^{L-1} & \Sigma x_1^L & \Sigma x_1^{L+1} & \dots & \Sigma x_1^{2L-2}
 \end{pmatrix}
 \begin{pmatrix}
 c_1 \\
 c_2 \\
 c_3 \\
 \vdots \\
 c_L
 \end{pmatrix}
 =
 \begin{pmatrix}
 \Sigma Y_r \\
 \Sigma Y_r * x_1^1 \\
 \Sigma Y_r * x_1^2 \\
 \vdots \\
 \Sigma Y_r * x_1^{L-1}
 \end{pmatrix}
 \quad (2.3.)$$

$\mathbf{A} \quad \mathbf{C} = \mathbf{E}$

- $\mathbf{A}$  ... System Matrix
- $\mathbf{C}$  ... Coefficients vector
- $\mathbf{E}$  ... Right hand side vector

## 2.4. Equation System

### 2.4.1. System Matrix A

Since all Matrix elements on diagonal lines running up from left to right (2. order, as defined in Appendix A) are identical, the elements are renamed for simplification as follows:

$$a_1 = a_{11}$$

$$a_2 = a_{21} = a_{12}$$

$$a_3 = a_{31} = a_{22} = a_{13}$$

...

$$a_L = a_{L1} = a_{L-1,2} = a_{L-2,3} = \dots = a_{2,L-1} = a_{1,L}$$

$$a_{L+1} = a_{L,2} = a_{L-1,3} = \dots = a_{3,L-1} = a_{2,L}$$

$$a_{L+2} = a_{L,3} = \dots = a_{1,L-1} = a_{3,L}$$

$$a_{2L-1} = a_{L,L}$$

so that the set of elements,  $a_k$ , can be written as

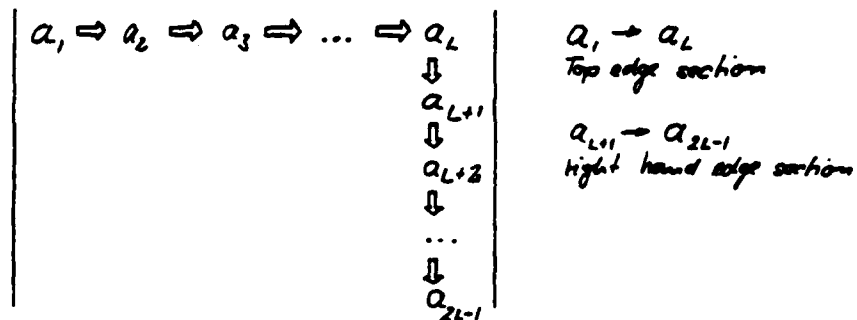
$$a_k = \sum_{r=1}^{NPNTS1} X1_r^{k-1} \quad k=1, \dots, 2L-1 \quad (2.4.)$$

This operation is programmed in REAL FUNCTION S2 (NPNTS1, IPOWR1, IY).

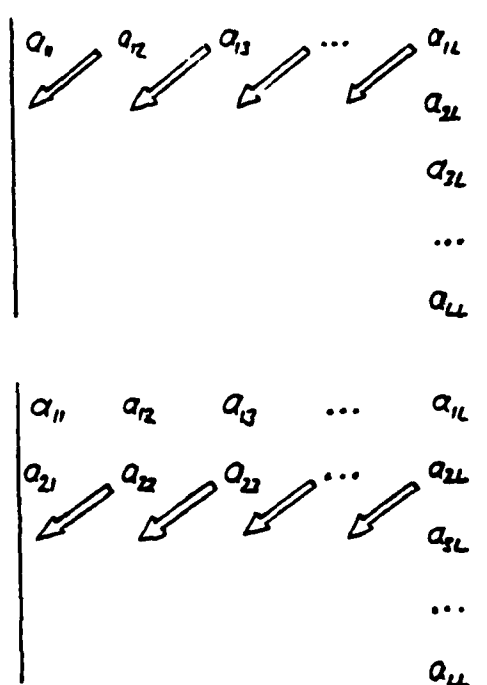
The data X1 and Y are known to this function through a common block named DTA2. So only NPNTS1, IPOWR1 (= k-1) and IY (=0) have to be passed to the function, and the value of  $a_k$  is returned through the function name S2.

SUBROUTINE MAT2 presets the system matrix in the following way:

- i) Preset edge section elements (using REAL FUNCTION S2)



- ii) Copy defined elements diagonally





$$\begin{array}{ccccc}
 a_{11} & a_{12} & a_{13} & \dots & a_{1L} \\
 a_{21} & a_{22} & a_{23} & \dots & a_{2L} \\
 a_{31} & a_{32} & a_{33} & \dots & a_{3L} \\
 & & & & \dots \\
 & & & & a_{4L}
 \end{array}$$

$$\begin{array}{ccccc}
 a_{11} & a_{12} & a_{13} & \dots & a_{1L} \\
 a_{21} & a_{22} & a_{23} & \dots & a_{2L} \\
 a_{31} & a_{32} & a_{33} & \dots & a_{3L} \\
 \dots & \dots & \dots & \dots & \dots
 \end{array}$$

$$A = \begin{array}{ccccc}
 a_{11} & a_{12} & a_{13} & \dots & a_{1L} \\
 a_{21} & a_{22} & a_{23} & \dots & a_{2L} \\
 a_{31} & a_{32} & a_{33} & \dots & a_{3L} \\
 \dots & \dots & \dots & \dots & \dots \\
 a_{41} & a_{42} & a_{43} & \dots & a_{4L}
 \end{array}$$

#### 2.4.2. Right hand side vector B

The elements of the right hand side vector B can be written as

$$b_k = \sum_{r=1}^{NPNTS1} (Y_r \cdot X_r^{k-1}) \quad k = 1, \dots, L \quad (2.5.)$$

To calculate  $b_k$ , again REAL FUNCTION S2 is used. Data X1 and Y are available through COMMON block DTA2. NPNTS1, IPOWR1 (=k-1) and IY (=1) have to be passed to the function, and the value of  $b_k$  is returned through the function name S2.

## 2.5. Software:

The software to compute the coefficients for a two dimensional approximation is described in Appendix B and is implemented in the TPL HP-21MX computer system

To work correctly with these program modules, the user must conform to the following conventions:

- 1) Provide the data in two arrays (Type: REAL) of 256 elements through a COMMON block, named DTA2.

```
COMMON / DTA2 / X1,Y  
REAL X1(256),Y(256)
```

- ii) Dimension an array (Type: REAL) of 7 (seven) elements to contain the coefficients.

```
REAL COEF(7)
```

- iii) Define the parameters NPNTS1, L and IPRINT (Type: all INTEGER)

```
NPNTS1    ... # of data points  
           L ≤ NPNTS1 ≤ 256
```

```
L         ... (desired order of polynomial) + 1  
           1 ≤ L ≤ 7
```

```
IPRINT    ... controls quantity of print out  
           2    ... Print system matrix and right  
                   hand side vector before and  
                   after Gauss Jordan Elimination  
           1    ... Print equation system after  
                   Gauss Jordan Elimination  
           <0   ... No print out  
           >0   ... Print equation (1,1) with the  
                   actual parameters
```

- iv) When loading a program, that uses the subroutine MAT2, the binary library file has to be searched for externals.

Suppose, the source file &USER::26 contains a user program named USER. This program calls MAT2. After the compilation, the relocatable binary file for this user program is ZUSER::26. To load the program, the following procedure is recommended:

Type (from FMGR)

:RU,LOADR

and the loader program will respond

/LOADR: RE,ZUSER::26

where the underlined information already is the user's input. This causes the loader to load all program modules of the user program; a load map is listed on the terminal. Upon completion the loader prompts

/LOADR: MS,ZTPLBL::26

where the underlined information already is the user's input. Now the loader conducts a search for all unsatisfied externals of the user program in the binary library file ZTPLBL::26. Since some library programs have externals themselves the search has to be repeated (MS ... multiple search) as many time as is necessary for all externals to be satisfied. The loader prompts

/LOADR: END

where the underlined information is the user's input. The loader now loads all system programs, outputs a load map and generates the program. Upon completion, the loader outputs a ready message

/LOADR:USER READY AT 1:28 PM WED., 10 SEPT, 1980

/LOADR: END

Now the program can be run

:RU,USER

Again, the underlined information is the user's input.

If all these requirements are met, the correct call for the sub-routine is:

CALL MAT2 (NPNTS1,L,COEF,IPRINT)

Upon completion, the array COEF contains the coefficients (COEF(1)=C<sub>1</sub>, ..., COEF(L)=C<sub>L</sub>). Externals used by MAT2 are: AB2, DTA2, S2 under no circumstances may the user use any of these names for modules of

his own user program. In some cases the program may not be able to perform a Gauss Jordan Elimination to the system matrix and the right hand side vector and thus cause the program to stop. If this happens, an error message is displayed.

It is highly encouraged, to use the system function FNP to calculate the value of a polynomial at one specified X1 rather than a loop such as the following:

```
S=.0
DO 08 I=1,L,1
08 S=S+X**(I-1)*COEF(I)
YCALC=S
```

To call FNP is easier and faster, because FNP is a programmed Horner Scheme. The recommended call is then;

```
L1=L-1
YCALC=FNP(COEF,X,L1)
```

## 2.6. Sample User Program

### 2.6.1. Listing

PAGE 0001 FTN. 12:40 PM THU., 18 SEP., 1980

```

0001 FTN4,L
0002 PROGRAM DEMO2 (3,99)
0003 .....
0004 C
0005 : THIS IS A DEMONSTATION PROGRAM AND IT SHOWS THE CORRECT USE
0006 : OF THE TPL BINARY LIBRARY (ON TYPE 6 FILE XTPLBL) USING
0007 : EXAMPLES. FOR FURTHER QUESTIONS READ THE TPL-LIBRARY-BINDER
0008 : OR CONSULT THE SOFTWARE MANAGER.
0009 C
0010 * ..... DEMONSTRATE THE USE OF THE TPL BINARY LIBRARY.
0011 *
0012 COMMON / AFLD / A
0013 .....
0014 REAL A(256)
0015 .....
0016 .....
0017 .....
0018 .....
0019 .....
0020 INTEGER NOLF,NOCR(2),ICLR(3)
0021 .....
0022 DATA NOLF /006537B/
0023 DATA NOCR /000033B,040433B/
0024 DATA ICLR /015524B,015515B,006537B/
0025 DATA PI /3.141593/
0026 .....
0027 C FORMATS DEMO2 START
0028 101 FORMAT (///" HELLO! THIS IS PROGRAM DEMO AND WE'LL SEE H
0029 : OM TO USE THE MARVELLOUS TPL BINARY" A2/" LIBRARY! YOUR INTEREST
0030 : WILL BE GREATLY AWARDED BY EASIER PROGRAMMING."///)
0031 102 FORMAT (" GENERATING DATA POINTS"/SX"I" SX"X(I)" SX"Y(I)")
0032 103 FORMAT (2X,I4,2F9.3,A2)
0033 104 FORMAT (" DATA POINTS GENERATED!"/)
0034 105 FORMAT (" INITIALIZING THE PLOTTER")
0035 106 FORMAT (" PLOTTER INITIALIZED!"/)
0036 107 FORMAT (" DEFINE PLOTTER AND USER AREAS AND DRAW AXES")
0037 108 FORMAT (9X="6X"XMIN "6X"XMAX "6X"YMIN
0038 : "6X"YMAX"/9X,4("10X"")/
0039 : " HP7872"4("F10.3")"/
0040 : " USER"4("F10.3")"/)
0041 109 FORMAT (" PLOTTER AREAS DEFINED AND AXES DRAWN!"/)
0042 110 FORMAT (" DRAWING DATA POINTS INTO COORDINATES SYSTEM")
0043 111 FORMAT (" DATA POINTS DRAWN!"/)
0044 112 FORMAT (" CALCULATING A CURVE FIT THROUGH THE DATA POINTS
0045 : ")
0046 113 FORMAT (" ENTER DEGREE OF POLYNOMIAL (NORDER) TO FIT THRO
0047 : UGH THE POINTS "2A2)
0048 114 FORMAT (" ENTER IPRINT "2A2)
0049 115 FORMAT (/ " CURVE FIT DONE!"/)
0050 116 FORMAT (" CURVE FIT DONE!"/)
0051 117 FORMAT (" PLOTTING THE CURVE FIT")
0052 118 FORMAT (" CURVE FIT DRAWN!"/)
0053 149 FORMAT ("((3A2)))
0054 C FORMATS DEMO2 STOP
0055 .....
0056 C
0057 : .....
0058 : GET THE LU OF THE TERMINAL.
0059 :
0060 C
0061 LI = LOGLU(I)
0062 WRITE (LI, 101) NOLF
0063 .....
0064 C
0065 : .....
0066 : GENERATE DATA POINTS SCATTERED AROUND A POLYNOMIAL.
0067 :
0068 C
0069 : .....
0070 : CALL INITG (13)
0071 : WRITE (LI, 102)
0072 XSTART = -6.
0073 XSTOP = +2.
0074 .....
0075

```

```

0076      *****
0077      DX = (XSTOP-XSTART)/(NPNTS1-1)
0078      NORDER = 4
0079      COEF(1) = -1.
0080      COEF(2) = -0.500000
0081      COEF(3) = -0.000340
0082      COEF(4) = +0.300000
0083      COEF(5) = +0.055000
0084      A = (2.0*PI)*0.03125
0085      DO 01 I=1, NPNTS1-1
0086      X1(I) = XSTART+(I-1)*DX
0087      Y(I) = FNP(COEF,X1(I),NORDER) + 0.25*SIN(X1(I)/A)
0088      01 WRITE (LI, 103) I,X1(I),Y(I),NOLF
0089      ISTOP1 = NPNTS1-1
0090      DO 02 I=1, ISTOP1,2
0091      DUMMY = Y(I+1)
0092      Y(I+1) = Y(I)
0093      02 Y(I) = DUMMY
0094      WRITE (LI, 149) (ICLR,I=1,3,1)
0095      WRITE (LI, 104)
0096
0097      CCCCC
0098      CCCCC
0099      CCCCC
0100      :
0101      : INITIALIZE THE PLOTTER.
0102      :
0103      :
0104      WRITE (LI, 105)
0105      LP = 13
0106      CALL INITC (13)
0107      WRITE (LI, 149) ICLR
0108      WRITE (LI, 106)
0109
0110      CCCCC
0111      CCCCC
0112      CCCCC
0113      :
0114      : DEFINE PLOTTER AND USER AREAS; DRAW AXES.
0115      :
0116      :
0117      WRITE (LI, 107)
0118      XPMIN = 2.
0119      XPMAX = 12.
0120      YPMIN = 12.
0121      YPMAX = 22.
0122      XUMIN = -6.
0123      XUMAX = +2.
0124      YUMIN = -5.
0125      YUMAX = 10.
0126      WRITE (LI, 108) XPMIN,XPMAX,YPMIN,YPMAX,XUMIN,XUMAX,YUMIN,YUMAX
0127      ALPHA = 0.00
0128      XA = (XPMAX - XPMIN)/(XUMAX-XUMIN)
0129      XB = (XPMIN*XUMAX-XPMAX*XUMIN)/(XUMAX-XUMIN)
0130      XL = (XPMAX - XPMIN)
0131      ALPHAY = 90.00
0132      YA = (YPMAX - YPMIN)/(YUMAX-YUMIN)
0133      YB = (YPMIN*YUMAX-YPMAX*YUMIN)/(YUMAX-YUMIN)
0134      YL = (YPMAX - YPMIN)
0135      CALL SETSH (113,1.)
0136      CALL AXIS (XPMIN,YPMIN,XL,ALPHAY,2HX1,+2,XUMIN,XUMAX,4HF4.1,4,4)
0137      CALL AXIS (XPMIN,YPMIN,YL,ALPHAY,2H Y,-2,YUMIN,YUMAX,2HI2,2,3)
0138      WRITE (LI, 149) (ICLR,I=1,6,1)
0139      WRITE (LI, 109)
0140
0141      CCCCC
0142      CCCCC
0143      CCCCC
0144      :
0145      : PLOT DATA POINTS.
0146      :
0147      :
0148      WRITE (LI, 110)
0149      CALL SETSH (113,3.)
0150      DO 03 I=1, NPNTS1,1

```

PAGE 0003 DEM02 12:40 PM THU., 18 SEP., 1980

```

0151      XPLOT = X1(I)*XA+XB
0152      YPLOT = Y(I)*YA+YB
0153      CALL PLOT (XPLOT,YPLOT,2)
0154      03 CALL SYMBL (1)
0155          WRITE (LI, 149) ICLR
0156          WRITE (LI, 111)
0157
0158      CCCCC
0159      : .....
0160      : CALCULATE CURVE FIT THROUGH DATA POINTS.
0161      : .....
0162      :
0163      :
0164      :
0165      04 WRITE (LI, 112)
0166          WRITE (LI, 113) NOCR
0167          READ (LI, 113) NORDER
0168          WRITE (LI, 149) ICLR
0169          IF (NORDER.LT.0 .OR. NORDER.GT.6) GO TO 04
0170          WRITE (LI, 114) NOCR
0171          WRITE (LI, 149) (ICLR,I=1,2,1)
0172          WRITE (LI, 149) (ICLR,I=1,2,1)
0173          WRITE (LI, 149) (ICLR,I=1,2,1)
0174          IF (IPRINT.GE.0) WRITE (LI, 115)
0175          IF (IPRINT.LT.0) WRITE (LI, 116)
0176
0177      CCCCC
0178      : .....
0179      : PLOT POLYNOMIAL FIT.
0180      : .....
0181      :
0182      :
0183      :
0184      :
0185      :
0186      :
0187      :
0188      :
0189      :
0190      :
0191      :
0192      :
0193      :
0194      :
0195      :
0196      :
0197      :
0198      :
0199      :
0200      :
0201      :
0202      :
0203      :
0204      :
0205      :

```

FTN4 COMPILER: HP92060-16092 REV. 1926 (790430)

\*\* NO WARNINGS \*\* NO ERRORS \*\* PROGRAM = 01409

COMMON = 00000

# 2.6.2. Load map

DEMO2 10042 12642  
FNP 12643 12746

DEMONSTRATE THE USE OF THE TPL BINARY LIBRARY.  
Calculate n-th order polynomial

to load these programs, type (from LOADR): MS,ZTFLBL

DEMO2	12747	14746	2D-Approximation	/	DTA2	/	(DATA2 DTA2).
DEMO2	14747	16777	2D-Approximation	arrange	system	matrix and vector.	
DEMO2	17000	17131	2D-Approximation			compute summations.	
DEMO2	17132	17311	2D-Approximation	/	AB2	/	

AFLD	17312	20311	GSP	/	AFLD	/	A(256)	
AXIS	20312	22631	GSP					Draw and label axes.
INITG	22632	23644	GSP					read control array A; initialize plotter.
PLOT	23645	24375	GSP					move pen to a defined point.
SETSM	24376	25336	GSP					change plotter modes.
STOPG	25337	25401	GSP					terminate graphics.
SYMBL	25402	26136	GSP					print symbol at location of pen.
FCTR	26137	26142	GSP	/	FCTR	/	FX,FY	
FACTR	26143	26162	GSP					vary size of the plot.

LOGLU	26163	26240	92067-16268	REV.1903	790228
READF	26241	27202	92067-16125	REV.1940	790719
OPEN	27203	27500	92067-16125	REV.1903	790215
CLOSE	27501	27710	92067-16125	REV.1903	781229
CLRIO	27711	27717	750701	24998-16001	
QVRD	27720	27720	92067-16125	REV.1903	780526
\$SHVE	27721	30007	92067-16268	REV.1903	790202
LURQ	30010	30372	92067-16268	REV.1903	790223
.DADS	30373	30502	780818	24998-16001	
.DMP	30503	30650	780818	24998-16001	
.DDI	30651	31151	781021	24998-16001	
SESSN	31152	31167	92067-16125	REV.1903	780413
R/W\$	31170	31326	92067-16125	REV.1903	781214
P.PAS	31327	31355	92067-16125	REV.1903	740801
.DNC	31356	31365	780818	24998-16001	
PAUSE	31366	31466	771122	24998-16001	
\$ALRN	31467	31604	92067-16268	REV.1903	770715
FMTIO	31605	33103	24998-16002	REV.1926	790417
ERR0	33104	33173	771122	24998-16001	
TAN	33174	33300	780424	24998-16001	
ABS	33301	33307	750701	24998-16001	
.SNCS	33310	33451	780424	24998-16001	
.DDE	33452	33463	780818	24998-16001	
.DIN	33464	33471	780818	24998-16001	
.RTOI	33472	33565	780921	24998-16001	
.FPR	33566	33627	781106	24998-16001	
.SBT	33630	33670	770518	24998-16001	
.FCH	33671	33705	750701	24998-16001	
PAU.E	33706	33706	750701	24998-16001	
ER0.E	33707	33707	750701	24998-16001	
.CHRS	33710	33773	780424	24998-16001	
\$OPEN	33774	34150	92067-16125	REV.1903	790103
RW0UB	34151	34516	92067-16125	REV.1903	781003
RW0D\$	34517	34641	92067-16125	REV.1903	780801
FRMTR	34642	40277	24998-16002	REV.1926	798503
FMT.E	40300	40300	24998-16002	REV.1901	781107
IB4A2	40301	41601	56310-1X013	REV.1940	790802 1153
REIO	41602	41726	92067-16268	REV.1903	790316
RMPAR	41727	41771	781106	24998-16001	
PNAME	41772	42037	771121	24998-16001	
LUTRU	42040	42146	92067-16268	REV.1903	790223
IPUT	42147	42167	92067-16125	REV.1903	740801
\$SETP	42170	42214	781106	24998-16001	
.CFER	42215	42272	750701	24998-16001	
.LBT	42273	42323	770518	24998-16001	

15 PAGES RELOCATED 15 PAGES REQ'D NO PAGES EMA NO PAGES MSEG  
LINKS:BP PROGRAM:LB LOAD:TE COMMON:NC  
/LOADR:DEMO2 READY AT 12:44 PM THU., 18 SEPT, 1980  
/LOADR:SEND



### 2.6.3. Results

#### Affect of IPRINT on quantity of print out

##### 1) printed output

System matrix A and vector B before Gauss Jordan Elimination

	1	2	3	4	5	
1	.510E+02	-.102E+03	.487E+03	-.211E+04	.104E+05	-.578E+02
2	-.102E+03	.487E+03	-.211E+04	.104E+05	-.525E+05	.998E+02
3	.487E+03	-.211E+04	.104E+05	-.525E+05	.274E+06	-.113E+03
4	-.211E+04	.104E+05	-.525E+05	.274E+06	-.146E+07	-.944E+03
5	.104E+05	-.525E+05	.274E+06	-.146E+07	.787E+07	.113E+05
	1	2	3	4	5	

Equation system after Gauss Jordan Elimination

	1	2	3	4	5	
1	.100E+01	.000E+00	.000E+00	.000E+00	.000E+00	-.970E+00
2	.000E+00	.100E+01	.000E+00	.000E+00	.000E+00	-.416E+00
3	.000E+00	.000E+00	.100E+01	.000E+00	.000E+00	-.162E-01
4	.000E+00	.000E+00	.000E+00	.100E+01	.000E+00	.275E+00
5	.000E+00	.000E+00	.000E+00	.000E+00	.100E+01	.514E-01
	1	2	3	4	5	

IPRINT = 2

Coefficients COEF(I)

```

I = 1 -.970E+00
I = 2 -.416E+00
I = 3 -.162E-01
I = 4 .275E+00
I = 5 .514E-01

```

Equation system after Gauss Jordan Elimination

	1	2	3	4	5	
1	.100E+01	.000E+00	.000E+00	.000E+00	.000E+00	-.970E+00
2	.000E+00	.100E+01	.000E+00	.000E+00	.000E+00	-.416E+00
3	.000E+00	.000E+00	.100E+01	.000E+00	.000E+00	-.162E-01
4	.000E+00	.000E+00	.000E+00	.100E+01	.000E+00	.275E+00
5	.000E+00	.000E+00	.000E+00	.000E+00	.100E+01	.514E-01
	1	2	3	4	5	

IPRINT = 1

Coefficients COEF(I)

```

I = 1 -.970E+00
I = 2 -.416E+00
I = 3 -.162E-01
I = 4 .275E+00
I = 5 .514E-01

```

Coefficients COEF(I)

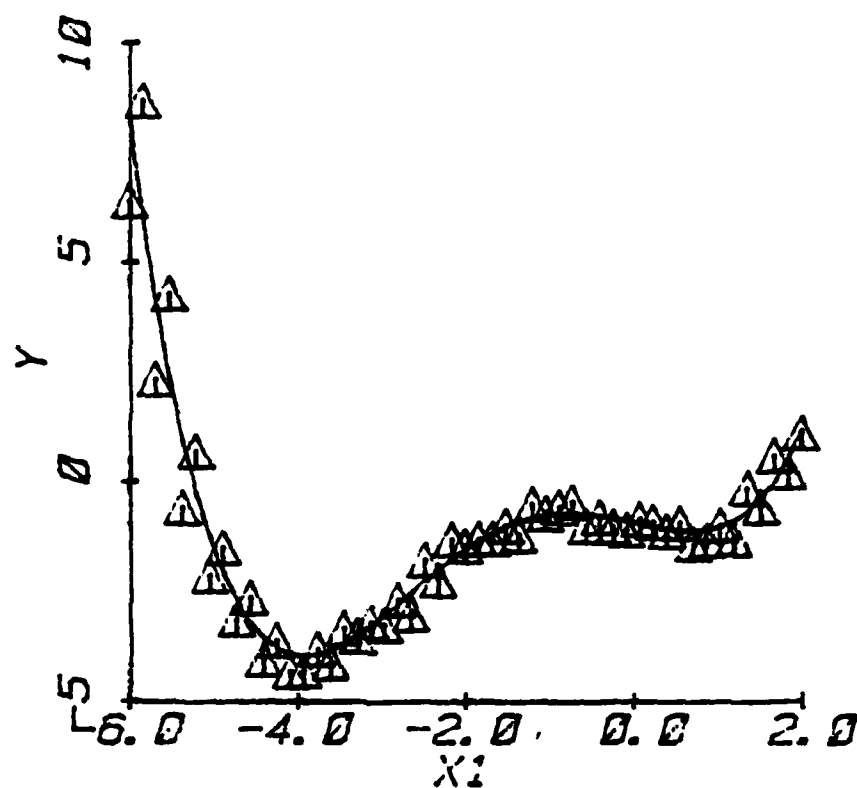
```

I = 1 -.970E+00
I = 2 -.416E+00
I = 3 -.162E-01
I = 4 .275E+00
I = 5 .514E-01

```

IPRINT < 1  
> 2

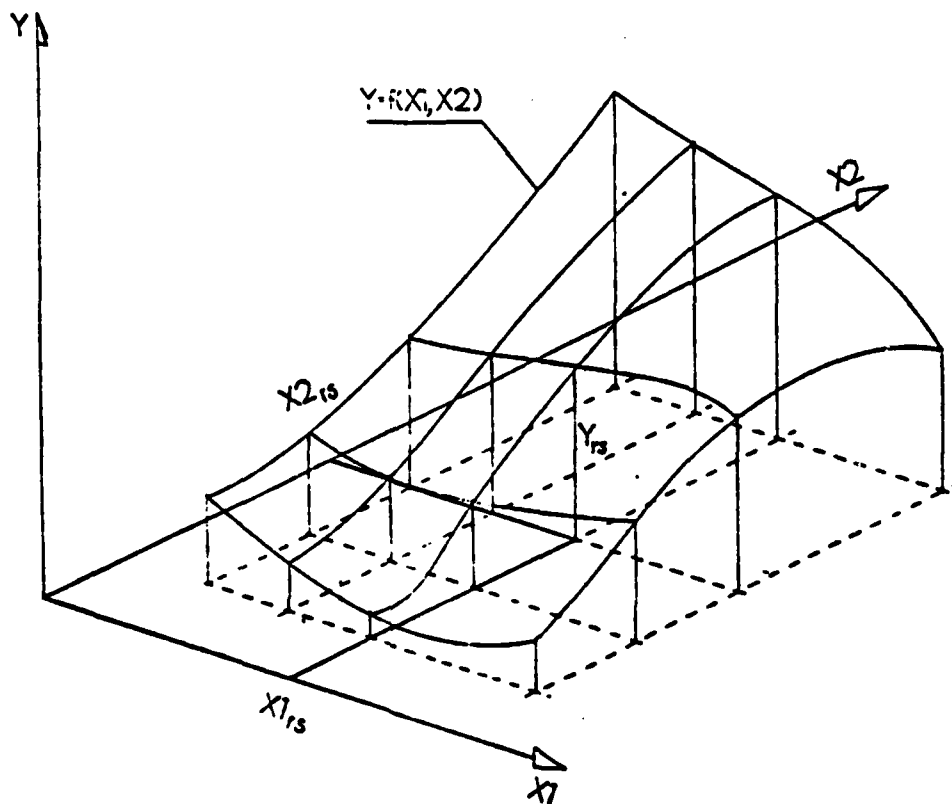
11) graphic output



Graphic output from PROGRAM DEMO2

### 3. THREE DIMENSIONAL APPROXIMATION

#### 3.1. Problem:



A data set of  $NPNTS1 * NPNTS2$  data points is given, where  $Y$  depends on the parameters  $X_1$  and  $X_2$ . The data pattern is to be approximated by a function  $Y = f(X_1, X_2)$ , so that the error between data points and analytically determined points is lowest.

### 3.2. Approach:

As for the two dimensional approximation, a polynomial, now with two independent variables  $X_1$  and  $X_2$ , is used.

$$Y = f(X_1, X_2)$$

$$\begin{aligned} Y = & C_{11} + C_{12} \cdot X_2 + C_{13} \cdot X_2^2 + \dots + C_{1M} \cdot X_2^{(M-1)} + \\ & + [C_{21} + C_{22} \cdot X_2 + C_{23} \cdot X_2^2 + \dots + C_{2M} \cdot X_2^{(M-1)}] \cdot X_1 + \\ & + [C_{31} + C_{32} \cdot X_2 + C_{33} \cdot X_2^2 + \dots + C_{3M} \cdot X_2^{(M-1)}] \cdot X_1^2 + \\ & \dots \\ & + [C_{L1} + C_{L2} \cdot X_2 + C_{L3} \cdot X_2^2 + \dots + C_{LM} \cdot X_2^{(M-1)}] \cdot X_1^{(L-1)} \\ Y = & \sum_{i=1}^L \left\{ \sum_{j=1}^M C_{ij} \cdot X_2^{(j-1)} \right\} \cdot X_1^{(i-1)} \end{aligned} \quad (3.1.)$$

The data arrangement for the three dimensional approximation already yields to the application of this method: at each constant  $X_1$  (i.e.:  $r = \text{constant}$ , too) there is a data set for various  $X_2$ . This is the case for example, when a pneumatic-velocity probe is calibrated. At each Mach number, the probe is balanced in yaw and the pitch angle is varied over the range of application. The probe output (each of two separate non-dimensional pressure differences) is a function of the Mach number and the pitch angle.

The least squares criterion becomes

$$R = \sum_{r=1}^{NPNTS1} \cdot \left\{ \sum_{s=1}^{NPNTS2} [f(x1_{rs}, x2_{rs}) - Y_{rs}]^2 \right\} \quad (3.2.)$$

where the indices r and s denote the individual data points. Using equation (2.1.) R becomes

$$R = \sum_{r=1}^{NPNTS1} \cdot \left\{ \sum_{s=1}^{NPNTS2} \left[ C_0 + C_{12} x2_{rs} + \dots + C_{m1} x2_{rs}^{m-1} + \right. \right. \\ \left. \left. + (C_{21} + C_{22} x2_{rs} + \dots + C_{2M} x2_{rs}^{M-1}) \cdot x1_{rs} \right. \right. \\ \left. \left. + (C_{31} + C_{32} x2_{rs} + \dots + C_{3M} x2_{rs}^{M-1}) \cdot x1_{rs}^2 + \right. \right. \\ \left. \left. \dots \right. \right. \\ \left. \left. + (C_{L1} + C_{L2} x2_{rs} + \dots + C_{LM} x2_{rs}^{M-1}) \cdot x1_{rs}^{L-1} \right. \right. \\ \left. \left. - Y_{rs} \right]^2 \right\}$$

The term in the [ ] - bracket shall be called B.

$$R = \sum_{r=1}^{NPNTS1} \cdot \left\{ \sum_{s=1}^{NPNTS2} \cdot B^2 \right\}$$

### 3.3. Solution:

To minimize the error,  $R$  is partially differentiated to the coefficients  $C_{ij}$  and then the partial derivatives are set to zero

$$\frac{\partial R}{\partial C_{ij}} \stackrel{!}{=} 0 \quad i = 1, \dots, L; \quad j = 1, \dots, M$$

$$\frac{\partial R}{\partial C_{ij}} = \sum_{r=1}^{NPTS1} \cdot \left\{ \sum_{s=1}^{NPTS2} 2 \cdot \delta \cdot \frac{\partial \delta}{\partial C_{ij}} \right\} = 0$$

Assuming, that the summations extend over all data points,  $\Sigma\Sigma$  should be understood to mean  $\sum_{r=1}^{NPTS1} \cdot \left\{ \sum_{s=1}^{NPTS2} \right\}$ . Performing the differentiations and rearranging the equations in matrix notation (which, while lengthy, is logically straight forward), we get, in matrix notation,

		1				2			
		(1-1) * M+1		1 & M		(2-1) * M+1		2 * M	
1	(1-1) * M+1	NPNTS1 * NPNTS2	$\Sigma \Sigma X_{rs}$	...	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}$	$\Sigma \Sigma X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{M-1}$
	...	$\Sigma \Sigma X_{rs}$	$\Sigma \Sigma X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^M$	$\Sigma \Sigma X_{rs}$	$\Sigma \Sigma X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{M-1}$
	1 & M	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^M$	...	$\Sigma \Sigma X_{rs}^{2M-2}$	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^M$	...	$\Sigma \Sigma X_{rs}^{2M-2}$
2	(2-1) * M	$\Sigma \Sigma X_{rs}$	$\Sigma \Sigma X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^2$	$\Sigma \Sigma X_{rs}^3$	...	$\Sigma \Sigma X_{rs}^{M-1}$
	...	$\Sigma \Sigma X_{rs}$	$\Sigma \Sigma X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^M$	$\Sigma \Sigma X_{rs}^2$	$\Sigma \Sigma X_{rs}^3$	...	$\Sigma \Sigma X_{rs}^{M-1}$
	2 * M	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^M$	...	$\Sigma \Sigma X_{rs}^{2M-2}$	$\Sigma \Sigma X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^M$	...	$\Sigma \Sigma X_{rs}^{2M-2}$
...									
L	(L-1) * M	$\Sigma \Sigma X_{rs}^{L-1}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^{L-1}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^{M-1}$
	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^M$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^M$
	L * M	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^{M-1}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^M$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^{2M-2}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}$	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^2$	...	$\Sigma \Sigma X_{rs}^{L-1} \cdot X_{rs}^M$

A





### 3.4. Equation System:

#### 3.4.1. System Matrix A

This system matrix needs some examination. A crucial step is the introduction of submatrices 1. class into the system matrix.

$$A = \begin{bmatrix} A_{11}^* & A_{12}^* & A_{1L}^* \\ A_{21}^* & A_{22}^* & A_{2L}^* \\ A_{L1}^* & A_{L2}^* & A_{LL}^* \end{bmatrix}$$

The asterisk \* denotes a submatrix 1. class where

$$A_{11}^* = \begin{bmatrix} NPNTS1 * NPNTS2 & \sum \sum x_{1s} x_{2s} & \sum \sum x_{1s} x_{2s}^{M-1} \\ \sum \sum x_{1s} x_{2s} & \sum \sum x_{1s}^2 x_{2s}^2 & \sum \sum x_{1s}^2 x_{2s}^M \\ \sum \sum x_{1s}^{M-1} x_{2s} & \sum \sum x_{1s}^M x_{2s}^M & \sum \sum x_{1s}^{2M-2} x_{2s}^M \end{bmatrix}$$

$$A_{12}^* = \begin{bmatrix} \sum \sum x_{1s} & \sum \sum x_{1s} x_{2s} & \sum \sum x_{1s} x_{2s}^{M-1} \\ \sum \sum x_{1s} x_{2s} & \sum \sum x_{1s} x_{2s}^2 & \sum \sum x_{1s} x_{2s}^M \\ \sum \sum x_{1s} x_{2s}^{M-1} & \sum \sum x_{1s} x_{2s}^M & \sum \sum x_{1s} x_{2s}^{2M-2} \end{bmatrix}$$

$$A_{1L}^* = \begin{bmatrix} \sum \sum x_{1s}^2 & \sum \sum x_{1s}^2 x_{2s} & \sum \sum x_{1s}^2 x_{2s}^{M-1} \\ \sum \sum x_{1s}^2 x_{2s} & \sum \sum x_{1s}^2 x_{2s}^2 & \sum \sum x_{1s}^2 x_{2s}^M \\ \sum \sum x_{1s}^2 x_{2s}^{M-1} & \sum \sum x_{1s}^2 x_{2s}^M & \sum \sum x_{1s}^2 x_{2s}^{2M-2} \end{bmatrix}$$

etc.

Now it can be seen that all submatrices on the same diagonal bands

2. order are identical. The submatrices  $A_{ij}^*$  subsequently are renamed  $A_k^*$

$$A_1^* = A_{11}^*$$

$$A_2^* = A_{21}^* = A_{12}^*$$

$$A_3^* = A_{31}^* = A_{22}^* = A_{13}^*$$

$$A_4^* = A_{41}^* = A_{32}^* = A_{23}^* = A_{14}^*$$

⋮

$$A_L^* = A_{L1}^* = A_{L-1,2}^* = A_{L-2,3}^* = \dots = A_{2,L-1}^* = A_{1,L}^*$$

$$A_{L+1}^* = A_{L,2}^* = A_{L-1,3}^* = \dots = A_{2,L-1}^* = A_{2,L}^*$$

$$A_{L+2}^* = \dots = A_{L,3}^* = \dots = A_{2,L-1}^* = A_{3,L}^*$$

!

$$A_{2L-1}^* = A_{L,L}^*$$

where  $A_k^*$  can be written as

$$A_k^* = \begin{vmatrix} \sum \sum x_{rs}^{k-1} x_{rs}^0 & \sum \sum x_{rs}^{k-1} x_{rs}^1 & \sum \sum x_{rs}^{k-1} x_{rs}^{N-1} \\ \sum \sum x_{rs}^{k-1} x_{rs}^1 & \sum \sum x_{rs}^{k-1} x_{rs}^2 & \sum \sum x_{rs}^{k-1} x_{rs}^N \\ \sum \sum x_{rs}^{k-1} x_{rs}^{N-1} & \sum \sum x_{rs}^{k-1} x_{rs}^N & \sum \sum x_{rs}^{k-1} x_{rs}^{2N-2} \end{vmatrix}$$

$k = 1, \dots, L-1$

Like the system matrix for the two dimensional approximation this submatrix 1. class is not only symmetrical to its main diagonal line 1. order, but also the elements on each diagonal line 2. order are identical. The elements  $a_{k;ij}$  of the submatrix  $A_k$  therefore are

renamed to  $a_{k;L}$ , where  $k$  specifies the place of the submatrix in the system matrix and  $i$  and  $j$  define the place of  $a_{k;ij}$  in the submatrix  $A_k^*$ .

$$\begin{aligned}
 a_{k;1} &= a_{k;11} \\
 a_{k;2} &= a_{k;21} = a_{k;12} \\
 a_{k;3} &= a_{k;31} = a_{k;22} = a_{k;13} \\
 &\vdots \\
 a_{k;M} &= a_{k;M1} = a_{k;M-12} = a_{k;M-23} = \dots = a_{k;2M-1} = a_{k;1M} \\
 a_{k;M+1} &= a_{k;M2} = a_{k;M-13} = \dots = a_{k;3M-1} = a_{k;2M} \\
 a_{k;M+2} &= a_{k;M3} = \dots = a_{k;4M-1} = a_{k;3M} \\
 &\vdots \\
 a_{k;2M-1} &= a_{k;MM}
 \end{aligned}$$

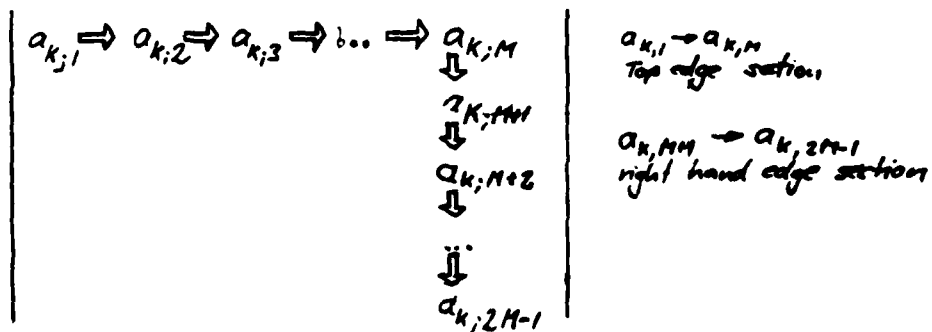
where  $a_{k;L}$  can be written as

$$\begin{aligned}
 a_{k;L} &= \sum_{r=1}^{NPNTS1} \sum_{s=1}^{NPNTS2} x1_{rs}^{k-1} x2_{rs}^{L-1} \\
 k &= 1, \dots, 2L-1 ; \quad L = 1, \dots, 2M-1
 \end{aligned} \tag{3.4.}$$

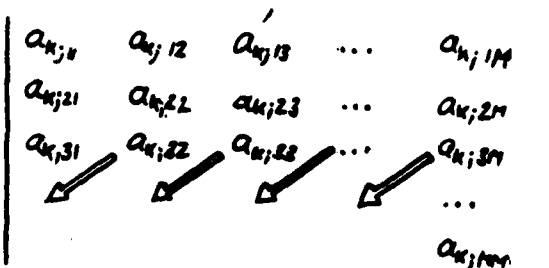
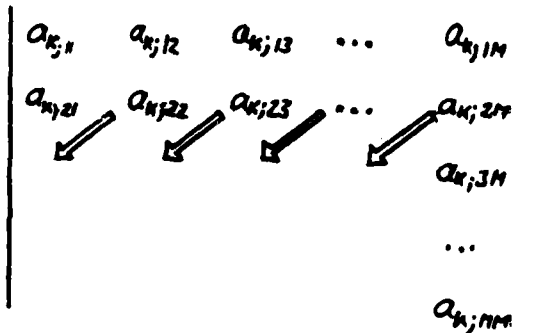
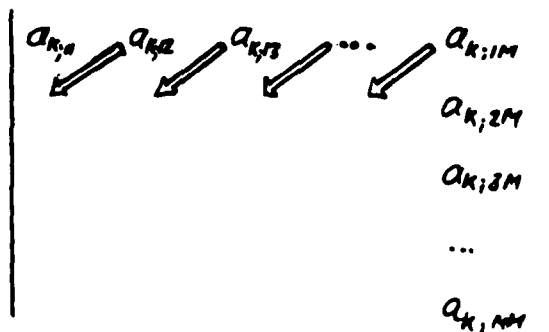
Equation (3.4.) therefore is the fundamental step towards the programming of the system matrix and is programmed in REAL FUNCTION S3. Since the data  $X1$ ,  $X2$  and  $Y$  are already provided through a COMMON block named DATA3, only NPNTS1, NPNTS2, IPOWR1 (=  $k-1$ ) and IPOWR2 (=  $L-1$ ) have to be passed to the function that returns the value of  $a_{k;L}$  through the function name S3.

SUBROUTINE MAT31 presets a submatrix  $A_k^*$  in the following way:

- i) Preset edge section elements (using REAL FUNCTION S3)



- ii) Copy defined elements diagonally



$$\begin{array}{c}
 \begin{array}{ccccc}
 a_{k;11} & a_{k;12} & a_{k;13} & \dots & a_{k;1M} \\
 a_{k;21} & a_{k;22} & a_{k;23} & \dots & a_{k;2M} \\
 a_{k;31} & a_{k;32} & a_{k;33} & \dots & a_{k;3M} \\
 \dots & \dots & \dots & \dots & \dots \\
 & \swarrow & \swarrow & \swarrow & \swarrow \\
 & & & & a_{k;MM}
 \end{array} \\
 \\
 \begin{array}{ccccc}
 a_{k;11} & a_{k;12} & a_{k;13} & \dots & a_{k;1M} \\
 a_{k;21} & a_{k;22} & a_{k;23} & \dots & a_{k;2M} \\
 a_{k;31} & a_{k;32} & a_{k;33} & \dots & a_{k;3M} \\
 \dots & \dots & \dots & \dots & \dots \\
 a_{k;M1} & a_{k;M2} & a_{k;M3} & \dots & a_{k;MM}
 \end{array}
 \end{array}
 \quad A_k^* =$$

This subroutine that needs NPNTS1, NPNTS2, M and IPOWR1 (= k-1) as input parameters, returns  $A_k^*$  (SUBM1(7,7)) to SUBROUTINE MAT3, which presets the entire system matrix in the following way

- 1) Preset edge section submatrices (using SUBROUTINE MAT31)

$$\begin{array}{c}
 A_1^* \Rightarrow A_2^* \Rightarrow \dots \Rightarrow A_L^* \\
 \downarrow \\
 A_{L+1}^* \\
 \downarrow \\
 \vdots \\
 \downarrow \\
 A_{2L-1}^*
 \end{array}
 \quad
 \begin{array}{l}
 A_1^* \rightarrow A_L^* \\
 \text{Top edge section} \\
 A_{L+1}^* \rightarrow A_{2L-1}^* \\
 \text{right hand edge section}
 \end{array}$$

ii) Copy defined submatrices diagonally

$$\begin{array}{c}
 \left| \begin{array}{cccc}
 A_{11}^* & A_{12}^* & \dots & A_{1L}^* \\
 & A_{22}^* & & A_{2L}^* \\
 & & & \dots \\
 & & & A_{LL}^*
 \end{array} \right| \\
 \\
 \left| \begin{array}{cccc}
 A_{11}^* & A_{12}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & \dots & A_{2L}^* \\
 & & & \dots \\
 & & & A_{LL}^*
 \end{array} \right| \\
 \\
 \left| \begin{array}{cccc}
 A_{11}^* & A_{12}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & \dots & A_{2L}^* \\
 \dots & \dots & \dots & \dots \\
 & & & A_{LL}^*
 \end{array} \right| \\
 \\
 A = \left| \begin{array}{cccc}
 A_{11}^* & A_{12}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & \dots & A_{2L}^* \\
 \dots & \dots & \dots & \dots \\
 A_{L1}^* & A_{L2}^* & \dots & A_{LL}^*
 \end{array} \right|
 \end{array}$$

The parameters to call this subroutine are NPNTS1 and NPNTS2 (# of data points), L and M (<the order of the polynomial fit + 1> in X1 and X2 - direction) and IPRINT. The coefficients  $C_{ij}$  are returned.

### 3.4.2. Right hand side vector B

The right hand side vector B will be divided up into L subvectors 1. class.

$$B = \begin{bmatrix} B_1^* \\ B_2^* \\ \dots \\ B_L^* \end{bmatrix}$$

Again, the asterisk (\*) denotes a subvector 1. class. Now the subvectors 1. class  $B_k^*$  can be written as

$$B_k^* = \begin{bmatrix} \sum \sum Y_{rs} \cdot X1_{rs}^{k-1} \\ \sum \sum Y_{rs} \cdot X1_{rs}^{k-1} \cdot X2_{rs} \\ \dots \\ \sum \sum Y_{rs} \cdot X1_{rs}^{k-1} \cdot X2_{rs}^M \end{bmatrix}$$

$$k = 1, \dots, L$$

Finally, the elements of the subvectors 1. class  $B_k^*$  can be written as

$$b_{k;l} = \sum \sum Y_{rs} \cdot X1_{rs}^{k-1} \cdot X2_{rs}^{l-1}$$

$$k = 1, \dots, L; \quad l = 1, \dots, M$$

To compute  $b_{k;l}$ , REAL FUNCTION S3 is used. Data X1, X2 and Y are available through a COMMON block named DATA3. NPNTS1, NPNTS2, IPOWR1 (= k-1), IPOWR2 (= L-1) and IY (=1) have to be passed to the function and the value of  $b_{k;l}$  is returned through the function name S3.

Using FORTRAN programming language, the allocation of  $b_{k;L}$  can be performed by two stacked DO loops.

```
I=0
DO 08 L1=1,L,1
  IPOWR1=L1-1
  DO 08 M1=1,M,1
    IPOWR2=M1-1
    I=I+1
08 B(I) = S3(NPNTS1,NPNTS2,IPOWR1,IPOWR2,1)
```



### 3.5. Software:

The software to compute the coefficients for a three dimensional approximation is described in Appendix A and is implemented in the TPL HP-21MX computer system. To work correctly with these program modules, the user has to conform to the following conventions:

- i) Provide the data in three arrays (Type: REAL) of 16 \* 16 elements through a COMMON block named DATA3.  
  
COMMON / DATA3 / X1, X2, Y  
REAL X1 (16,16), X2(16,16), Y(16,16)
- ii) Dimension an array (Type: REAL) of 7\*7 elements to contain the coefficients.  
  
REAL COEF (7,7)
- iii) Define the parameters NPNTS1, NPNTS2, L, M and IPRINT.  
  
NPNTS1 ... # of X1 variations  
          L ≤ NPNTS1 ≤ 16  
  
NPNTS2 ... # of X2 variations  
          M ≤ NPNTS2 ≤ 16  
  
L ... (desired order of approximation  
      polynomial w.r.t. X1) + 1  
  
M ... (desired order of approximation  
      polynomial w.r.t. X2) + 1  
  
IPRINT ... controls quantity of print out  
          2 ... Print system matrix and  
              right hand side vector  
              before and after Gauss  
              Jordan Elimination  
          1 ... Print equation system after  
              Gauss Jordan Elimination  
          <0 ... No print out  
          >0 ... Display equation (3.1.)  
              with the actual parameters

- iv) If a user program uses subroutine MAT3, the software modules have to be loaded using the procedure outlined in section 2.4. iv.

If all these requirements are met, the correct call for the subroutine is:

```
CALL MAT (NPNTS1,NPNTS2,L,M,COEF,IPRINT)
```

Upon completed execution of this approximation routine the array COEF contains the coefficients. Externals used by MAT3 are: AB3, DATA3, MAT31, IEL3, S3. Under no circumstances may the user use any of these names for modules of his own program.

### 3.6. Sample User Program

#### 3.6.1. FTN4 Compiler listing of sample user program

```

0001 FTN4,L
0002 PROGRAM LAB (3,99)
0003 *Plot calibration points and calculate coefficients.
0004
0005 COMMON / AFLD / PLOTR
0006 COMMON / DATA3 / X1,X2,Y
0007 COMMON / AB3 / A,B
0008
0009 REAL PLOTR(256)
0010 REAL X1(16,16),X2(16,16),Y(16,16)
0011 REAL A(49,49),B(49)
0012
0013 REAL YT(16,16),R(16),COEFF(7,7)
0014 INTEGER IDC(144),IFILE(3),NOCR(2),ICLR(3),YTITLE(4),ZTITLE(4)
0015
0016 DATA PI /3.141593/
0017 DATA NOLF /006537B/
0018 DATA NOCR /000033B,040433B/
0019 DATA ICLR /015524B,015515B,006537B/
0020
0021 C FORMATS LAB START
0022 121 FORMAT (" Enter raw data file name "2A2)
0023 122 FORMAT (3A2)
0024 123 FORMAT (" Select 1 ... Pitch angle or 2 .
0025 * Velocity X "2A2)
0026 124 FORMAT (" Select 1 ... Pitch angle v/s Gamma and Beta (
0027 * = ( , ) )"/
0028 * = 2 ... "1H" " "1H" " v/s Gamma a
0029 *nd Delta ( = ( ) ) "2A2)
0030 125 FORMAT (" Select 1 ... Velocity v/s Gamma and Beta ( X=
0031 *X( , ) )"/
0032 * = 2 ... "1H" " v/s Gamma and Delta
0033 * ( X=X( , ) ) "2A2)
0034 126 FORMAT (" Enter # of Mach Numbers and # of pitch angle
0035 * "2A2)
0036 160 FORMAT (" Enter 1 to print input data "2A2)
0037 131 FORMAT (/ " Calibrate = ( , ) )"/
0038 132 FORMAT (/ " Calibrate = ( , ) )"/
0039 133 FORMAT (/ " Calibrate X=X( , ) )"/
0040 134 FORMAT (/ " Calibrate X=X( , ) )"/
0041 127 FORMAT (24X"9X"9X"9X")
0042 128 FORMAT (24X"9X"9X"9X")
0043 129 FORMAT (24X"9X"9X"9X")
0044 130 FORMAT (24X"9X"9X"9X")
0045 150 FORMAT (" minimum values",3(1X,F9.3))
0046 151 FORMAT (" maximum values",3(1X,F9.3))
0047 152 FORMAT (" of user coordinates"3X"X1"8X"X2"9X"Y")
0048 153 FORMAT (" Enter 1 to redefine these data "2A2)
0049 154 FORMAT (" Enter min, max, min, ",
0050 * " max, # partitions, # " )
0051 155 FORMAT (" Enter min, max, min, ",
0052 * " max, # partitions, # " )
0053 156 FORMAT (" Enter min, max, min, max, X min, X ma
0054 * " # partitions, # X" )
0055 157 FORMAT (" Enter min, max, min, max, X min, X ma
0056 * " # partitions, # X" )
0057 158 FORMAT (1X,3(F8.3,1X,F8.3,5X),3I4)
0058 101 FORMAT (" Enter order M of X1 - approximation "2A2)
0059 102 FORMAT (" Enter order N of X2 - approximation "2A2)
0060 107 FORMAT (" Enter IPRINT "2A2)
0061 105 FORMAT (" Select 1 ... to calculate the absolute error"/8X
0062 * "2 9X"the relative error "2A2)
0063 103 FORMAT (" Change coefficients ... 1"/"2A2)
0064 * " Change axes
0065 104 FORMAT (" Enter ALPHAX, ALPHAY, X0 and Y0")
0066 149 FORMAT (" ((3A2)))
0067 611 FORMAT ("1///// Three new data from file "3A
0068 612 FORMAT (/16X,12" Mach number"/)
0069 613 FORMAT (6X"Gamma"6X"Beta"7X"Phi"/)
0070 614 FORMAT (6X"Gamma"5X"Delta"7X"Phi"/)
0071 615 FORMAT (6X"Gamma"6X"Beta"5X"X vel"/)
0072 616 FORMAT (6X"Gamma"5X"Delta"5X"X vel"/)
0073 617 FORMAT (1X,4(1X,F9.3))
0074 618 FORMAT (/9X"X1"8X"X2"9X"Y")
0075 602 FORMAT ((3X,16(6X,12)))
0076 603 FORMAT (1X,12,16(1X,F7.3)/3(3X,16(1X,F7.3)/))

```

```

0077 604 FORMAT (///" Absolute error at each point ( error = (calculated
0078 *value - measured value) )"/)
0079 644 FORMAT (///" Relative error at each point in % ( error = (calcul
0080 *ted value - measured value)/measured value )"/)
0081 606 FORMAT (1X,12,18(1X,F6.1)/3(3X,18(1X,F7.3)/))
0082 1101 FORMAT (8H Beta )
0083 1102 FORMAT (8H Delta )
0084 1103 FORMAT (8H Phi )
0085 1104 FORMAT (8H X vel )
0086 LI = LOGLU(I)
0087 LO = 6
0088
0089
0090
0091
0092
0093
0094
0095
0096
0097
0098
0099
0100
0101
0102
0103
0104
0105
0106
0107
0108
0109
0110
0111
0112
0113
0114
0115
0116
0117
0118
0119
0120
0121
0122
0123
0124
0125
0126
0127
0128
0129
0130
0131
0132
0133
0134
0135
0136
0137
0138
0139
0140
0141
0142
0143
0144
0145
0146
0147
0148
0149
0150
0151
0152

```

.....

read raw data file.

.....

```

41 WRITE (LI, 121) NOCR
READ (LI, 122) IFILE
WRITE (LI, 149) ICLR
42 WRITE (LI, 123) NOCR
READ (LI, *) IPX
WRITE (LI, 149) ICLR
IF (IPX.LT.1 .OR. IPX.GT.2) GO TO 42
43 CONTINUE
IF (IPX.EQ.1) WRITE (LI, 124) NOCR
IF (IPX.EQ.2) WRITE (LI, 125) NOCR
READ (LI, *) IBD
WRITE (LI, 149) ICLR,I=1,2,1)
IF (IBD.LT.1 .OR. IBD.GT.2) GO TO 43
IF (IPX.EQ.1) IRECY=13
IF (IPX.EQ.2) IRECY=9
IF (IPX.EQ.1 .AND. IBD.EQ.1 ) WRITE (LI, 131)
IF (IPX.EQ.1 .AND. IBD.EQ.2 ) WRITE (LI, 132)
IF (IPX.EQ.2 .AND. IBD.EQ.1 ) WRITE (LI, 133)
IF (IPX.EQ.2 .AND. IBD.EQ.2 ) WRITE (LI, 134)
CALL OPEN (IDCB,IERR,IFILE,IOPTN,0,28,144)
IF (IERR.LT.0) GO TO 41
CALL READF (IDCB,IERR,X1,512,LEN,1)
IF ( IERR .LT. 0 ) STOP 0002
CALL READF (IDCB,IERR,X2,512,LEN,5)
IF ( IERR .LT. 0 ) STOP 0003
CALL READF (IDCB,IERR,YT,512,LEN,IRECY)
IF ( IERR .LT. 0 ) STOP 0004
CALL CLOSE (IDCB,IERR,0)
44 WRITE (LI, 126) NOCR
READ (LI, *) NMACH,NPITCH
WRITE (LI, 149) ICLR
IF (NMACH.GT.16) GO TO 44
IF (NPITCH.GT.16) GO TO 44
DO 801 I=1,NMACH,1
DO 801 J=1,NPITCH,1
801 Y(I,J)=YT(I,J)

```

.....

Correct input data and optional output of data, coefficient  
are based upon.

.....

```

IF (IBD.EQ.2) GO TO 48
DO 47 I=1,NMACH,1
DO 47 J=1,NPITCH,1
47 X2(I,J)=X2(I,J)/X1(I,J)
48 WRITE (LI, 160) NOCR
READ (LI, *) IDUM
WRITE (LI, 149) ICLR
IF (IDUM.NE.1) GO TO 55
WRITE (LO, 611) IFILE
DO 54 I=1,NMACH,1
WRITE (LO, 612) I
54 IF (IPX.EQ.1 .AND. IBD.EQ.1) WRITE (LO, 613)

```

```

0153 IF (IPX.EQ.1 .AND. IBD.EQ.2) WRITE (LO, 614)
0154 IF (IPX.EQ.2 .AND. IBD.EQ.1) WRITE (LO, 615)
0155 IF (IPX.EQ.2 .AND. IBD.EQ.2) WRITE (LO, 616)
0156 DO 54 J=1, NPITCH, 1
0157 54 WRITE (LO, 617) X1(I,J), X2(I,J), Y(I,J)
0158 WRITE (LO, 618)
0159
0160
0161
0162
0163
0164 .....
0165 . Calculate minimum and maximum values for all coordinates and
0166 . optional redefinition of minimum and maximum values by user.
0167 .....
0168 55 XUMAX=X1(1,1)
0169 XUMIN=X1(1,1)
0170 YUMAX=X2(1,1)
0171 YUMIN=X2(1,1)
0172 ZUMAX=Y(1,1)
0173 ZUMIN=Y(1,1)
0174 DO 45 I=1, NMACH, 1
0175 DO 45 J=1, NPITCH, 1
0176 IF (X1(I,J).GT.XUMAX) XUMAX=X1(I,J)
0177 IF (X1(I,J).LT.XUMIN) XUMIN=X1(I,J)
0178 IF (X2(I,J).GT.YUMAX) YUMAX=X2(I,J)
0179 IF (X2(I,J).LT.YUMIN) YUMIN=X2(I,J)
0180 IF (Y(I,J).GT.ZUMAX) ZUMAX=Y(I,J)
0181 IF (Y(I,J).LT.ZUMIN) ZUMIN=Y(I,J)
0182 45 CONTINUE
0183 IF (IPX.EQ.1 .AND. IBD.EQ.1) WRITE (LI, 127)
0184 IF (IPX.EQ.1 .AND. IBD.EQ.2) WRITE (LI, 128)
0185 IF (IPX.EQ.2 .AND. IBD.EQ.1) WRITE (LI, 129)
0186 IF (IPX.EQ.2 .AND. IBD.EQ.2) WRITE (LI, 130)
0187 WRITE (LI, 150) XUMIN, YUMIN, ZUMIN
0188 WRITE (LI, 151) XUMAX, YUMAX, ZUMAX
0189 WRITE (LI, 152)
0190 INCX = 4
0191 INCY = 4
0192 INCZ = 4
0193 WRITE (LI, 153) NOCR
0194 READ (LI, *) I
0195 WRITE (LI, 149) ICLR
0196 IF (I.NE.1) GO TO 49
0197 IF (IPX.EQ.1 .AND. IBD.EQ.1) WRITE (LI, 154)
0198 IF (IPX.EQ.2 .AND. IBD.EQ.2) WRITE (LI, 155)
0199 IF (IPX.EQ.2 .AND. IBD.EQ.1) WRITE (LI, 156)
0200 IF (IPX.EQ.1 .AND. IBD.EQ.2) WRITE (LI, 157)
0201 READ (LI, *) XUMIN, XUMAX, YUMIN, YUMAX, ZUMIN, ZUMAX, INCX, INCY, INCZ
0202 WRITE (LI, 158) XUMIN, XUMAX, YUMIN, YUMAX, ZUMIN, ZUMAX, INCX, INCY, INCZ
0203
0204
0205
0206
0207 .....
0208 . Initialize plotter; define plotter area; calculate scaling
0209 . coefficients XA, XB, YA, YB, ZA and ZB.
0210 .....
0211 49 CALL INITG (13)
0212 XPHIN = 0.
0213 XPMAX = 10.
0214 YPHIN = 0.
0215 YPMAX = 10.
0216 ZPHIN = 0.
0217 ZPMAX = 10.
0218 ALPHAX = 330.
0219 ALPHAY = 30.
0220 ALPHAZ = 90.
0221 XA = (XPMAX - XPHIN) / (XUMAX - XUMIN)
0222 XB = (XPHIN * XUMAX - XPMAX * XUMIN) / (XUMAX - XUMIN)
0223 XL = (XPMAX - XPHIN) / (YUMAX - YUMIN)
0224 YA = (YPMAX - YPHIN) / (YUMAX - YUMIN)
0225 YB = (YPHIN * YUMAX - YPMAX * YUMIN) / (YUMAX - YUMIN)
0226 YL = (YPMAX - YPHIN) / (ZUMAX - ZUMIN)
0227 ZA = (ZPMAX - ZPHIN) / (ZUMAX - ZUMIN)
0228

```

```

0229      ZB      = (ZPMIN*ZUMAX-ZPMAX*ZUMIN)/(ZUMAX-ZUMIN)
0230      ZL      = (ZPMAX      -ZPMIN      )
0231      XO      = 8.
0232      YO      = 12.
0233      85 PLOTR(61) = (ALPHAX*PI)/180.
0234      PLOTR(62) = (ALPHAY*PI)/180.
0235      PLOTR(63) = (ALPHAZ*PI)/180.
0236      PLOTR(64) = XO
0237      PLOTR(65) = YO
0238      PLOTR(66) = XA
0239      PLOTR(67) = XB
0240      PLOTR(68) = YA
0241      PLOTR(69) = YB
0242      PLOTR(70) = ZA
0243      PLOTR(71) = ZB
0244      GO TO (86,87) IBD
0245      86 CALL CODE
0246      WRITE (YTITLE,1101)
0247      GO TO 88
0248      87 CALL CODE
0249      WRITE (YTITLE,1102)
0250      GO TO (301,302) IPX
0251      301 CALL CODE
0252      WRITE (ZTITLE,1103)
0253      GO TO 303
0254      302 CALL CODE
0255      WRITE (ZTITLE,1104)
0256      303 CALL SETSM (113,1.)
0257      CALL AXIS (XO,YO,XL,ALPHAX,6H Gamma, 8,XUMIN,XUMAX,4HF6.2,6,INCX)
0258      CALL AXIS (XO,YO,ZL,ALPHAY,YTITLE,-8,YUMIN,YUMAX,4HF6.2,6,INCY)
0259      CALL AXIS (XO,YO,ZL,ALPHAZ,ZTITLE,-8,ZUMIN,ZUMAX,4HF6.2,6,INCZ)
0260
0261
0262
0263
0264
0265
0266
0267
0268
0269
0270
0271
0272
0273
0274
0275
0276
0277
0278
0279
0280
0281
0282
0283
0284
0285
0286
0287
0288
0289
0290
0291
0292
0293
0294
0295
0296
0297
0298
0299
0300
0301
0302
0303
0304

```

.....

Plot measured calibration surface.

.....

```

CALL SETSM (113,2.)
DO 01 I=1,NMACH,1
DO 01 J=1,NPITCH,1
CALL THRTM (XPLOT,YPLOT,X1(I,J),X2(I,J),Y(I,J))
IF ( J .EQ. 1 ) CALL PLOT (XPLOT,YPLOT,2)
IF ( J .GT. 1 ) CALL PLOT (XPLOT,YPLOT,3)
01 CONTINUE
DO 02 J=1,NPITCH,1
DO 02 I=1,NMACH,1
CALL THRTM (XPLOT,YPLOT,X1(I,J),X2(I,J),Y(I,J))
IF ( I .EQ. 1 ) CALL PLOT (XPLOT,YPLOT,2)
IF ( I .GT. 1 ) CALL PLOT (XPLOT,YPLOT,3)
02 CONTINUE

```

.....

Calculate calibration surface coefficients.

.....

```

91 CALL SETSM (113,0.)
WRITE (LI, 101) NOCR
READ (LI, 8) MORDER
WRITE (LI, 149) ICLR
IF (MORDER.GT.6) GO TO 91
92 WRITE (LI, 102) NOCR
READ (LI, 8) NORDER
WRITE (LI, 149) ICLR
WRITE (LI, 107) NOCR
READ (LI, 8) IPRINT
WRITE (LI, 149) ICLR
IF (NORDER.GT.6) GO TO 92
M=NORDER+1
N=NORDER+1
CALL MAT3 (NMACH,NPITCH,M,N,COEFF,IPRINT)

```

```

0305 .....
0306 .....
0307 .....
0308 .....
0309 .....
0310 .....
0311 .....
0312 .....
0313 .....
0314 .....
0315 .....
0316 .....
0317 .....
0318 .....
0319 .....
0320 .....
0321 .....
0322 .....
0323 .....
0324 .....
0325 .....
0326 .....
0327 .....
0328 .....
0329 .....
0330 .....
0331 .....
0332 .....
0333 .....
0334 .....
0335 .....
0336 .....
0337 .....
0338 .....
0339 .....
0340 .....
0341 .....
0342 .....
0343 .....
0344 .....
0345 .....
0346 .....
0347 .....
0348 .....
0349 .....
0350 .....
0351 .....
0352 .....
0353 .....
0354 .....
0355 .....
0356 .....
0357 .....
0358 .....
0359 .....
0360 .....
0361 .....
0362 .....
0363 .....
0364 .....
0365 .....
0366 .....
0367 .....
0368 .....
0369 .....
0370 .....
0371 .....
0372 .....
0373 .....
0374 .....
0375 .....
0376 .....
0377 .....
0378 .....
0379 .....
0380 .....

```

```

CCCCC .....
: Overwrite data array with calculated data, based on the
: just obtained coefficients.
CCCCC .....
0411 WRITE (LI, 105) NOCR
: READ (LI, *) IAR
: WRITE (LI, 149) (ICLR, I=1, 2)
: IF (IAR.LT.1 .OR. IAR.GT.2) GO TO 411
: IF (IAR.EQ.1) WRITE (LO, 604)
: IF (IAR.EQ.2) WRITE (LO, 644)
: WRITE (LO, 602) (J, J=1, NPITCH, 1)
DO 20 I=1, NMACH, 1
DO 19 J=1, NPITCH, 1
SUM=0.
DO 96 I1=1, M, 1
: IEXP1=I1-1
: IF (IEXP1.EQ.0) GO TO 401
X1EXP=X1(I, J)**IEXP1
GO TO 402
401 X1EXP=1
402 CONTINUE
DO 96 J1=1, N, 1
: IEXP2=J1-1
: IF (IEXP2.EQ.0) GO TO 403
X2EXP=X2(I, J)**IEXP2
GO TO 404
403 X2EXP=1
404 CONTINUE
96 SUM=SUM+COEFF(I1, J1)*X1EXP*X2EXP
GO TO (412, 413) IAR
412 R(J)=SUM-Y(I, J)
GO TO 19
413 R(J)=((SUM-Y(I, J))/Y(I, J))*100.
19 Y(I, J)=SUM
20 WRITE (6, 603) I, (R(J), J=1, NPITCH, 1)
WRITE (6, 602) (J, J=1, NPITCH, 1)
CCCCC .....
: Plot calculated calibration surface.
CCCCC .....
CALL SETSH (113, 3.)
DO 81 I=1, NMACH, 1
DO 81 J=1, NPITCH, 1
CALL THRTH (XPLT, YPLT, X1(I, J), X2(I, J), Y(I, J))
: IF (J.EQ.1) CALL PLOT (XPLT, YPLT, 2)
: IF (J.GT.1) CALL PLOT (XPLT, YPLT, 3)
81 CONTINUE
DO 82 J=1, NPITCH, 1
DO 82 I=1, NMACH, 1
CALL THRTH (XPLT, YPLT, X1(I, J), X2(I, J), Y(I, J))
: IF (I.EQ.1) CALL PLOT (XPLT, YPLT, 2)
: IF (I.GT.1) CALL PLOT (XPLT, YPLT, 3)
82 CONTINUE
CCCCC .....
: Redefine Y.
CCCCC .....
DO 802 I=1, NMACH, 1
DO 802 J=1, NPITCH, 1
802 Y(I, J)=YT(I, J)
CCCCC .....

```

PAGE 0006 LAB 3:03 PM FRI., 26 SEP., 1980

```
0381 C      : Next step ?
0382 C
0383 C
0384 C
0385      WRITE (LI,103) NOCR
0386      READ (LI, *) IDUM
0387      WRITE (LI, 149) (ICLR,I=1,2,1)
0388      IF (IDUM.EQ.1) GO TO 91
0389      IF (IDUM.EQ.2) GO TO 89
0390      CALL STOPC
0391      STOP 077
0392      89 WRITE (LI,104)
0393      READ (LI, *) ALPHAX,ALPHAY,XO,YO
0394      WRITE (LI, 149) ICLR
0395      GO TO 85
0396      END
```

FTN4 COMPILER: HP92060-16092 REV. 1926 (790430)

\*\* NO WARNINGS \*\* NO ERRORS \*\* PROGRAM = 04465

COMMON = 00000



### 3.6.2. Load map

LAB 10042 20622 Plot calibration points and calculate coefficients.

to load these program modules, enter (from LOADR): MS,XTPLBL

DATA3	20623	23622	3D-Approximation	/ DATA3 /
MAT3	23623	27455	3D-Approximation	arrange system matrix and vector.
MAT31	27456	27651	3D-Approximation	arrange submatrix 1. class.
S3	27652	30070	3D-Approximation	compute summations.
AB3	30071	41534	3D-Approximation	/ AB3 /

AFLD	41535	42534	GSP	/ AFLD / A(256)	
AXIS	42535	45054	GSP		draw and label axes.
INITC	45055	46067	GSP	read control array A;	initialize plotter.
PLOT	46070	46620	GSP		move pen to a defined point.
SETSM	46621	47561	GSP		change plotter modes.
STOPC	47562	47624	GSP		terminate graphics.
THRTW	47625	50111	GSP	converts 3D user coordinates to plotter units.	
FCR	50112	50113	GSP	/ FCTR / FX,FY	
FACTR	50116	50135	GSP		vary size of the plot.

LOCLU	50136	50213	92067-16268	REV.1903	790228
READF	50214	51155	92067-16125	REV.1940	790719
OPEN	51156	51453	92067-16125	REV.1903	790215
CLOSE	51454	51663	92067-16125	REV.1903	781229
CLRIO	51664	51672	750701	24998-16001	
OVRD.	51673	51673	92067-16125	REV.1903	780526
SSMVE	51674	51762	92067-16268	REV.1903	790202
LURQ	51763	52349	92067-16268	REV.1903	790223
.DADS	52346	52455	780818	24998-16001	
.DMP	52456	52623	780818	24998-16001	
.DDI	52624	53124	781021	24998-16001	
SESSN	53125	53142	92067-16125	REV.1903	780413
R/MS	53143	53301	92067-16125	REV.1903	781214
P.PAB	53302	53330	92067-16125	REV.1903	740801
.DNC	53331	53348	780818	24998-16001	
PAUSE	53349	53441	771122	24998-16001	
.SARRN	53442	53557	92067-16268	REV.1903	770715
FHTIO	53558	53566	24998-16002	REV.1926	790417
ERR0	53567	55146	771122	24998-16001	
TAN	55147	55253	780424	24998-16001	
ABS	55254	55262	750701	24998-16001	
.SNCS	55263	55424	780424	24998-16001	
.DDE	55425	55436	780818	24998-16001	
.DIN	55437	55444	780818	24998-16001	
.RTOI	55445	55540	780921	24998-16001	
.FPMR	55541	55602	781106	24998-16001	
.SBT	55603	55643	770518	24998-16001	
.FCH	55644	55660	750701	24998-16001	
PAU.E	55661	55661	750701	24998-16001	
ERS.E	55662	55662	750701	24998-16001	
.CHRS	55663	55746	780424	24998-16001	
SOPE	55747	56123	92067-16125	REV.1903	790103
RMUB	56124	56471	92067-16125	REV.1903	781003
RWUD	56472	56614	92067-16125	REV.1903	780801
FRMR	56615	62252	24998-16002	REV.1926	790503
FHT.E	62253	62253	24998-16002	REV.1901	781107
IBAA2	62254	63554	59310-1X013	REV.1940	790802 1153
REIO	63555	63701	92067-16268	REV.1903	790316
RMPAR	63702	63744	781106	24998-16001	
RNAME	63745	64012	771121	24998-16001	
LUTRU	64013	64121	92067-16268	REV.1903	790223
IPUT	64122	64142	92067-16125	REV.1903	740801
SETP	64143	64167	781106	24998-16001	
.CFR	64178	64245	750701	24998-16001	
.LBT	64246	64276	770518	24998-16001	

24 PAGES RELOCATED 24 PAGES REQ'D NO PAGES EMA NO PAGES MSEG  
 LINKS:BP PROGRAM:LB LOAD:TE COMMON:NC  
 /LOADR:LAB READY AT 3:10 PM FRI., 26 SEPT, 1980  
 /LOADR:END

### 3.6.3. Results

i) printed output

Thermal data from file CALIAA.

#### 1. Mach number

Gamma	Beta	Phi
-.818	-.005	-.33.000
-.519	-.005	-.33.000
-.280	-.005	-.33.000
-.182	-.005	-.33.000
-.103	-.006	-.33.000
-.059	-.006	-.33.500
-.018	-.006	-.33.000
-.019	-.006	-.33.500
-.054	-.006	-.33.000
-.133	-.006	-.33.000
-.197	-.006	-.33.000
-.346	-.006	-.33.000
-.605	-.006	-.33.000

#### 2. Mach number

Gamma	Beta	Phi
-.781	-.016	-.33.000
-.506	-.016	-.33.000
-.280	-.016	-.33.000
-.178	-.017	-.33.000
-.092	-.018	-.33.000
-.048	-.018	-.33.500
-.004	-.018	-.33.000
-.032	-.018	-.33.500
-.061	-.018	-.33.000
-.130	-.019	-.33.000
-.190	-.019	-.33.000
-.346	-.020	-.33.000
-.583	-.018	-.33.000

#### 3. Mach number

Gamma	Beta	Phi
-.787	-.024	-.33.000
-.510	-.024	-.33.000
-.276	-.024	-.33.000
-.172	-.026	-.33.000
-.085	-.027	-.33.000
-.050	-.027	-.33.500
-.003	-.027	-.33.000
-.037	-.027	-.33.500
-.066	-.027	-.33.000
-.131	-.027	-.33.000
-.191	-.029	-.33.000
-.345	-.029	-.33.000
-.592	-.027	-.33.000

#### 4. Mach number

Gamma	Beta	Phi
-.790	-.035	-.33.000
-.519	-.035	-.33.000
-.276	-.035	-.33.000
-.174	-.037	-.33.000
-.055	-.039	-.33.000
-.043	-.039	-.33.500
-.003	-.038	-.33.000
-.037	-.038	-.33.500
-.064	-.039	-.33.000
-.133	-.039	-.33.000
-.196	-.041	-.33.000
-.349	-.042	-.33.000
-.596	-.039	-.33.000

#### 5. Mach number

Gamma	Beta	Phi
-.795	-.045	-.000
-.531	-.045	-.000
-.282	-.045	-.000
-.175	-.045	-.000
-.082	-.050	-.000
-.040	-.050	-.500
-.010	-.050	-.000
-.034	-.049	-.500
-.067	-.049	-.000
-.134	-.050	-.000
-.198	-.053	-.000
-.250	-.055	-.000
-.593	-.050	-.000

6. Mach number

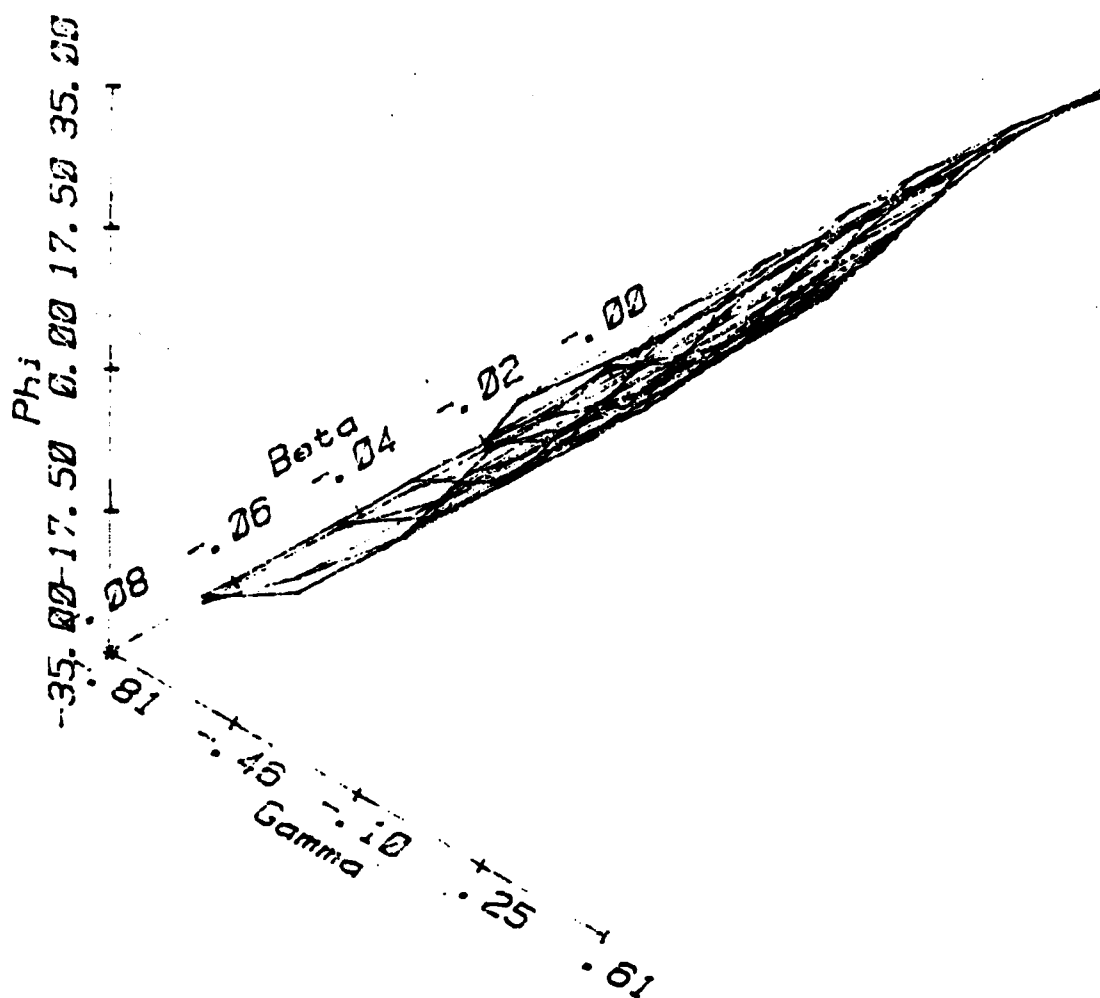
Gamma	Beta	Phi
-.803	-.065	-.000
-.528	-.065	-.000
-.284	-.065	-.000
-.173	-.068	-.000
-.085	-.071	-.000
-.032	-.072	-.500
-.004	-.071	-.000
-.024	-.071	-.500
-.058	-.071	-.000
-.131	-.071	-.000
-.199	-.075	-.000
-.349	-.078	-.000
-.603	-.073	-.000

X1	X2	Y
----	----	---



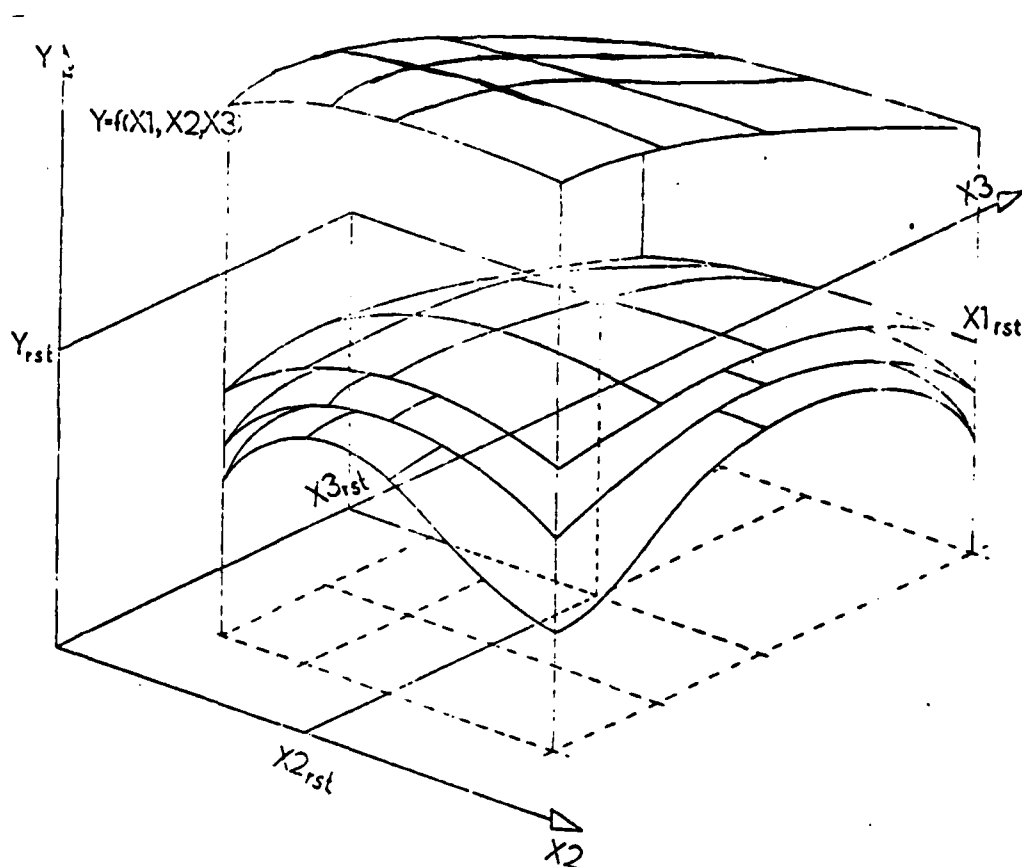


ii) graphic output



#### 4. FOUR DIMENSIONAL APPROXIMATION

##### 4.1. Problem:



A data set of  $NPNTS1 * NPNTS2 * NPNTS3$  data points is given, where  $Y$  depends on the parameters  $X_1$ ,  $X_2$  and  $X_3$ . The data pattern is to be approximated by a function  $Y = f(X_1, X_2, X_3)$ , so that the error between data points and analytically determined points is lower.

#### 4.2. Approach

Again, as in the two and three dimensional case, a polynomial method is used. For

$$Y = f(x_1, x_2, x_3)$$

we assume

$$\begin{aligned} Y = & \{ [C_{11} + C_{12}x_3 + \dots + C_{1N}x_3^{N-1}] + \\ & [C_{21} + C_{22}x_3 + \dots + C_{2N}x_3^{N-1}] \cdot x_2 + \\ & \dots \\ & + [C_{M1} + C_{M2}x_3 + \dots + C_{MN}x_3^{N-1}] \cdot x_2^{M-1} \} + \\ & + \{ [C_{21} + C_{22}x_3 + \dots + C_{2N}x_3^{N-1}] + \\ & [C_{221} + C_{222}x_3 + \dots + C_{22N}x_3^{N-1}] \cdot x_2 + \\ & \dots \\ & + [C_{2M1} + C_{2M2}x_3 + \dots + C_{2MN}x_3^{N-1}] \cdot x_2^{M-1} \} x_1 + \\ & \dots \\ & + \{ [C_{L1} + C_{L2}x_3 + \dots + C_{LN}x_3^{N-1}] + \\ & + [C_{L21} + C_{L22}x_3 + \dots + C_{L2N}x_3^{N-1}] \cdot x_2 + \\ & \dots \\ & + [C_{LM1} + C_{LM2}x_3 + \dots + C_{LMN}x_3^{N-1}] \cdot x_2^{M-1} \} x_1^{L-1} \end{aligned}$$

or

$$Y = \sum_{i=1}^L \left\{ \sum_{j=1}^M \left\{ \sum_{k=1}^N C_{ijk} x_3^{k-1} \right\} x_2^{j-1} \right\} x_1^{i-1} \quad (4.1.)$$



This data arrangement which discriminates between data points taken at constant  $X_1$ ,  $X_2$  and  $X_3$ , respectively is needed in the application of this method to the description of pneumatic-velocity probe characteristics. The various surfaces in the above sketch for example can be seen as data taken at different Mach number ( $X_1$ ). At each constant Mach number ( $X_1 = \text{constant}$ ) yaw angle ( $X_2$ ) and pitch angle ( $X_3$ ) are varied, and the probe is measured to be  $Y$ . Even though probes are usually balanced to the average flow yaw angle in some cases this is not possible and the four dimensional approximation is then needed. A similar method for describing probe calibration data is used at the institute for Jet Engines and Turbomachines at the Technical University Aachen (Ref. 2).

The least squares criterion is then

$$R = \sum_{rst} \left\{ \sum_{s=1}^{NPN12} \left\{ \sum_{t=1}^{NPN123} [f(X_{1rst}, X_{2rst}, X_{3rst}) - Y_{rst}]^2 \right\} \right\} \quad (4.2.)$$

where the indices rst denote data points. Using equation (4.1.),  $R$  becomes

$$R = \sum_{rst} \left\{ \sum_{s=1}^{NPN12} \left\{ \sum_{t=1}^{NPN123} \left[ c_{111} + c_{112} X_{3rst}^{N-1} + \dots + c_{11N} X_{3rst}^{N-1} + \right. \right. \right. \\ \left. \left. + (c_{121} + c_{122} X_{3rst}^{N-1} + \dots + c_{12N} X_{3rst}^{N-1}) \cdot X_{2rst} + \right. \right. \\ \left. \left. \dots \right. \right. \\ \left. \left. + (c_{1M1} + c_{1M2} X_{3rst}^{N-1} + \dots + c_{1MN} X_{3rst}^{N-1}) \cdot X_{2rst}^{M-1} + \right. \right. \\ \left. \left. + (c_{211} + c_{212} X_{3rst}^{N-1} + \dots + c_{21N} X_{3rst}^{N-1}) \cdot \right. \right. \\ \left. \left. + (c_{221} + c_{222} X_{3rst}^{N-1} + \dots + c_{22N} X_{3rst}^{N-1}) \cdot X_{2rst} + \right. \right. \\ \left. \left. \dots \right. \right. \\ \left. \left. + (c_{2M1} + c_{2M2} X_{3rst}^{N-1} + \dots + c_{2MN} X_{3rst}^{N-1}) \cdot X_{2rst}^{M-1} \right) \cdot X_{1rst} \right. \\ \left. \dots \right\}$$

$$\begin{aligned}
& + \langle C_{L11} + C_{L12} X_{rst}^3 + \dots + C_{L1N} X_{rst}^{N-1} + \\
& (C_{L21} + C_{L22} X_{rst}^3 + \dots + C_{L2N} X_{rst}^{N-1}) \cdot X_{rst}^2 + \\
& \dots \\
& (C_{LM1} + C_{LM2} X_{rst}^3 + \dots + C_{LMN} X_{rst}^{N-1}) \cdot X_{rst}^{M-1} \cdot X_{rst}^{L-1} ] \} \}
\end{aligned}$$

The term in the [ ] - bracket shall be called B

$$R = \sum_{r=1}^{N^2-1} \left\{ \sum_{s=1}^{N^2-1} \left\{ \sum_{t=1}^{N^2-1} B^2 \right\} \right\}$$

#### 4.3. Solution:

To minimize the error, R is partially differentiated with respect to the coefficients  $C_{ijk}$  and the partial derivatives are set to zero, thus:

$$\frac{\partial R}{\partial C_{ijk}} \stackrel{!}{=} 0 \quad i=1 \dots L; j=1 \dots M; k=1 \dots N$$

$$\frac{\partial R}{\partial C_{ijk}} = \sum_{r=1}^{NPNR1} \left\{ \sum_{s=1}^{NPNR2} \left\{ \sum_{t=1}^{NPNR3} 2B \cdot \frac{\partial B}{\partial C_{ijk}} \right\} \right\} = 0$$

Assuming, that the summations extend over all data points,  $\Sigma\Sigma\Sigma$  should be understood to be  $\sum_{r=1}^{NPNR1} \sum_{s=1}^{NPNR2} \sum_{t=1}^{NPNR3}$ . Now even a very large sheet of paper can't take the system matrix, unless we skip to submatrix notation.

$$\begin{vmatrix} A_{11}^* & A_{12}^* & \dots & A_{1L}^* \\ A_{21}^* & A_{22}^* & \dots & A_{2L}^* \\ \dots & \dots & \dots & \dots \\ A_{L1}^* & A_{L2}^* & \dots & A_{LL}^* \end{vmatrix} \begin{vmatrix} C_1^* \\ C_2^* \\ \dots \\ C_L^* \end{vmatrix} = \begin{vmatrix} B_1^* \\ B_2^* \\ \dots \\ B_L^* \end{vmatrix} \quad (4.3.)$$

The asterisk indicates a submatrix 1. class. On the following pages the submatrices and subvectors 1. class are listed.



$\Sigma \Sigma x_{m1}^1 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^1 \cdot x_{m1}^{3^{n-1}}$	...	$\Sigma \Sigma \Sigma x_{m1}^{n-1}$	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^{3^{n-1}}$
$\Sigma \Sigma x_{m1}^1 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^1 \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^3$	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^3$
$\Sigma \Sigma x_{m1}^1 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^1 \cdot x_{m1}^{3^{n-2}}$		$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^{3^{n-2}}$	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{n-1} \cdot x_{m1}^{3^{n-1}}$
$\Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^{3^{n-1}}$	...	$\Sigma \Sigma \Sigma x_{m1}^3$	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^{3^{n-1}}$
$\Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$
$\Sigma \Sigma x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^{3^{n-2}}$		$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^{3^{n-2}}$	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^3 \cdot x_{m1}^{3^{n-1}}$
...							
$\Sigma \Sigma x_{m1}^n \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^n \cdot x_{m1}^{3^{n-1}}$	...	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}}$	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^{3^{n-1}}$
$\Sigma \Sigma x_{m1}^n \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^n \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^3$	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^3$
$\Sigma \Sigma x_{m1}^n \cdot x_{m1}^3$		$\Sigma \Sigma \Sigma x_{m1}^n \cdot x_{m1}^{3^{n-2}}$		$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^{3^{n-1}}$	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^3$	...	$\Sigma \Sigma \Sigma x_{m1}^{3^{n-2}} \cdot x_{m1}^{3^{n-1}}$





[illegible]







2



...

...

$$\begin{array}{lll}
 \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+2} & \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+2} & \dots \quad \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+4} x_{\text{rel}}^{2+3} \\
 \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+2} & \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+1} & \dots \quad \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+1} \\
 \dots & \vdots & \dots \\
 \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+1} & \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+1} & \dots \quad \Sigma \Sigma \Sigma x_{\text{rel}}^{2-1} x_{\text{rel}}^{2+3} x_{\text{rel}}^{2+3}
 \end{array}$$

$\sum \sum \sum Y_{m1}$
$\sum \sum \sum Y_{m1} \cdot X_{3m1}$
...
$\sum \sum \sum Y_{m1} \cdot X_{3m1}^{m-1}$
$\sum \sum \sum Y_{m1} \cdot X_{2m1}$
$\sum \sum \sum Y_{m1} \cdot X_{2m1} \cdot X_{3m1}$
...
$\sum \sum \sum Y_{m1} \cdot X_{2m1} \cdot X_{3m1}^{m-1}$

$B_1^4 =$

$c_{m1}$
$c_{m2}$
...
$c_{m1}^{m-1}$
$c_{m1}$
$c_{m2}$
...
$c_{m1}^{m-1}$

$C_1^4 =$

$\sum \sum \sum Y_{m1} \cdot X_{1m1}$
$\sum \sum \sum Y_{m1} \cdot X_{1m1} \cdot X_{3m1}$
...
$\sum \sum \sum Y_{m1} \cdot X_{1m1} \cdot X_{3m1}^{m-1}$
$\sum \sum \sum Y_{m1} \cdot X_{1m1} \cdot X_{2m1}$
$\sum \sum \sum Y_{m1} \cdot X_{1m1} \cdot X_{2m1} \cdot X_{3m1}$
...
$\sum \sum \sum Y_{m1} \cdot X_{1m1} \cdot X_{2m1} \cdot X_{3m1}^{m-1}$

$B_2^4 =$

$c_{21}$
$c_{22}$
...
$c_{21}^{m-1}$
$c_{21}$
$c_{22}$
...
$c_{21}^{m-1}$

$C_2^4 =$



#### 4.4. Structure of Equation System:

##### 4.4.1. System Matrix A

It can be seen, that all submatrices 1. class on the same diagonal bands 2. order of the system matrix are identical and therefore will be renamed from  $A_{1,j}^*$  to  $A_k^*$

$$A_1^* = A_{1,1}^*$$

$$A_2^* = A_{3,1}^* = A_{1,2}^*$$

$$A_3^* = A_{3,1}^* = A_{2,2}^* = A_{1,3}^*$$

...

$$A_L^* = A_{L,1}^* = A_{L-1,2}^* = A_{L-2,3}^* = \dots = A_{2,L-1}^* = A_{1,L}^*$$

$$A_{L+1}^* = A_{L,2}^* = A_{L-1,3}^* = \dots = A_{3,L-1}^* = A_{2,L}^*$$

$$A_{L+2}^* = A_{L,3}^* = \dots = A_{4,L-1}^* = A_{3,L}^*$$

...

$$A_{2L-1}^* = A_{L,L}^*$$

The submatrices 1. class  $A_k^*$  ( $k=1, \dots, 2L-1$ ) themselves are divided up into submatrices 2. class.

$$A_k^* = \begin{vmatrix} A_{k,1,1}^{**} & A_{k,1,2}^{**} & \dots & A_{k,1,M}^{**} \\ A_{k,2,1}^{**} & A_{k,2,2}^{**} & \dots & A_{k,2,M}^{**} \\ \dots & \dots & \dots & \dots \\ A_{k,M,1}^{**} & A_{k,M,2}^{**} & \dots & A_{k,M,M}^{**} \end{vmatrix}$$



The double asterisk indicates a submatrix 2. class. All submatrices

2. class on the same diagonal bands 2. order of the submatrices 1. class

$A_k^*$  ( $k = 1, \dots, 2L$ ) are identical and therefore will be renamed from

$A_{k;1,j}^{**}$  to  $A_{k;1}^{**}$

$$A_{k;1}^{**} = A_{k;1}^{**}$$

$$A_{k;2}^{**} = A_{k;2,1}^{**} = A_{k;1,2}^{**}$$

$$A_{k;3}^{**} = A_{k;3,1}^{**} = A_{k;2,2}^{**} = A_{k;1,3}^{**}$$

...

$$A_{k;M}^{**} = A_{k;M,1}^{**} = A_{k;M-1,2}^{**} = A_{k;M-2,3}^{**} = \dots = A_{k;2,M-1}^{**} = A_{k;1,M}^{**}$$

$$A_{k;M+1}^{**} = A_{k;M,2}^{**} = A_{k;M-1,3}^{**} = \dots = A_{k;3,M-1}^{**} = A_{k;2,M}^{**}$$

$$A_{k;M+2}^{**} = A_{k;M,3}^{**} = \dots = A_{k;4,M-1}^{**} = A_{k;3,M}^{**}$$

...

$$A_{k;2M-1}^{**} = A_{k;1,M}^{**}$$

where  $A_{k;1}^{**}$  can be written as

$$A_{k;l}^{**} = \begin{vmatrix} \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^2 & \dots & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^{2M-1} \\ \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^2 & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^3 & \dots & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^N \\ \dots & \dots & \dots & \dots \\ \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^{M-1} & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^N & \dots & \sum \sum \sum x_{nt}^{l-1} \cdot x_{nt}^{l-1} \cdot x_{nt}^{2M-2} \end{vmatrix}$$

$$k = 1, \dots, 2L-1; l = 1, \dots, 2M-1$$

Note, that this submatrix 2. class is not only symmetrical, but also the elements on each diagonal line 2. order are identical. The elements  $a_{k;l;i,j}$  of the submatrix 2. class  $A_{k;l}^{**}$  therefore are renamed to  $a_{k;l;m}$ ;  $L$  specifies the place of the submatrix 2. class  $A_{k;l}^{**}$  in the submatrix 1. class  $A_k^*$ ,  $k$  specifies the place of the submatrix 1. class  $A_k^*$  in the system matrix  $A$  and  $i, j$  specifies the place of the element  $a_{k;l;i,j}$  in the submatrix 2. class  $A_{k;l}^{**}$

$$a_{k;l,1} = a_{k,l,1,1}$$

$$a_{k;l,2} = a_{k,l,2,1} = a_{k,l,1,2}$$

$$a_{k;l,3} = a_{k,l,3,1} = a_{k,l,2,2} = a_{k,l,1,3}$$

...

$$a_{k;l,N} = a_{k,l,N,1} = a_{k,l,N-1,2} = a_{k,l,N-2,3} = \dots = a_{k,l,2,N-1} = a_{k,l,1,N}$$

$$a_{k,l,N+1} = a_{k,l,N,2} = a_{k,l,N-1,3} = \dots = a_{k,l,3,N-1} = a_{k,l,2,N}$$

$$a_{k,l,N+2} = a_{k,l,N,3} = \dots = a_{k,l,4,N-1} = a_{k,l,3,N}$$

...

$$a_{k,l,2N-1} = a_{k,l,N,N}$$

where  $a_{k,l,m}$  can be written as

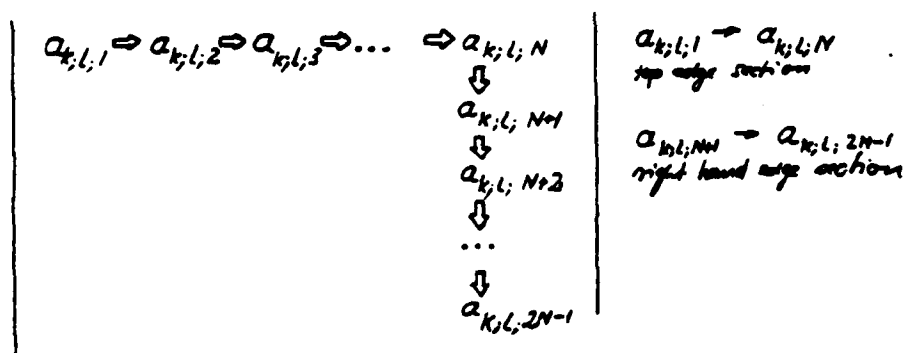
$$a_{k,l,m} = \sum_{r=1}^{N+L-1} \left\{ \sum_{s=1}^{N+M-1} \left\{ \sum_{t=1}^{N+M-1} X_1^{k-1} \cdot X_2^{L-1} \cdot X_3^{m-1} \right\} \right\} \quad (4.4.)$$

$$k = 1, \dots, 2L-1; \quad L = 1, \dots, 2M-1; \quad m = 1, \dots, 2N-1$$

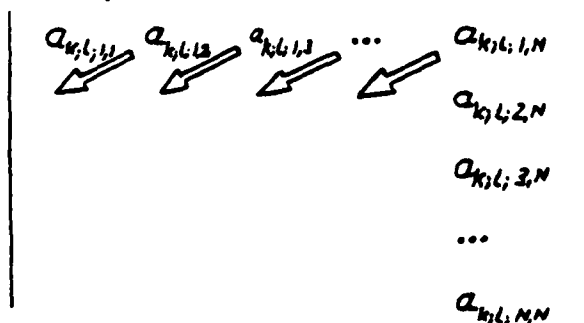
This operation is programmed in REAL FUNCTION S4 (NPNTS1, NPNTS2, NPNTS3, IPOWR1, IPOWR2, IPOWR3, IY). The data X1, X2, X3 and Y are known to this function through a common block named DATA4. So only NPNTS1, NPNTS2, NPNTS3, IPOWR1 (= k-1), IPOWR2 (= L-1), IPOWR3 (= m-1) and IY (=0) have to be passed to the function and the value of  $a_{k;l;m}$  is returned through the function name S4.

SUBROUTINE MAT42 presets the submatrices 2. class  $A_{k,l}^{**}$  in the following way

- i) Preset edge section elements (using REAL FUNCTION S4)



- ii) Copy defined elements diagonally



$$\begin{array}{cccccc}
 a_{k,l,1,1} & a_{k,l,1,2} & a_{k,l,1,3} & \dots & a_{k,l,1,N} \\
 \swarrow & \swarrow & \swarrow & \swarrow & \\
 a_{k,l,2,1} & a_{k,l,2,2} & a_{k,l,2,3} & \dots & a_{k,l,2,N} \\
 & & & & \swarrow \\
 & & & & a_{k,l,3,N} \\
 & & & & \dots \\
 & & & & a_{k,l,N,N}
 \end{array}$$

$$\begin{array}{cccccc}
 a_{k,l,1,1} & a_{k,l,1,2} & a_{k,l,1,3} & \dots & a_{k,l,1,N} \\
 a_{k,l,2,1} & a_{k,l,2,2} & a_{k,l,2,3} & \dots & a_{k,l,2,N} \\
 \swarrow & \swarrow & \swarrow & \swarrow & \\
 a_{k,l,3,1} & a_{k,l,3,2} & a_{k,l,3,3} & \dots & a_{k,l,3,N} \\
 & & & & \dots \\
 & & & & a_{k,l,N,N}
 \end{array}$$

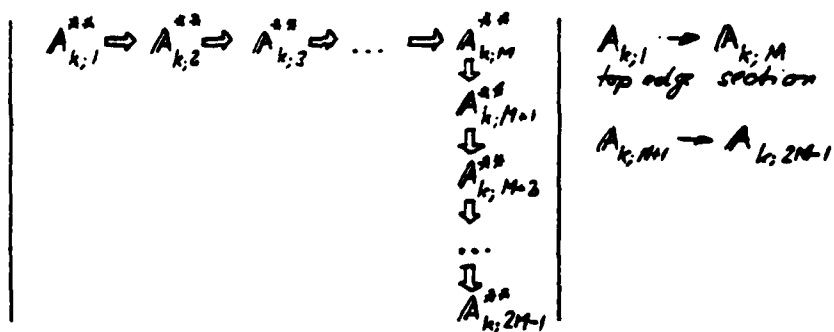
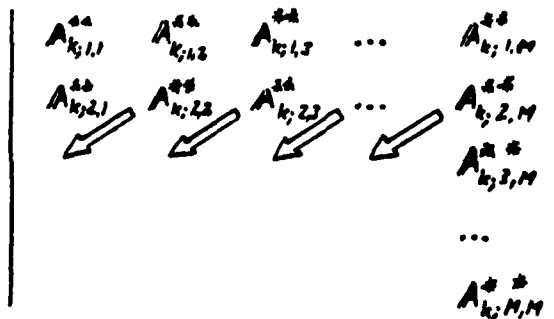
$$\begin{array}{cccccc}
 a_{k,l,1,1} & a_{k,l,1,2} & a_{k,l,1,3} & \dots & a_{k,l,1,N} \\
 a_{k,l,2,1} & a_{k,l,2,2} & a_{k,l,2,3} & \dots & a_{k,l,2,N} \\
 a_{k,l,3,1} & a_{k,l,3,2} & a_{k,l,3,3} & \dots & a_{k,l,3,N} \\
 \dots & \dots & \dots & \dots & \dots \\
 \swarrow & \swarrow & \swarrow & \swarrow & \\
 & & & & a_{k,l,N,N}
 \end{array}$$

$$A_{k,l}^{*}$$

$$\begin{array}{cccccc}
 a_{k,l,1,1} & a_{k,l,1,2} & a_{k,l,1,3} & \dots & a_{k,l,1,N} \\
 a_{k,l,2,1} & a_{k,l,2,2} & a_{k,l,2,3} & \dots & a_{k,l,2,N} \\
 a_{k,l,3,1} & a_{k,l,3,2} & a_{k,l,3,3} & \dots & a_{k,l,3,N} \\
 \dots & \dots & \dots & \dots & \dots \\
 a_{k,l,N,1} & a_{k,l,N,2} & a_{k,l,N,3} & \dots & a_{k,l,N,N}
 \end{array}$$

A<sup>\*\*</sup><sub>k:L</sub> (SUBM2(4,4)) to SUBROUTINE MAT41, which foresets the submatrices

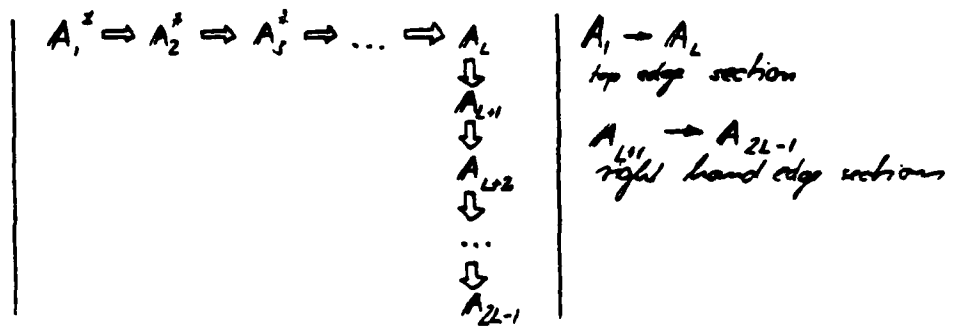
1) Preset edge section submatrices (using SUBROUTINE MAT42)


$$\begin{array}{ccccccc}
 A_{k;1,1}^{1,1} & A_{k;1,2}^{1,2} & A_{k;1,3}^{1,3} & \dots & A_{k;1,M}^{1,M} \\
 A_{k;2,1}^{2,1} & & & & A_{k;2,M}^{2,M} \\
 A_{k;3,1}^{3,1} & & & & A_{k;3,M}^{3,M} \\
 & & & & \dots \\
 & & & & A_{k;M,M}^{M,M}
 \end{array}$$


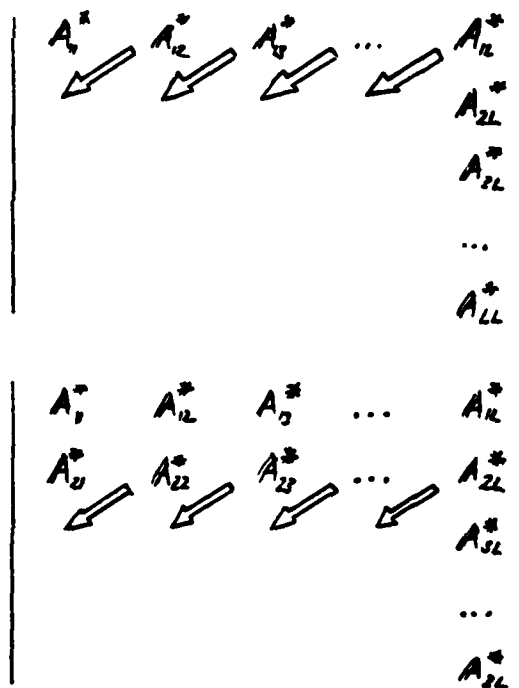
$$\begin{array}{c}
 \begin{array}{c}
 A_{k,1,1}^{**} \quad A_{k,1,2}^{**} \quad A_{k,1,3}^{**} \quad \dots \quad A_{k,1,M}^{**} \\
 A_{k,2,1}^{**} \quad A_{k,2,2}^{**} \quad A_{k,2,3}^{**} \quad \dots \quad A_{k,2,M}^{**} \\
 A_{k,3,1}^{**} \quad A_{k,3,2}^{**} \quad A_{k,3,3}^{**} \quad \dots \quad A_{k,3,M}^{**} \\
 \swarrow \quad \swarrow \quad \swarrow \quad \swarrow \quad \dots \\
 \dots \quad \dots \quad \dots \quad \dots \quad A_{k,M,M}^{**}
 \end{array} \\
 \\
 \begin{array}{c}
 A_{k,1,1}^{**} \quad A_{k,1,2}^{**} \quad A_{k,1,3}^{**} \quad \dots \quad A_{k,1,M}^{**} \\
 A_{k,2,1}^{**} \quad A_{k,2,2}^{**} \quad A_{k,2,3}^{**} \quad \dots \quad A_{k,2,M}^{**} \\
 A_{k,3,1}^{**} \quad A_{k,3,2}^{**} \quad A_{k,3,3}^{**} \quad \dots \quad A_{k,3,M}^{**} \\
 \dots \quad \dots \quad \dots \quad \dots \quad \dots \\
 \swarrow \quad \swarrow \quad \swarrow \quad \swarrow \quad \dots \\
 \dots \quad \dots \quad \dots \quad \dots \quad A_{k,M,M}^{**}
 \end{array} \\
 \\
 A_k^* = \begin{array}{c}
 \begin{array}{c}
 A_{k,1,1}^{**} \quad A_{k,1,2}^{**} \quad A_{k,1,3}^{**} \quad \dots \quad A_{k,1,M}^{**} \\
 A_{k,2,1}^{**} \quad A_{k,2,2}^{**} \quad A_{k,2,3}^{**} \quad \dots \quad A_{k,2,M}^{**} \\
 A_{k,3,1}^{**} \quad A_{k,3,2}^{**} \quad A_{k,3,3}^{**} \quad \dots \quad A_{k,3,M}^{**} \\
 \dots \quad \dots \quad \dots \quad \dots \quad \dots \\
 A_{k,M,1}^{**} \quad A_{k,M,2}^{**} \quad A_{k,M,3}^{**} \quad \dots \quad A_{k,M,M}^{**}
 \end{array}
 \end{array}
 \end{array}$$

This subroutine, that needs NPNTS1, NPNTS2, NPNTS3, M, N, IPOWR1 (=L-1) as input parameters, returns the subroutine 1st class  $A_k^*$  (SUBM1(16,16)) to SUBROUTINE MAT4, which presets the entire system matrix A in the following way.

i) Preset edge section submatrices (using SUBROUTINE MAT41)



ii) Copy defined submatrices diagonally



$$\begin{array}{ccccc}
 A_{11}^* & A_{12}^* & A_{13}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & A_{23}^* & \dots & A_{2L}^* \\
 A_{31}^* & A_{32}^* & A_{33}^* & \dots & A_{3L}^* \\
 & \searrow & \searrow & \searrow & \searrow \\
 & & & & \dots \\
 & & & & A_{LL}
 \end{array}$$

$$\begin{array}{ccccc}
 A_{11}^* & A_{12}^* & A_{13}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & A_{23}^* & \dots & A_{2L}^* \\
 A_{31}^* & A_{32}^* & A_{33}^* & \dots & A_{3L}^* \\
 \dots & \dots & \dots & \dots & \dots \\
 \searrow & \searrow & \searrow & \searrow & \searrow
 \end{array}$$

$$A = \begin{array}{ccccc}
 A_{11}^* & A_{12}^* & A_{13}^* & \dots & A_{1L}^* \\
 A_{21}^* & A_{22}^* & A_{23}^* & \dots & A_{2L}^* \\
 A_{31}^* & A_{32}^* & A_{33}^* & \dots & A_{3L}^* \\
 \dots & \dots & \dots & \dots & \dots \\
 A_{L1} & A_{L2} & A_{L3} & \dots & A_{LL}
 \end{array}$$



#### 4.4.2. Right Hand Side Vector B:

The right hand side subvectors 1. class  $B_k^*$  generally can be written as

$$\begin{array}{l}
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X3_{rst} \\
 \dots \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X3_{rst}^{N-1} \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst} \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst} \cdot X3_{rst} \\
 \dots \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst} \cdot X3_{rst}^{N-1} \\
 \\
 \dots \\
 \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{M-1} \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{M-1} \cdot X3_{rst} \\
 \dots \\
 \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{M-1} \cdot X3_{rst}^{N-1}
 \end{array}$$

$k = 1, \dots, L$

The subvectors 1. class  $B_k^*$  ( $k = 1, \dots, L$ ) themselves will be divided up into subvectors 2. class:

$$B_k^* = \begin{pmatrix} B_{k,1}^{**} \\ B_{k,2}^{**} \\ \dots \\ B_{k,L}^{**} \end{pmatrix}$$

where subvectors 2. class  $B_{k,L}^{**}$  can be written as

$$B_{k,L}^{**} = \begin{pmatrix} \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{L-1} \\ \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{L-1} \cdot X3 \\ \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{L-1} \cdot X3^{N-1} \end{pmatrix}$$

$$k = 1, \dots, L; L = 1, \dots, M; m = 1, \dots, N$$

Finally, the elements of the subvectors 2. class  $B_{k,1}$  can be written as

$$b_{k,L;m} = \sum \sum \sum Y_{rst} \cdot X1_{rst}^{k-1} \cdot X2_{rst}^{L-1} \cdot X3_{rst}^{m-1}$$

$$k = 1, \dots, L; L = 1, \dots, M; m = 1, \dots, N$$

To calculate  $b_{k,1;m}$ , REAL FUNCTION S4 is used, again. Data X1, X2, X3 and Y are available through COMMON block DATA4. NPNTS1, NPNTS2, NPNTS3, IPOWR1 (= k-1), IPOWR2 (= L-1) IPOWR3 (= m-1) and IY (=1) have to be passed to the function and the value of  $b_{k,1;m}$  is returned through the function name S4. Using FORTRAN programming language, the allocation of  $b_{k,1;m}$  can be performed by three stacked DO-loops.

```
I=0
DO 07 L1=1,L,1
IPOWER1=L1-1
DO 08 M1=1,M,1
IPOWER2=M1-1
DO 08 N1=1,N,1
IPOWER3=N1-1
I=I+1
08 B(I)=S4(NPNTS1,NPNTS2,NPNTS3,IPOWER1,IPOWER2,IPOWER3,1)
```

#### 4.5. Software:

The software to compute the coefficients for a two dimensional approximation is described in APPENDIX A and is implemented in the TPL HP=21MX computer system. To work correctly with these program modules, the user has to conform to the following conventions.

- 1) Provide the data in four arrays  
(Type: REAL) of 5\*5\*5\* elements  
through a COMMON block, named DATA4.

```
COMMON / DATA4 / X1,X2,X3,Y  
REAL X1(5,5,5),X2(5,5,5),X3(5,5,5),Y(5,5,5)
```

- 1i) Dimension an array (Type: REAL) of  
5\*5\*5 elements to contain the coefficients,

```
REAL COEF (5,5,5)
```

- 1ii) Define the parameters NPNTS1, NPNTS2,

```
NPNTS3,L,M,N and IPRINT
```

```
NPNTS1... # of X1 variations  
L ≤ NPNTS1 ≤ 5
```

```
NPNTS2... # of X2 variations  
M ≤ NPNTS2 ≤ 5
```

```
NPNTS3... # of X3 variations  
N ≤ NPNTS3 ≤ 5
```

```
L    ... (desired order of approximation  
         polynomial w.r.t. X1) + 1  
1 ≤ L ≤ 4
```

```
M    ... (desired order of approximation  
         polynomial w.r.t. X2) + 1  
1 ≤ M ≤ 4
```

```
N    ... (desired order of approximation  
         polynomial w.r.t. X3) + 1  
1 ≤ N ≤ 4
```

IPRINT ... Controls quantity of print out  
 2 ... Print system matrix and right  
       hand side vector before and  
       after Gauss Jordan Elimination.  
 1 ... Print equation system after  
       Gauss Jordan Elimination  
 <0... No print out  
 >0... Print equation (4.1.) with  
       the actual parameters.

- iv) If a user program uses subroutine MAT4, the software modules have to be loaded using the procedure in section 1.4.iv.

If all these requirements are met, the correct call for the subroutine is:

CALL MAT4 (NPNTS1,NPNTS2,NPNTS3,L,M,N,COEF,IPRINT)

Upon completed execution of this approximation routine array COEF contains the coefficients. Externals used by MAT4: AB4, DATA4, MAT41, MAT42, S4. Under no circumstances may the user use any of these names for a modules of his own user program.

#### 4.6. Sample User Program

##### 4.6.1 FTN4 Compiler listing of sample user program

```

0001 FTN4,L
0002 PROGRAM FITR (3,99)
0003 C
0004 APPROXIMATION PROGRAM TO FIT A SURFACE THROUGH DATA POINTS Y,
0005 WHICH DEPEND ON THREE INDEPENDENT PARAMETERS X1, X2 AND X3.
0006 C
0007 AUTHOR: HANS ZEBNER
0008 DATE : AUGUST 26, 1980
0009 C
0010 * , Hans Zebner: 4D-Approximation;  $Y=f(X1,X2,X3)$ .
0011 C
0012 COMMON / DATA4 / X1,X2,X3,Y
0013 COMMON / AB4 / A,B
0014 C
0015 REAL X1(5,5,5),X2(5,5,5),X3(5,5,5),Y(5,5,5)
0016 REAL A(64,64),B(64)
0017 C
0018 REAL COEF(5,5,5)
0019 INTEGER IDC8(144),IFILE(3),NOCR(2),ICLR(3)
0020 C
0021 DATA NOCR /000033B,040433B/, ICLR /015524B,015515B,006537B/
0022 C
0023 101 FORMAT (" ENTER NPNTS1 NPNTS2 NPNTS3 "2A2
0024 *)
0025 102 FORMAT (" ENTER L M N "2A2)
0026 103 FORMAT (" ENTER DATA FILE NAME "2A2)
0027 104 FORMAT (3A2)
0028 131 FORMAT (" ENTER IPRINT "2A2)
0029 132 FORMAT (" OUTPUT RAW DATA YES OR NO "2A2)
0030 105 FORMAT (/ " X1=")
0031 106 FORMAT (/ " X2=")
0032 107 FORMAT (3X,12I10)
0033 108 FORMAT (1X,12F10.3)
0034 109 FORMAT (/ " X3=")
0035 110 FORMAT (/ " Y=")
0036 111 FORMAT (/ " ")
0037 149 FORMAT ("=((3A2)))
0038 1101 FORMAT ("HANNES")
0039 C
0040 READ RAW DATA FROM FILE
0041 C
0042 01 WRITE (1,101) NOCR
0043 READ (1,*) NPNTS1,NPNTS2,NPNTS3
0044 WRITE (1,149) ICLR
0045 NMAX=5
0046 IF (NPNTS1.GT.NMAX) GO TO 01
0047 IF (NPNTS2.GT.NMAX) GO TO 01
0048 IF (NPNTS3.GT.NMAX) GO TO 01
0049 02 WRITE (1,102) NOCR
0050 READ (1,*) L,M,N
0051 WRITE (1,149) ICLR
0052 IF (L*M*N.GT.64) GO TO 02
0053 03 WRITE (1,103) NOCR
0054 READ (1,104) IFILE
0055 WRITE (1,149) ICLR
0056 31 WRITE (1,131) NOCR
0057 READ (1,*) IPRINT
0058 WRITE (1,149) ICLR
0059 WRITE (1,132) NOCR
0060 READ (1,104) IDUM
0061 WRITE (1,149) ICLR
0062 CALL OPEN (IDCB,IERR,IFILE)
0063 IF (IERR.LT.0) GO TO 03
0064 IL=2*NPNTS1*NPNTS2*NPNTS3
0065 CALL READF (IDCB,IERR,X1,IL,LEN,1)
0066 IF (IERR.LT.0) STOP 0001
0067 CALL READF (IDCB,IERR,X2,IL,LEN,54)
0068 IF (IERR.LT.0) STOP 0002
0069 CALL READF (IDCB,IERR,X3,IL,LEN,107)
0070 IF (IERR.LT.0) STOP 0003
0071 CALL READF (IDCB,IERR,Y,IL,LEN,160)
0072 IF (IERR.LT.0) STOP 0004
0073 CALL CLOSE (IDCB,IERR)
0074 IF (IERR.LT.0) STOP 0005
0075 IF (IDUM.NE.2HYES) GO TO 17
0076 WRITE (6,105)

```

PAGE 0002 FITR 3:47 PM FRI., 26 SEP., 1980

```

0077      DO 04 K1=1, NPNTS1,1
0078      WRITE (6,106) K1
0079      WRITE (6,107) (K3,K3=1, NPNTS3,1)
0080      DO 04 K2=1, NPNTS2,1
0081      04 WRITE (6,108) K2, (X1(K1,K2,K3),K3=1, NPNTS3,1)
0082      WRITE (6,109)
0083      DO 05 K1=1, NPNTS1,1
0084      WRITE (6,106) K1
0085      WRITE (6,107) (K3,K3=1, NPNTS3,1)
0086      DO 05 K2=1, NPNTS2,1
0087      05 WRITE (6,108) K2, (X2(K1,K2,K3),K3=1, NPNTS3,1)
0088      WRITE (6,110)
0089      DO 06 K1=1, NPNTS1,1
0090      WRITE (6,106) K1
0091      WRITE (6,107) (K3,K3=1, NPNTS3,1)
0092      DO 06 K2=1, NPNTS2,1
0093      06 WRITE (6,108) K2, (X3(K1,K2,K3),K3=1, NPNTS3,1)
0094      WRITE (6,111)
0095      DO 07 K1=1, NPNTS1,1
0096      WRITE (6,106) K1
0097      WRITE (6,107) (K3,K3=1, NPNTS3,1)
0098      DO 07 K2=1, NPNTS2,1
0099      07 WRITE (6,108) K2, (Y(K1,K2,K3),K3=1, NPNTS3,1)
0100      17 CALL MAT4 (NPNTS1, NPNTS2, NPNTS3, L, H, N, COEF, IPRINT)
0101      CALL CODE
0102      WRITE (IFILE,1101)
0103      CALL OPEN (IDCB, IERR, IFILE, IOPTN, 0, 26, 144)
0104      IF (IERR.LT.0) STOP 0006
0105      CALL WRITE (IDCB, IERR, COEF, 250)
0106      IF (IERR.LT.0) STOP 0007
0107      CALL CLOSE (IDCB, IERR)
0108      IF (IERR.LT.0) STOP 0010
0109      STOP 0077
0110      END

```

, FTN4 COMPILER: HP92060-16092 REV. 1926 (790430)

\*\* NO WARNINGS \*\* NO ERRORS \*\* PROGRAM = 01259

COMMON = 00000

#### 4.6.2. Load map of sample user program FITR

FITR 10042 12414 Hans Zebner: 4D-Approximation:  $Y=f(X1,X2,X3)$ .

to load these program modules, enter (from LOADR): MS,STPLBL

DATA4	12415	14364	4D-Approximation	/ DATA4 /
MAT4	14365	20251	4D-Approximation	arrange system matrix and vector.
MAT41	20252	20666	4D-Approximation	arrange submatrix 1. class.
MAT42	20667	21066	4D-Approximation	arrange submatrix 2. class.
S4	21067	21370	4D-Approximation	compute summations.
AB4	21371	41570	4D-Approximation	/ AB4 /

LOCLU	41571	41646	92067-16268	REV.1903	790228
READP	41647	42610	92067-16125	REV.1940	790719
OPEN	42611	43106	92067-16125	REV.1903	790215
CLOSE	43107	43316	92067-16125	REV.1903	781229
CLRIO	43317	43325	750701	24998-16001	
OURD	43326	43326	92067-16125	REV.1903	780526
SEHVE	43327	43415	92067-16268	REV.1903	790202
LURG	43416	44000	92067-16268	REV.1903	790223
.DADS	44001	44110	780818	24998-16001	
.DHP	44111	44256	780818	24998-16001	
.DDI	44257	44557	781021	24998-16001	
SESEN	44558	44575	92067-16125	REV.1903	780413
R/MS	44576	44734	92067-16125	REV.1903	781214
P.PAS	44735	44763	92067-16125	REV.1903	740801
.DNG	44764	44773	780818	24998-16001	
PAUSE	44774	45074	771122	24998-16001	
SALRN	45075	45212	92067-16268	REV.1903	770715
FHTIO	45213	46511	24998-16002	REV.1926	790417
ERR0	46512	46601	771122	24998-16001	
ABS	46602	46610	750701	24998-16001	
.DDE	46611	46622	780818	24998-16001	
.DIN	46623	46638	780818	24998-16001	
.RTOI	46639	46724	780921	24998-16001	
.FPUH	46725	46766	781106	24998-16001	
.SRT	46767	47027	770518	24998-16001	
.FCM	47028	47044	750701	24998-16001	
PAU.E	47045	47045	750701	24998-16001	
ER0.E	47046	47046	750701	24998-16001	
OPEN	47047	47223	92067-16125	REV.1903	790103
RWSUB	47224	47571	92067-16125	REV.1903	781003
RUND0	47572	47714	92067-16125	REV.1903	780801
FRMTR	47715	51322	24998-16002	REV.1926	790503
FHT.E	51323	51323	24998-16002	REV.1901	781107
REIO	51324	51500	92067-16268	REV.1903	790316
RMPAR	51501	53543	781106	24998-16001	
PNAME	53544	53611	771121	24998-16001	
LUTRU	53612	53720	92067-16268	REV.1903	790223
SETP	53721	53745	781106	24998-16001	
.CFER	53746	54023	750701	24998-16001	
.LBT	54024	54054	770518	24998-16001	

20 PAGES RELOCATED      20 PAGES REQ'D      NO PAGES EMA      NO PAGES MSEC  
 LINKS:BP      PROGRAM:LB      LOAD:TE      COMMON:NC  
 /LOADR:FITR      READY AT 3:53 PM FRI., 26 SEPT, 1980  
 /LOADR:END



# 4.6.3. Results

printed output

X1

## 1. DATA SET

	1	2	3	4	5
1	.100	.100	.100	.100	.100
2	.100	.100	.100	.100	.100
3	.100	.100	.100	.100	.100
4	.100	.100	.100	.100	.100
5	.100	.100	.100	.100	.100

## 2. DATA SET

	1	2	3	4	5
1	.200	.200	.200	.200	.200
2	.200	.200	.200	.200	.200
3	.200	.200	.200	.200	.200
4	.200	.200	.200	.200	.200
5	.200	.200	.200	.200	.200

## 3. DATA SET

	1	2	3	4	5
1	.300	.300	.300	.300	.300
2	.300	.300	.300	.300	.300
3	.300	.300	.300	.300	.300
4	.300	.300	.300	.300	.300
5	.300	.300	.300	.300	.300

## 4. DATA SET

	1	2	3	4	5
1	.400	.400	.400	.400	.400
2	.400	.400	.400	.400	.400
3	.400	.400	.400	.400	.400
4	.400	.400	.400	.400	.400
5	.400	.400	.400	.400	.400

## 5. DATA SET

	1	2	3	4	5
1	.500	.500	.500	.500	.500
2	.500	.500	.500	.500	.500
3	.500	.500	.500	.500	.500
4	.500	.500	.500	.500	.500
5	.500	.500	.500	.500	.500

X2

## 1. DATA SET

	1	2	3	4	5
1	-20.000	-20.000	-20.000	-20.000	-20.000
2	-10.000	-10.000	-10.000	-10.000	-10.000
3	0.000	0.000	0.000	0.000	0.000
4	10.000	10.000	10.000	10.000	10.000
5	20.000	20.000	20.000	20.000	20.000

## 2. DATA SET

	1	2	3	4	5
1	-20.000	-20.000	-20.000	-20.000	-20.000
2	-10.000	-10.000	-10.000	-10.000	-10.000
3	0.000	0.000	0.000	0.000	0.000
4	10.000	10.000	10.000	10.000	10.000
5	20.000	20.000	20.000	20.000	20.000

## 3. DATA SET

	1	2	3	4	5
1	-20.000	-20.000	-20.000	-20.000	-20.000
2	-10.000	-10.000	-10.000	-10.000	-10.000
3	0.000	0.000	0.000	0.000	0.000
4	10.000	10.000	10.000	10.000	10.000
5	20.000	20.000	20.000	20.000	20.000

## 4. DATA SET

	1	2	3	4	5
--	---	---	---	---	---

1	-20.000	-20.000	-20.000	-20.000	-20.000
2	-10.000	-10.000	-10.000	-10.000	-10.000
3	0.000	0.000	0.000	0.000	0.000
4	10.000	10.000	10.000	10.000	10.000
5	20.000	20.000	20.000	20.000	20.000

# 5. DATA SET

1	-20.000 <sup>1</sup>	-20.000 <sup>2</sup>	-20.000 <sup>3</sup>	-20.000 <sup>4</sup>	-20.000 <sup>5</sup>
2	-10.000	-10.000	-10.000	-10.000	-10.000
3	0.000	0.000	0.000	0.000	0.000
4	10.000	10.000	10.000	10.000	10.000
5	20.000	20.000	20.000	20.000	20.000

X3

# 1. DATA SET

1	-20.000 <sup>1</sup>	-10.000 <sup>2</sup>	0.000 <sup>3</sup>	10.000 <sup>4</sup>	20.000 <sup>5</sup>
2	-10.000	-10.000	0.000	10.000	20.000
3	0.000	0.000	0.000	10.000	20.000
4	10.000	10.000	0.000	10.000	20.000
5	20.000	10.000	0.000	10.000	20.000

# 2. DATA SET

1	-20.000 <sup>1</sup>	-10.000 <sup>2</sup>	0.000 <sup>3</sup>	10.000 <sup>4</sup>	20.000 <sup>5</sup>
2	-10.000	-10.000	0.000	10.000	20.000
3	0.000	0.000	0.000	10.000	20.000
4	10.000	10.000	0.000	10.000	20.000
5	20.000	10.000	0.000	10.000	20.000

# 3. DATA SET

1	-20.000 <sup>1</sup>	-10.000 <sup>2</sup>	0.000 <sup>3</sup>	10.000 <sup>4</sup>	20.000 <sup>5</sup>
2	-10.000	-10.000	0.000	10.000	20.000
3	0.000	0.000	0.000	10.000	20.000
4	10.000	10.000	0.000	10.000	20.000
5	20.000	10.000	0.000	10.000	20.000

# 4. DATA SET

1	-20.000 <sup>1</sup>	-10.000 <sup>2</sup>	0.000 <sup>3</sup>	10.000 <sup>4</sup>	20.000 <sup>5</sup>
2	-10.000	-10.000	0.000	10.000	20.000
3	0.000	0.000	0.000	10.000	20.000
4	10.000	10.000	0.000	10.000	20.000
5	20.000	10.000	0.000	10.000	20.000

# 5. DATA SET

1	-20.000 <sup>1</sup>	-10.000 <sup>2</sup>	0.000 <sup>3</sup>	10.000 <sup>4</sup>	20.000 <sup>5</sup>
2	-10.000	-10.000	0.000	10.000	20.000
3	0.000	0.000	0.000	10.000	20.000
4	10.000	10.000	0.000	10.000	20.000
5	20.000	10.000	0.000	10.000	20.000

Y

# 1. DATA SET

1	.400	.700	.800	.700	.400
2	.450	.750	.850	.750	.450
3	.500	.800	.900	.800	.500
4	.450	.750	.850	.750	.450
5	.400	.700	.800	.700	.400

# 2. DATA SET

1	.440	.760	.860	.760	.440
2	.490	.810	.910	.810	.490
3	.540	.860	.960	.860	.540
4	.490	.810	.910	.810	.490
5	.440	.760	.860	.760	.440



## 5. CONCLUSIONS AND RECOMMENDATIONS

Compared to the probe calibration approach described in Ref 3. the method presented here provides a much simpler way to both calculate the probe calibration surfaces from measured data and to apply the calibration to on-line data reduction. The iteration required in the method of Ref 3 is completely omitted. It is noted however, that the Gauss Jordan Elimination lacks some sophistication. The system matrix for example undergoes the elimination without prior conditioning. Particularly in the case of the 3-D and the 4-D approximations, numerical round off errors have an influence on the accuracy of the coefficients. Consequently it is recommended, that the Gauss Jordan Elimination routine be revised to use double precision constants and a routine that conditions the system matrix be added. Time constraints made these steps impossible for the author.

Following these refinements, a calibration procedure, similar to McGuire's (ref 3) should be formalised. Since the 4-D- approximation requires large arrays and therefore extensive CP - memory, extended memory access (EMA) is necessary if the HP 21MX computer is used to perform the calculations.

#### LIST OF REFERENCES

1. McCracken, D. D. and Dorn, W. S.: Numerical Methods and FORTRAN Programming, Third Printing. New York: Wiley & Sons Inc., 1965. Page 262ff.
2. Simon, H.: Anwendung verschiedener Berechnungsverfahren zur Auslegung eines Überschallveridchler-Laufrades, und dessen experimentielle Untersuchung, Dissertation, RWTH Aachen, 1973, Page 134f.  
  
[Application of Various Calculation Methods for the Design of a Supersonic Rotor and Its Experimental Investigation, Dissertation, Technical University Aachen, 1973.]
3. McGuire, A. G.: Pneumatic Velocity Probe Calibration-Users Manual for Data Acquisition and Reduction. Monterey: Naval Postgraduate School, Turbopropulsion Laboratory, Technical Note 80-01, 1980.

# APPENDIX A: Some Useful Matrix Conventions and Operations

## A1. Submatrix Notation

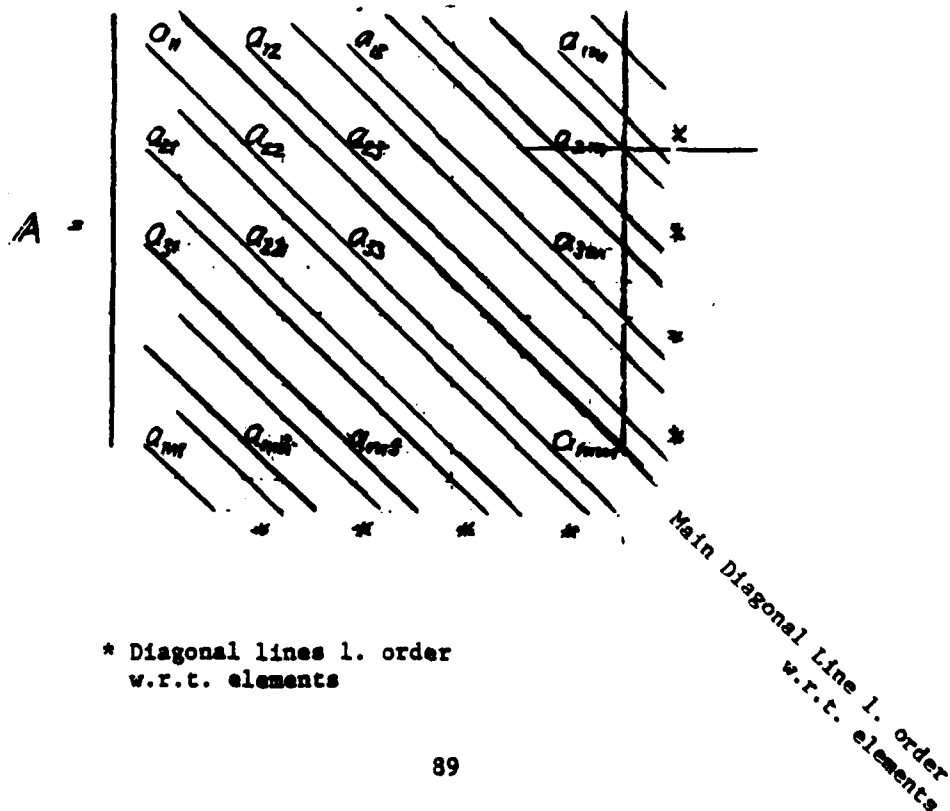
$$A = \begin{bmatrix} a_{11} & a_{12} & a_{13} & a_{14} & a_{1m} \\ a_{21} & a_{22} & a_{23} & a_{24} & a_{2m} \\ a_{31} & a_{32} & a_{33} & a_{34} & a_{3m} \\ a_{41} & a_{42} & a_{43} & a_{44} & a_{4m} \\ a_{m1} & a_{m2} & a_{m3} & a_{m4} & a_{mm} \end{bmatrix} \quad \text{or} \quad \begin{bmatrix} A_{11}^* & A_{12}^* & \dots & A_{1m}^* \\ A_{21}^* & A_{22}^* & \dots & A_{2m}^* \\ A_{31}^* & A_{32}^* & \dots & A_{3m}^* \\ A_{41}^* & A_{42}^* & \dots & A_{4m}^* \\ A_{m1}^* & A_{m2}^* & \dots & A_{mm}^* \end{bmatrix}$$

defined through elements  $a_{ij}$  or submatrices  $A_{ij}^*$

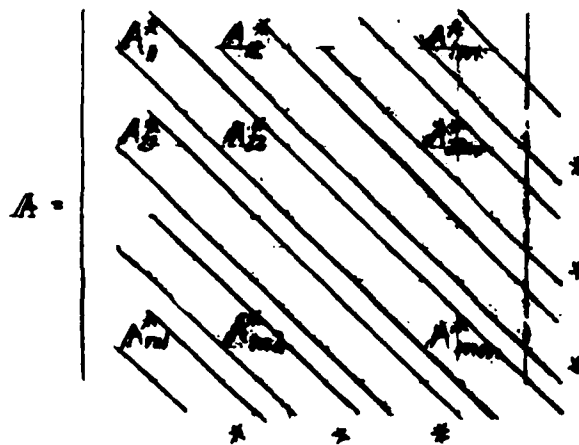
## A2. Diagonal Lines and Diagonal Bands 1. order

Diagonal lines 1. order and Main diagonal line 1. order

Diagonal bands 1. order and Main diagonal band 1. order



\* Diagonal lines 1. order  
w.r.t. elements



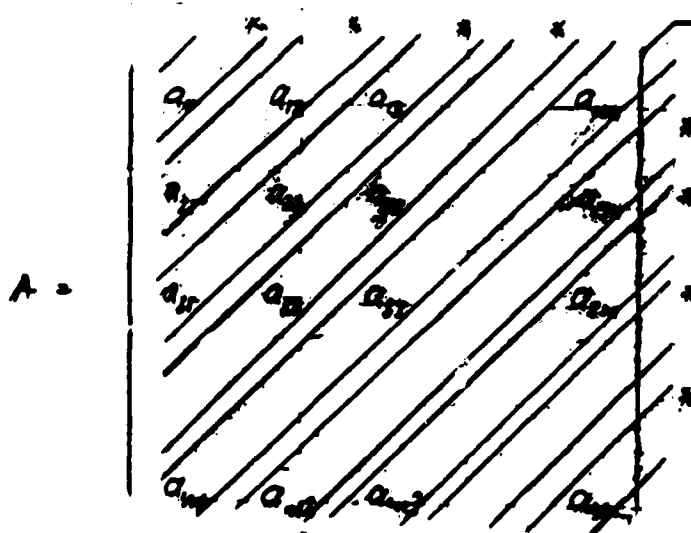
\* Diagonal bands 1. order  
w.r.t. submatrices

Main Diagonal Band 1. order  
w.r.t. Submatrices

### A3. Diagonal Lines and Diagonal Bands 2. order

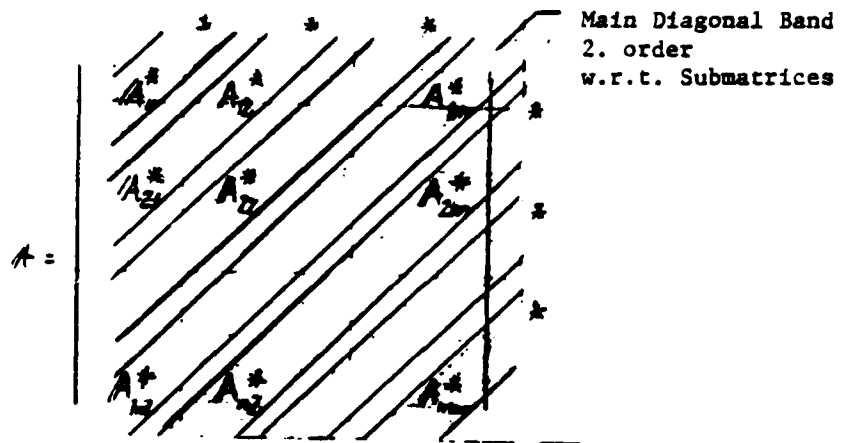
Diagonal lines 2. order and Main diagonal line 2. order

Diagonal bands 2. order and Main diagonal band 2. order



Main Diagonal Line  
1. order w.r.t. elements

\* Diagonal lines 2. order w.r.t. elements



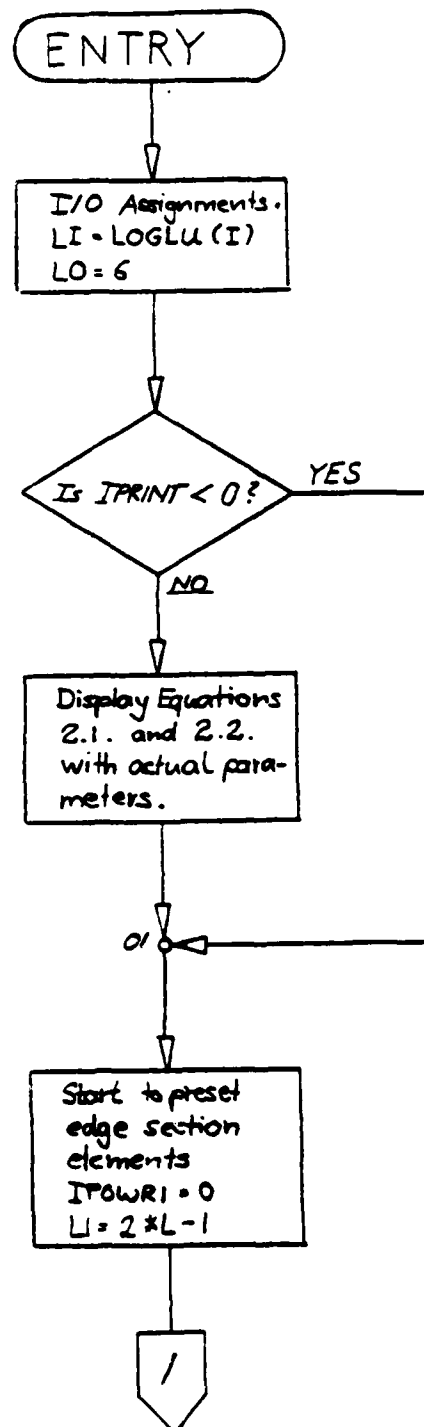


APPENDIX B: Software Description: Flow Charts

The following are given:

<u>Flow Chart</u>	<u>Page</u>
MAT2	93
S2	97
MAT3	98
MAT31	106
S3	109
MAT4	111
MAT41	119
MAT42	124
S4	127

Flow chart MAT2



F/8 20/4

AUG 80 H ZEBNER

N00014-78-C-0204

NL

**NPS-67-80-001CR**

2.2

24.

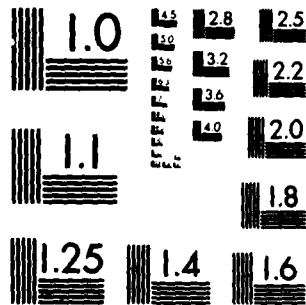
END

9016

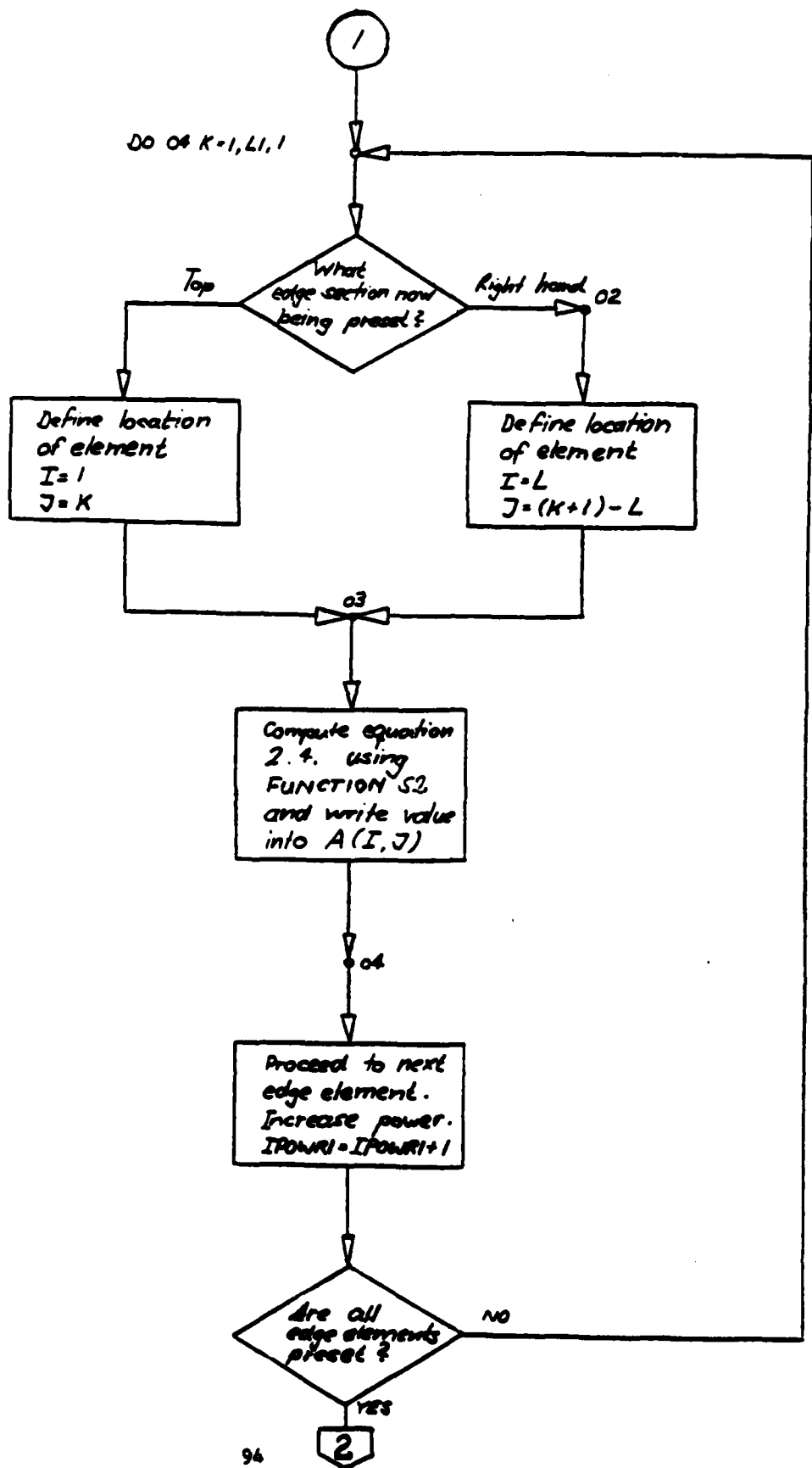
FILMED

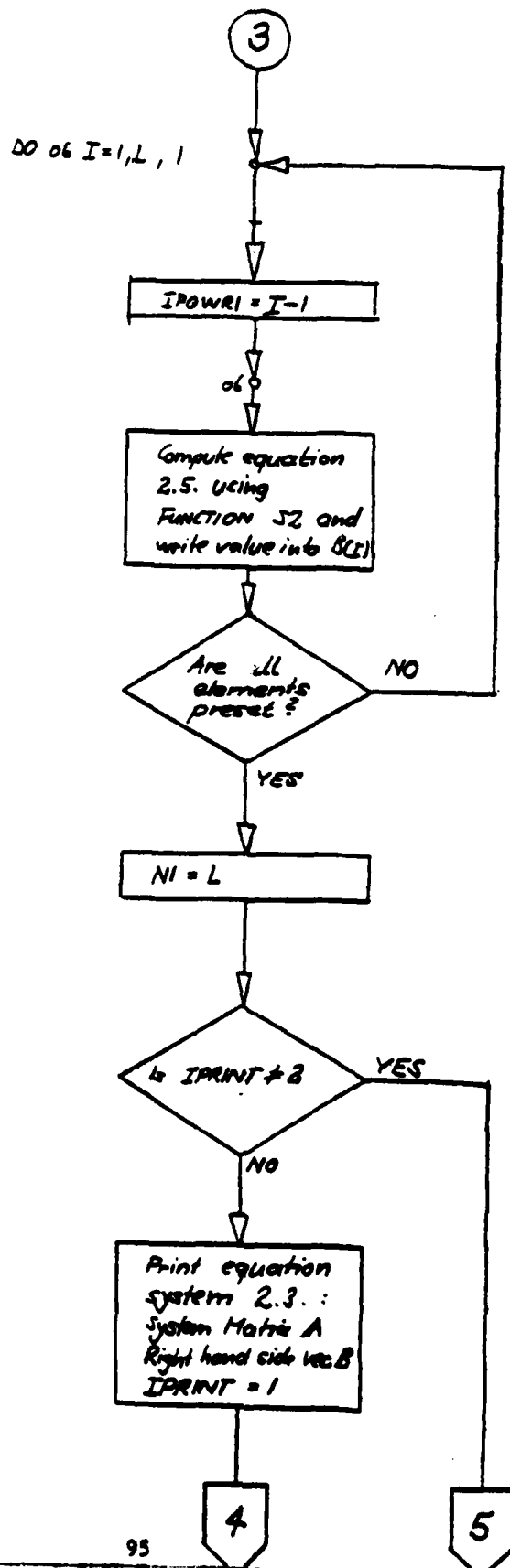
323

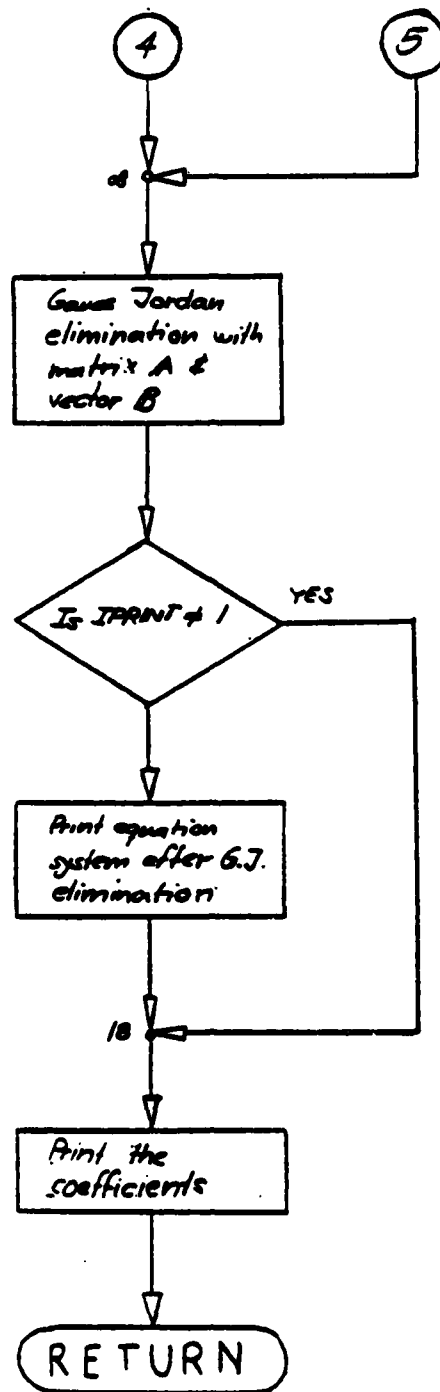
DTIC



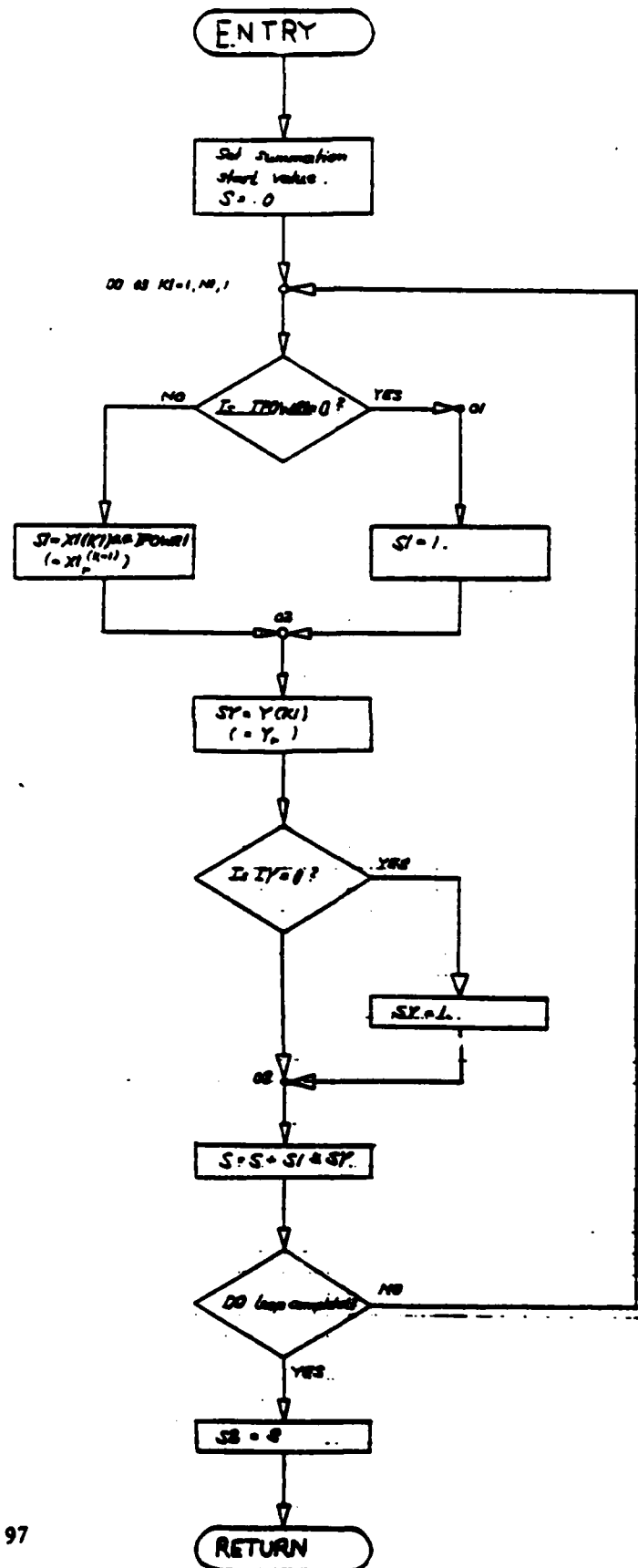
MICROCOPY RESOLUTION TEST CHART  
NATIONAL BUREAU OF STANDARDS-1963-A.





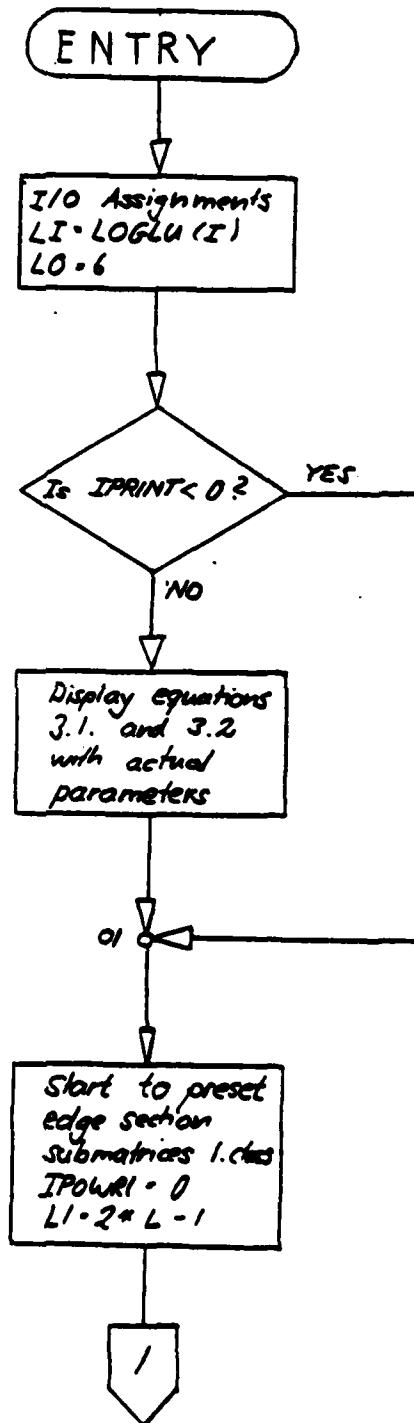


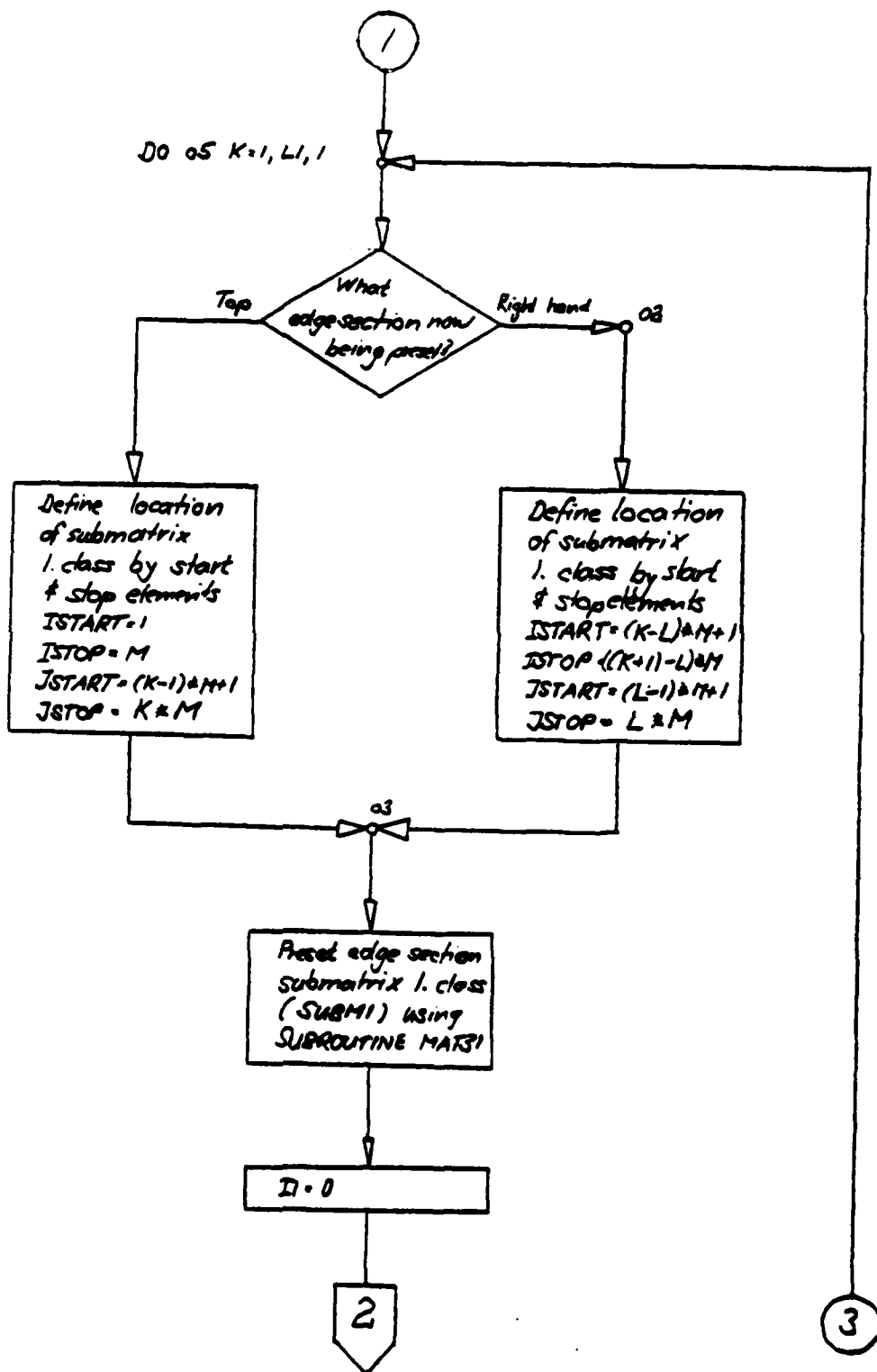
Flow chart S2

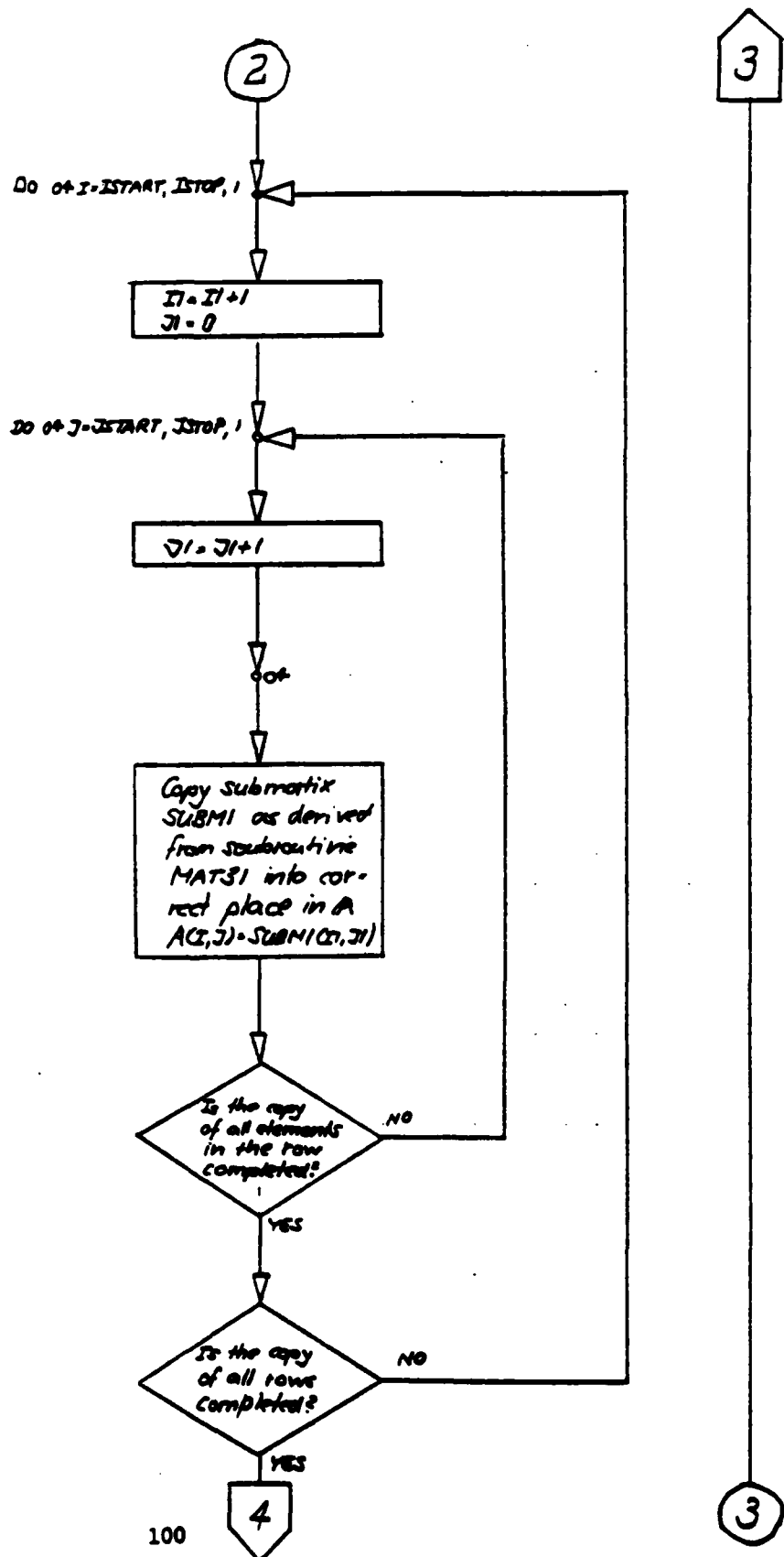


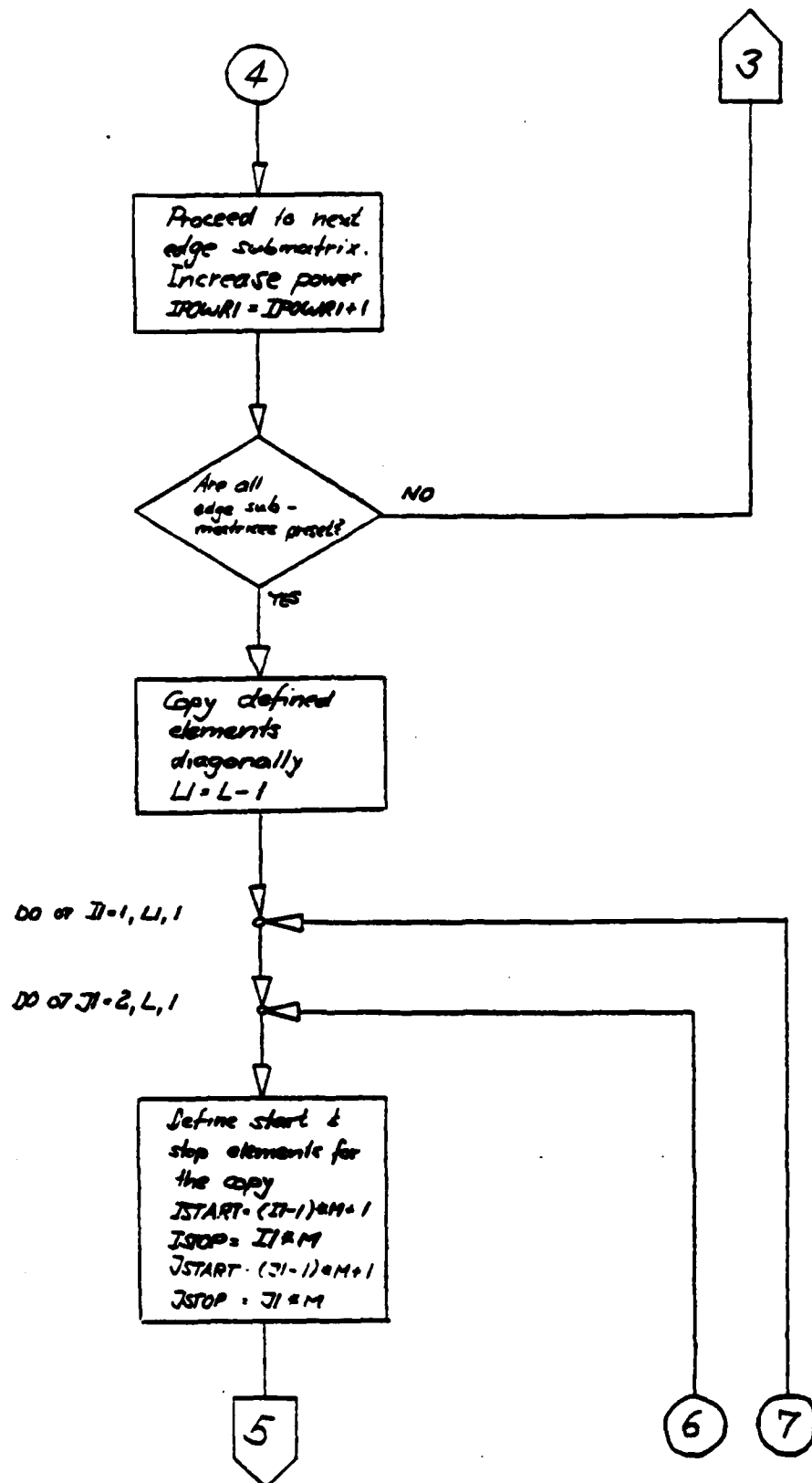


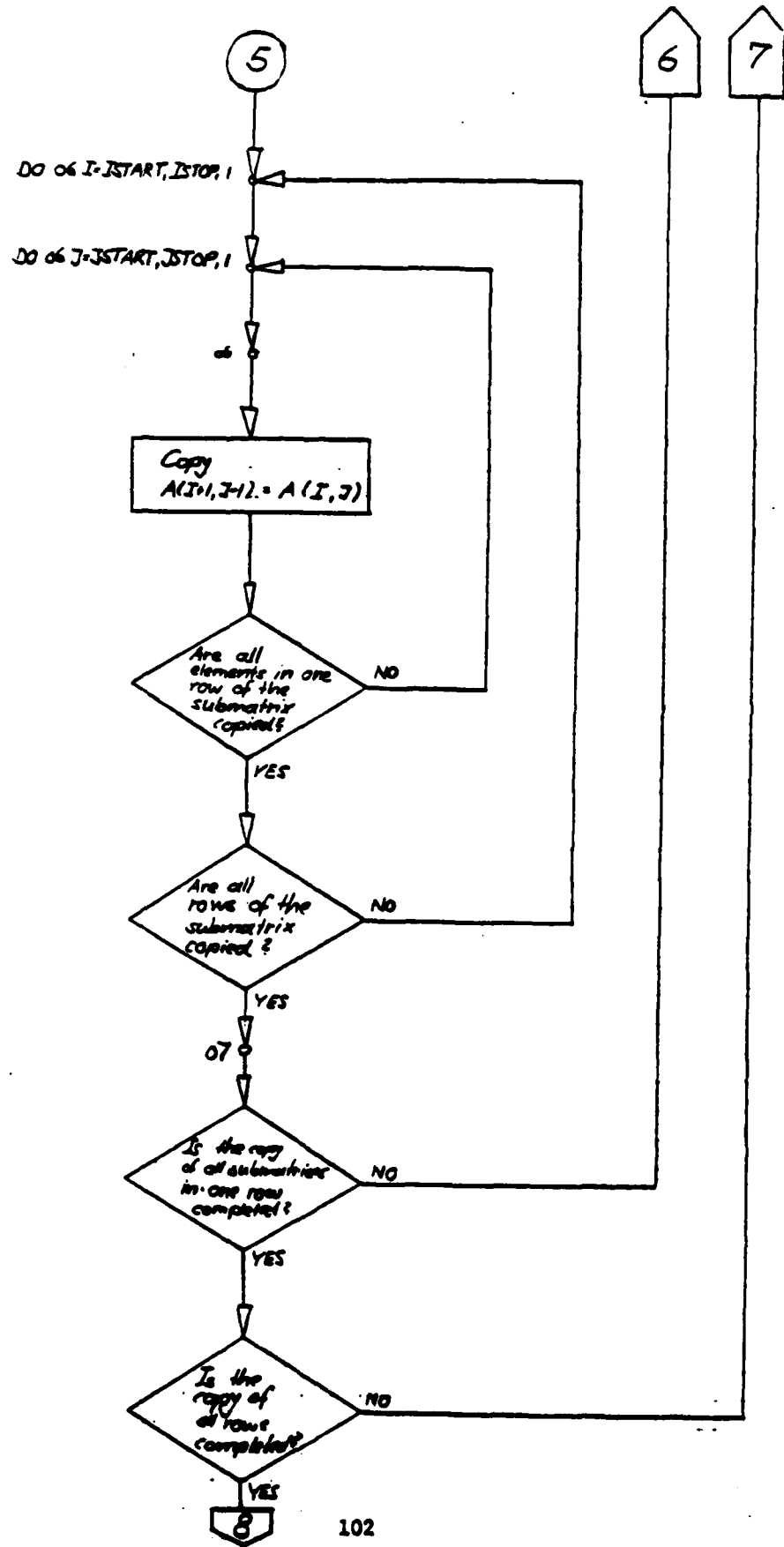
Flow chart MAT3

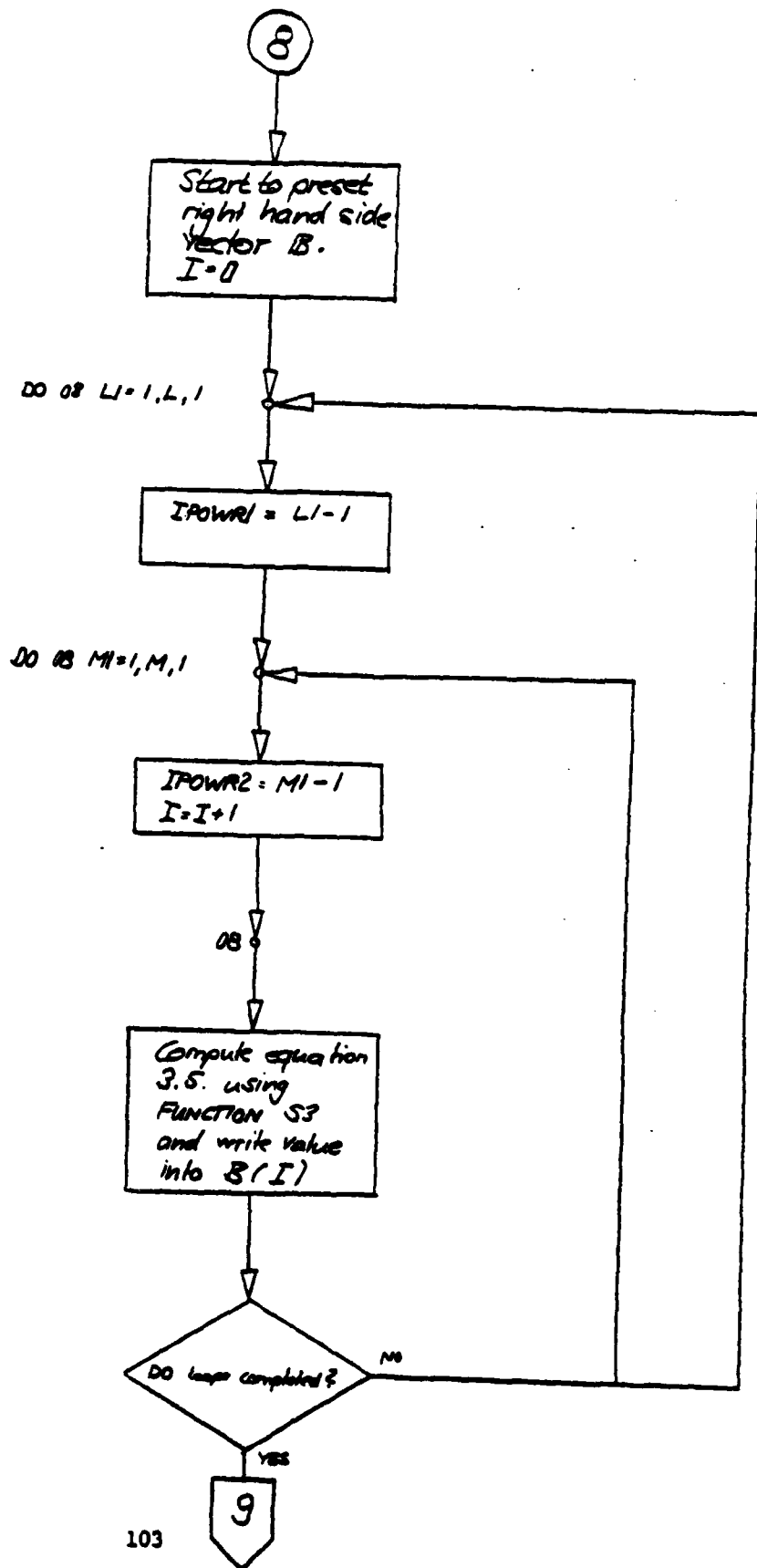


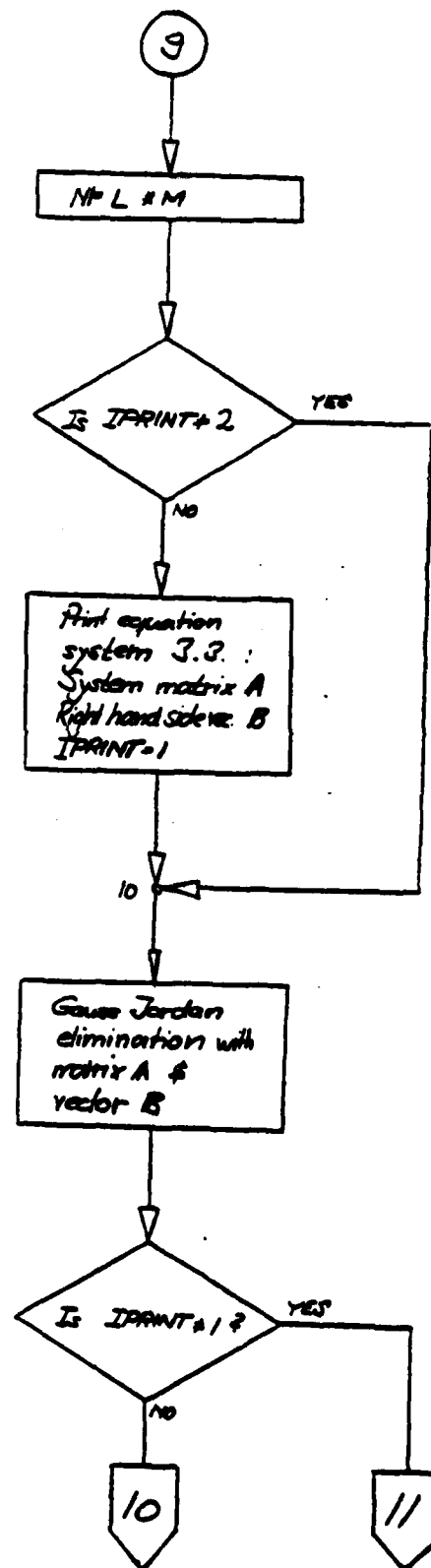


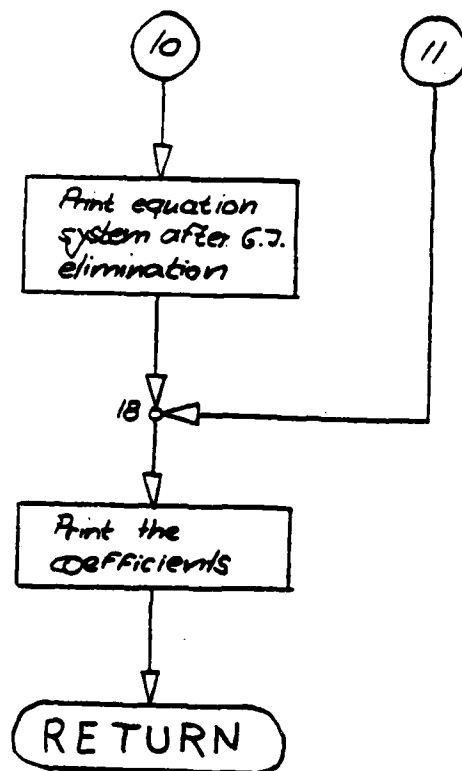






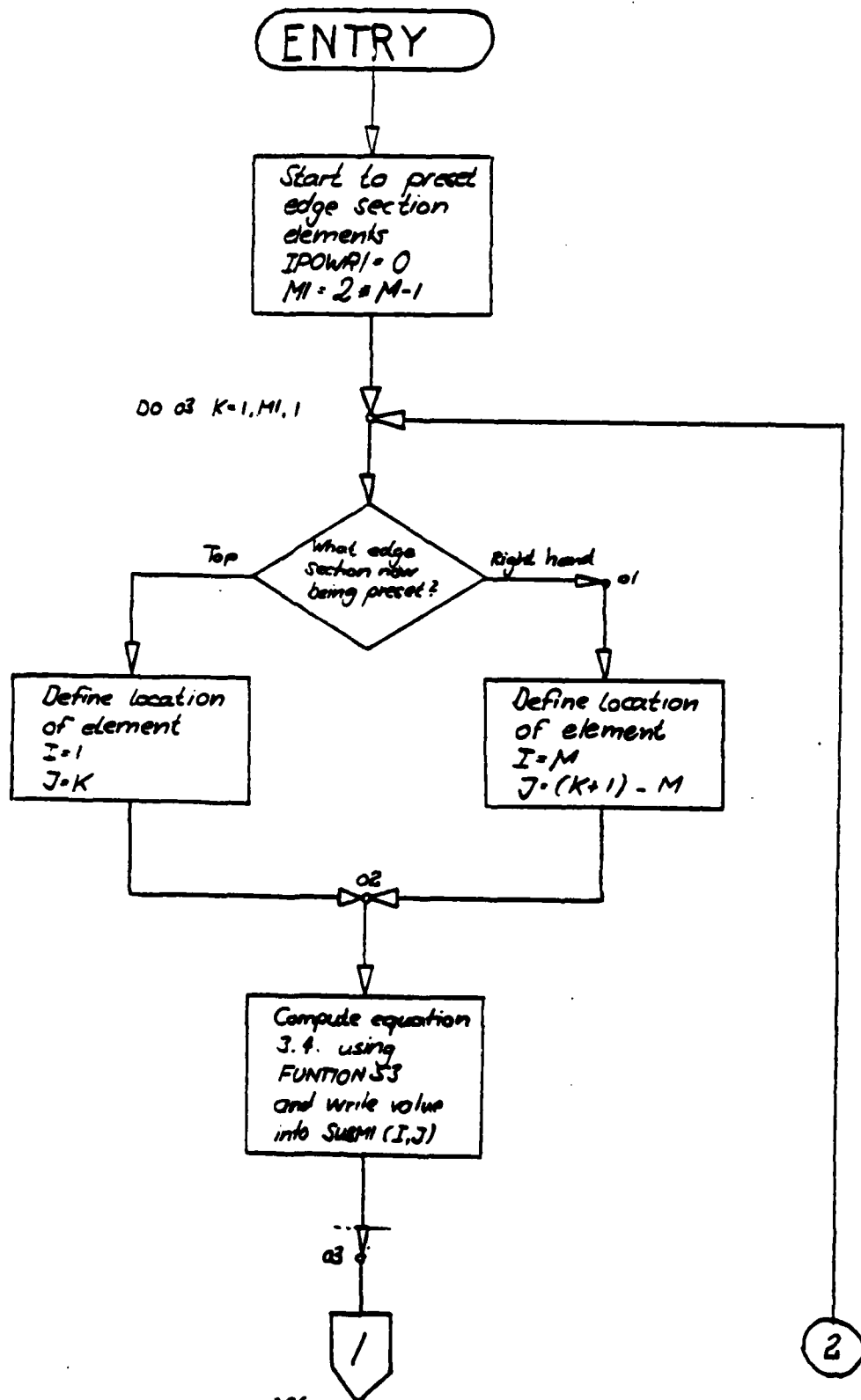


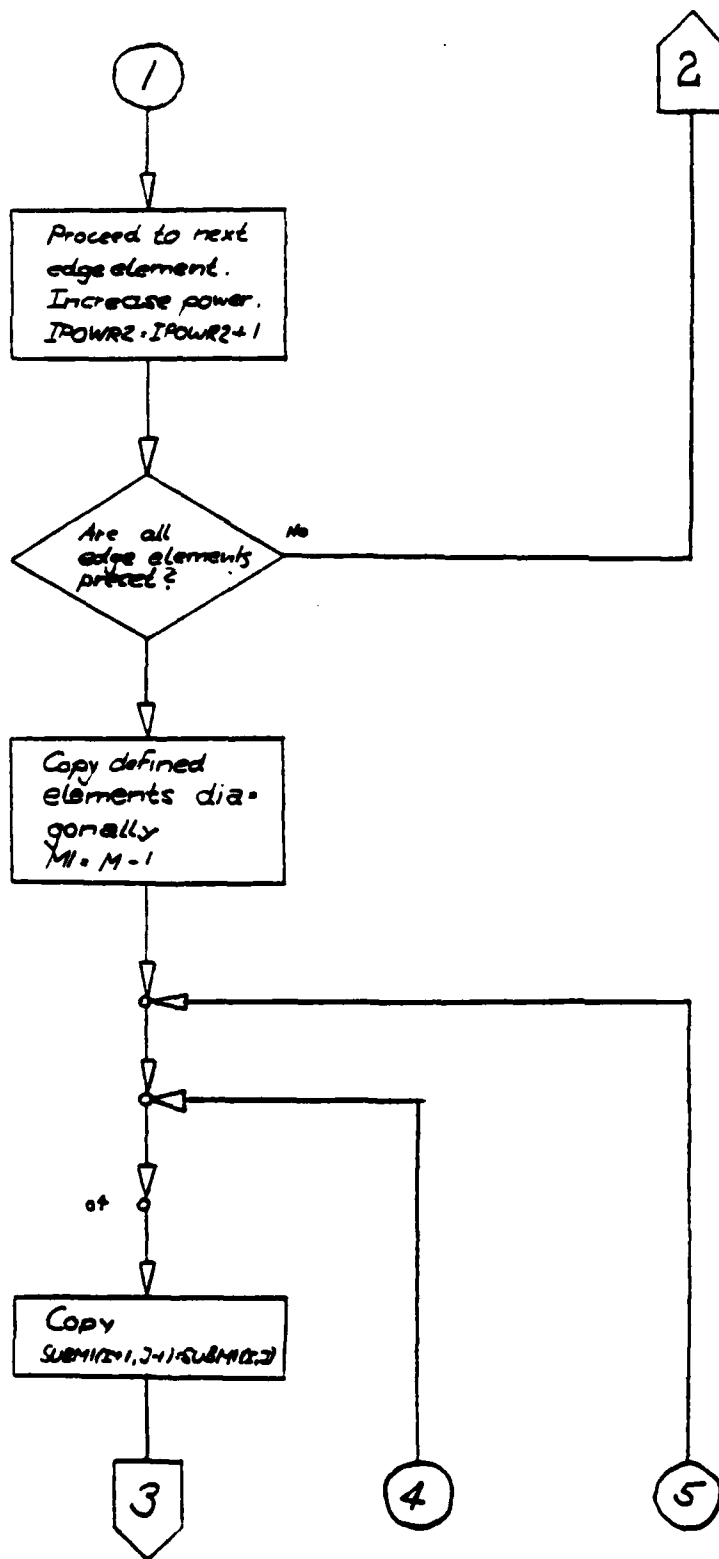


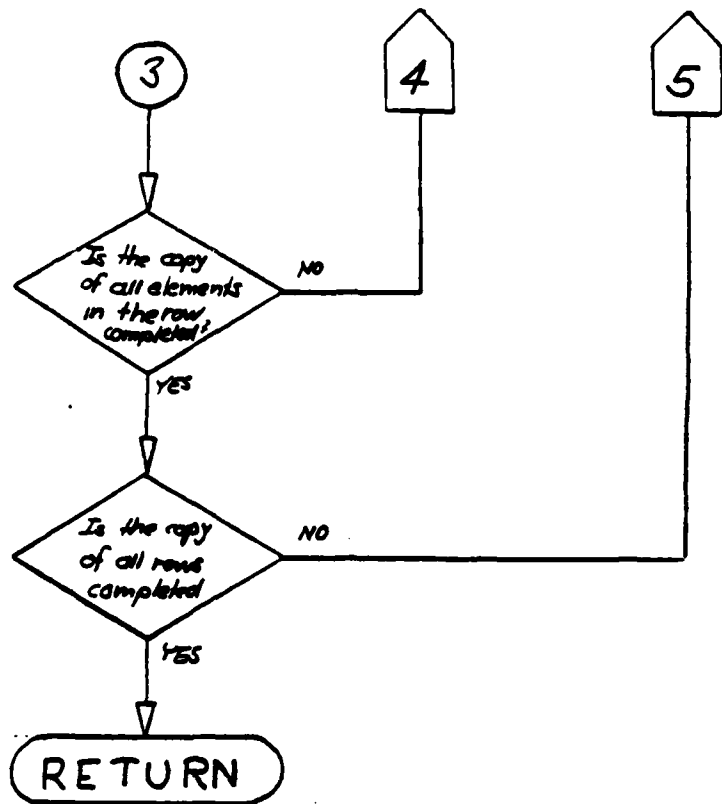




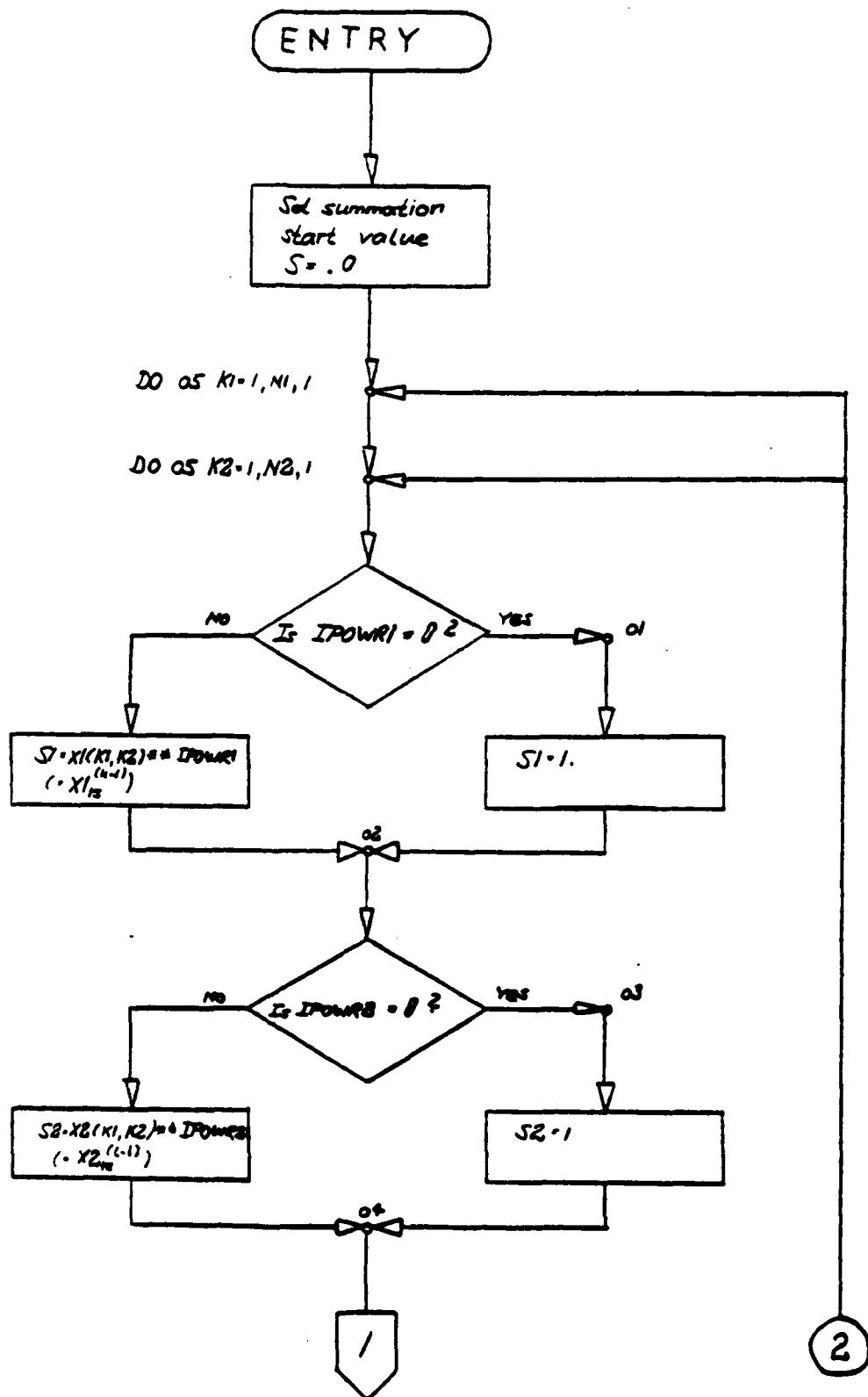
Flow chart MAT31

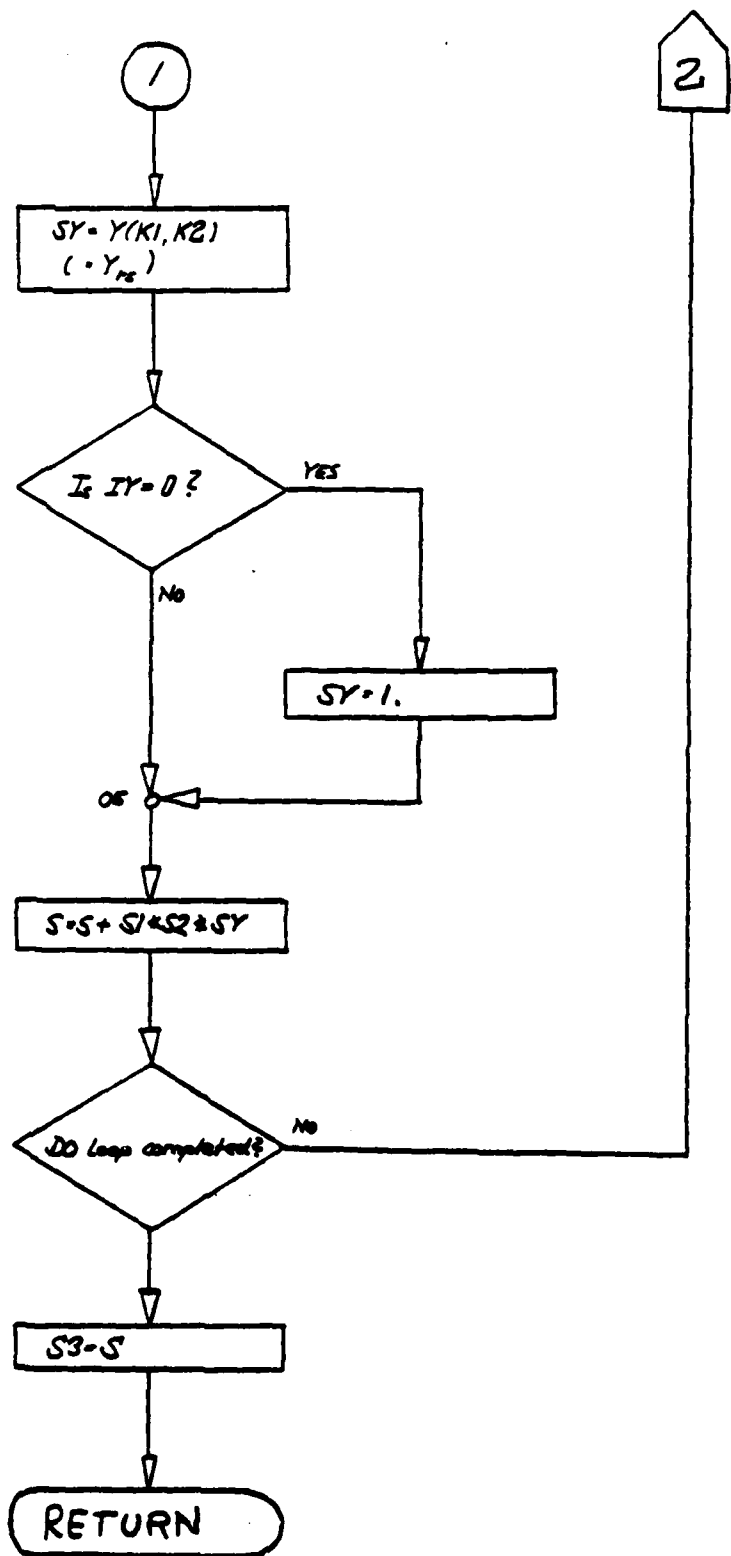




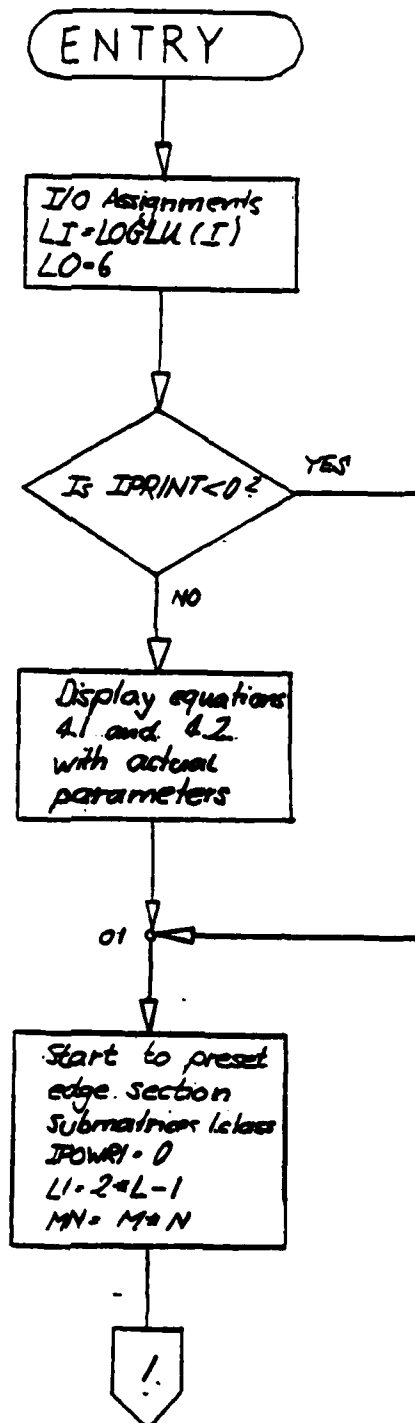


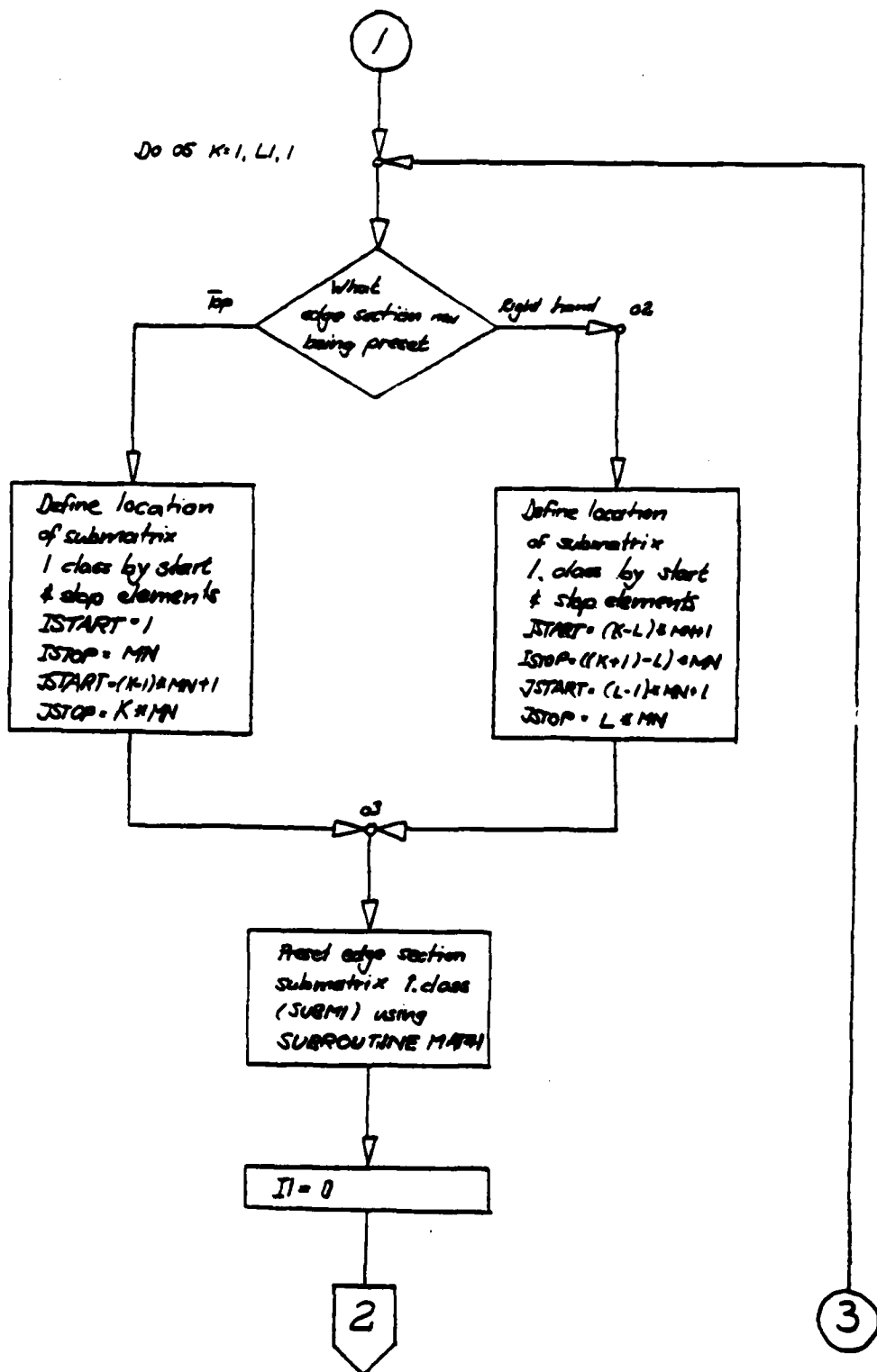
Flow chart S3

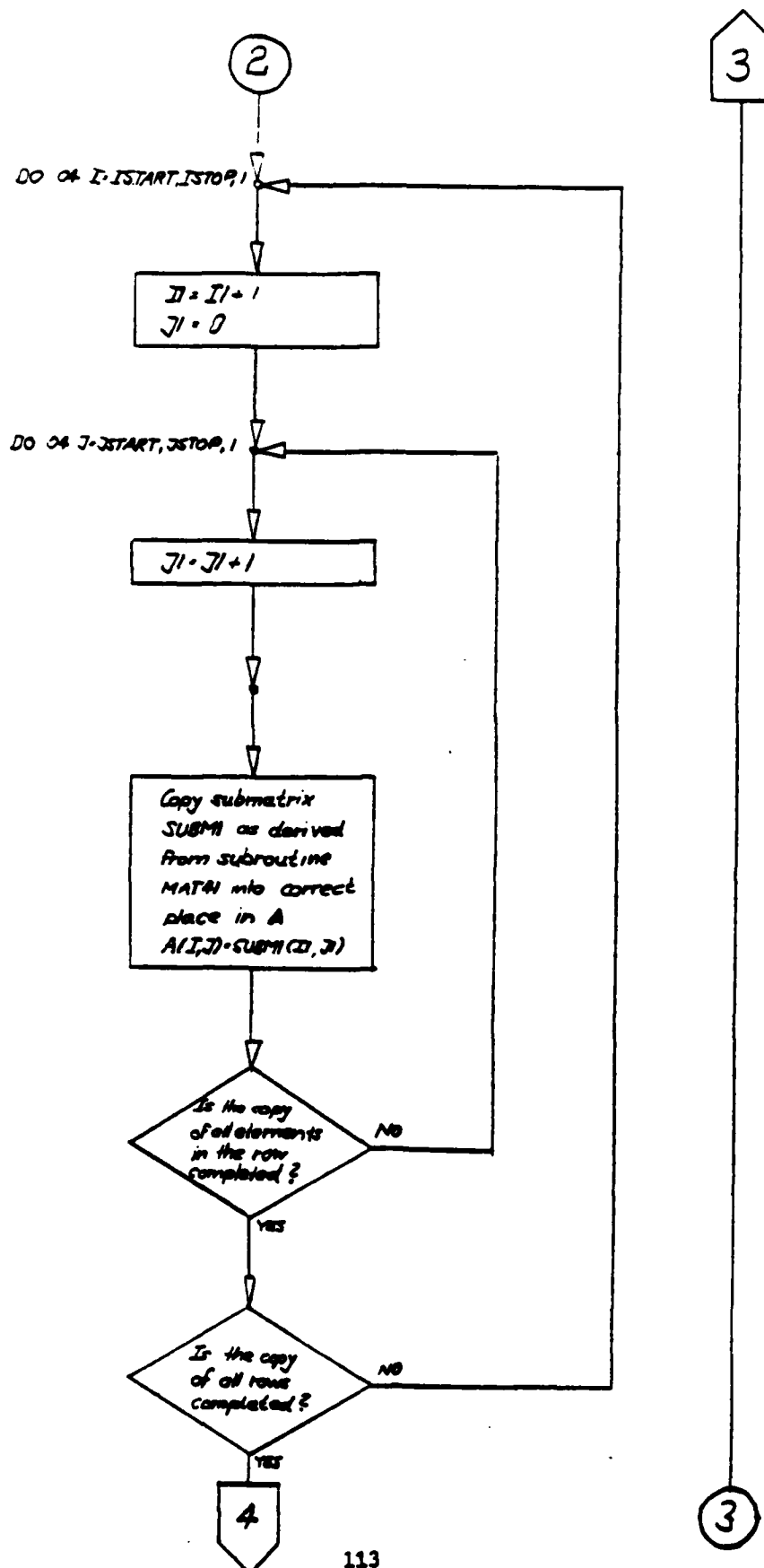




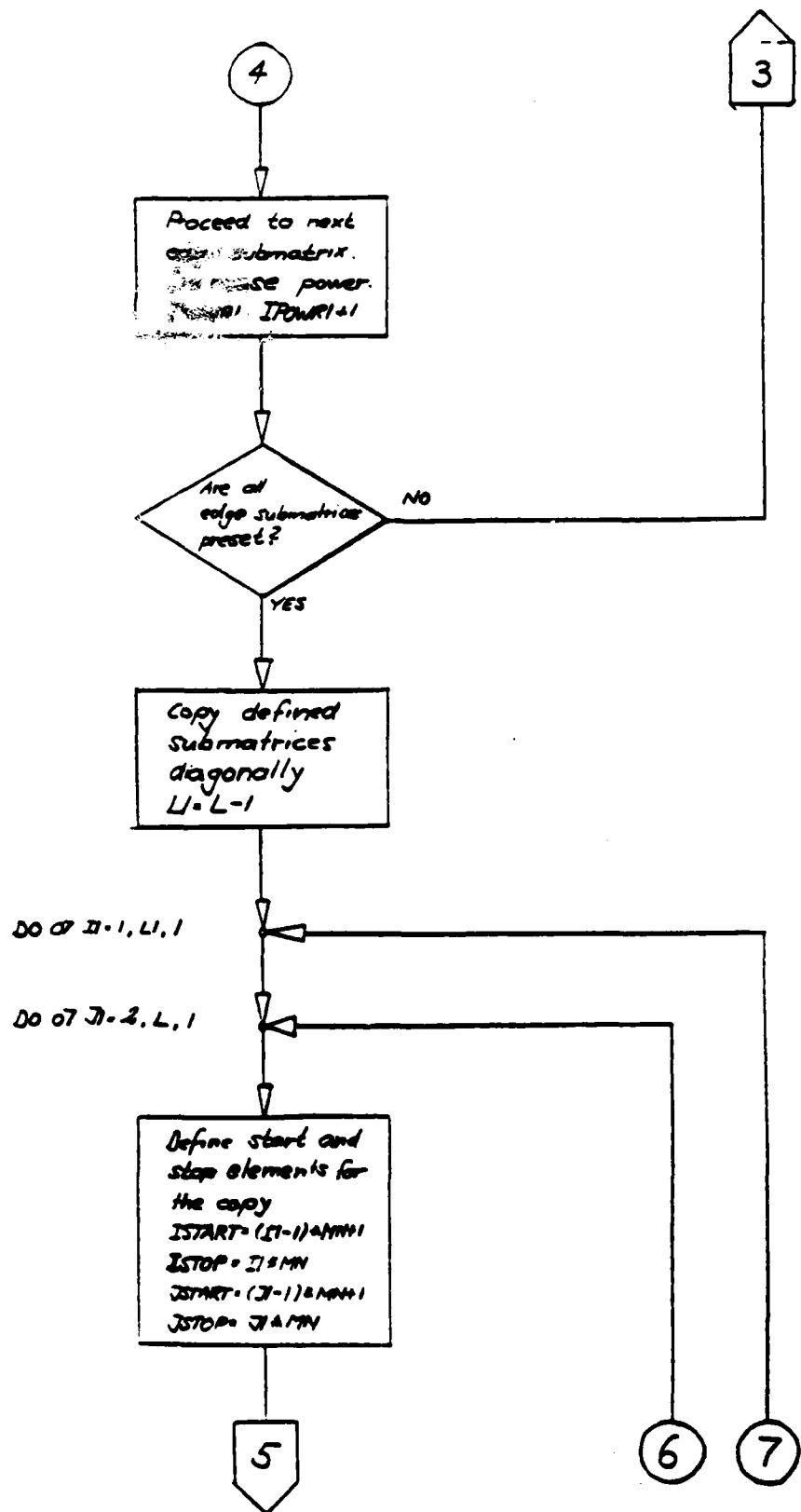
Flow chart MAT4

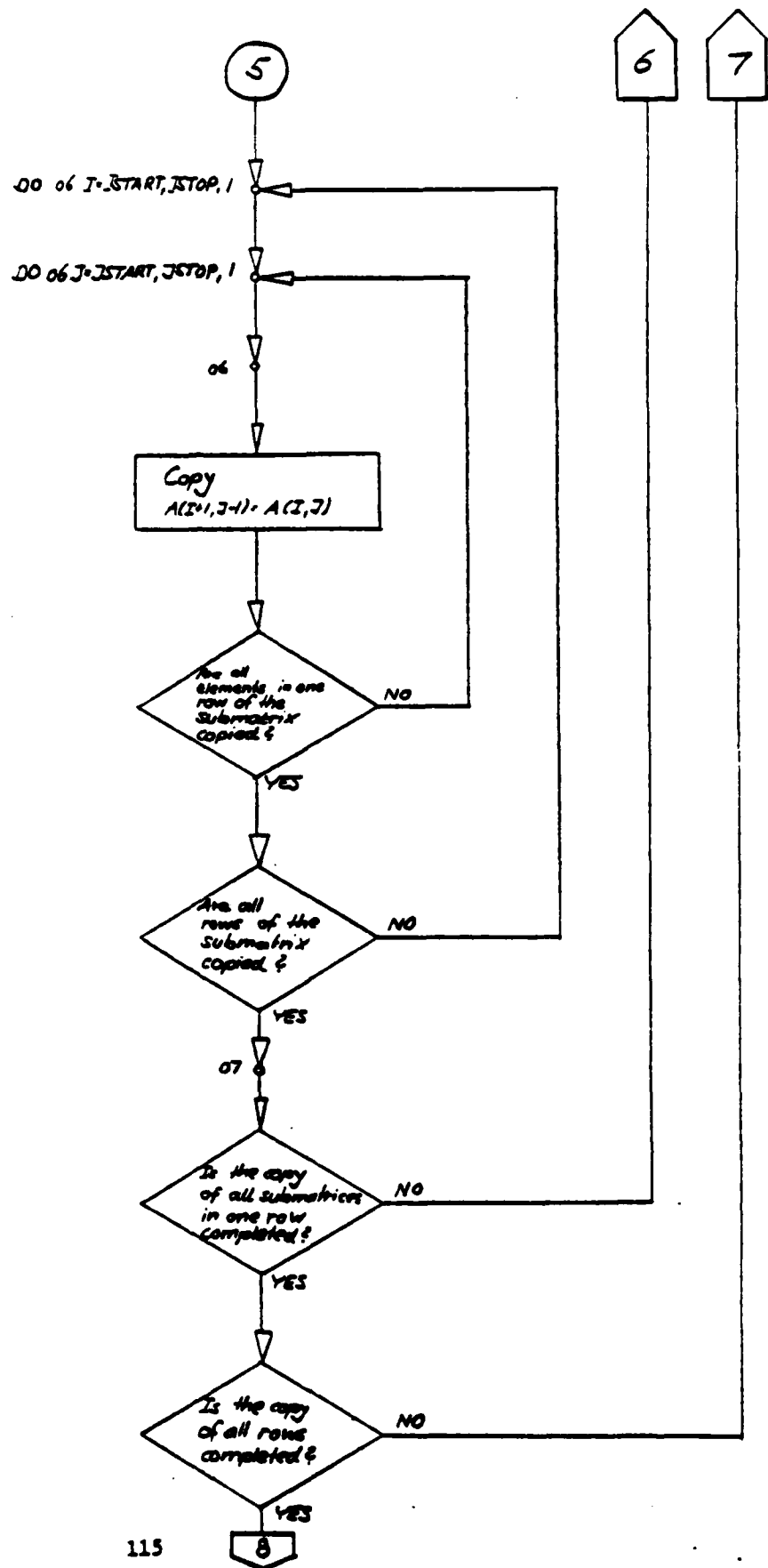


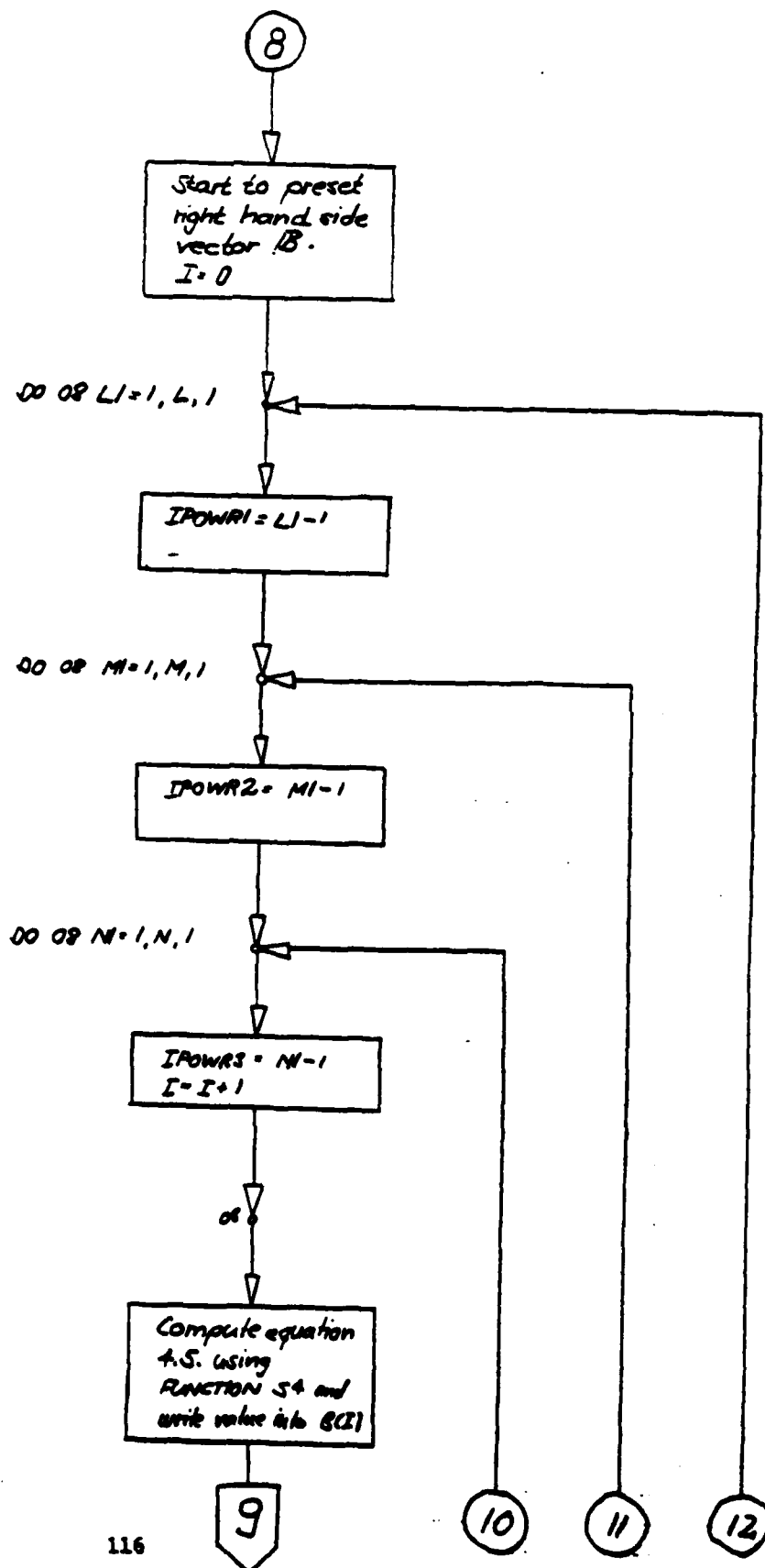


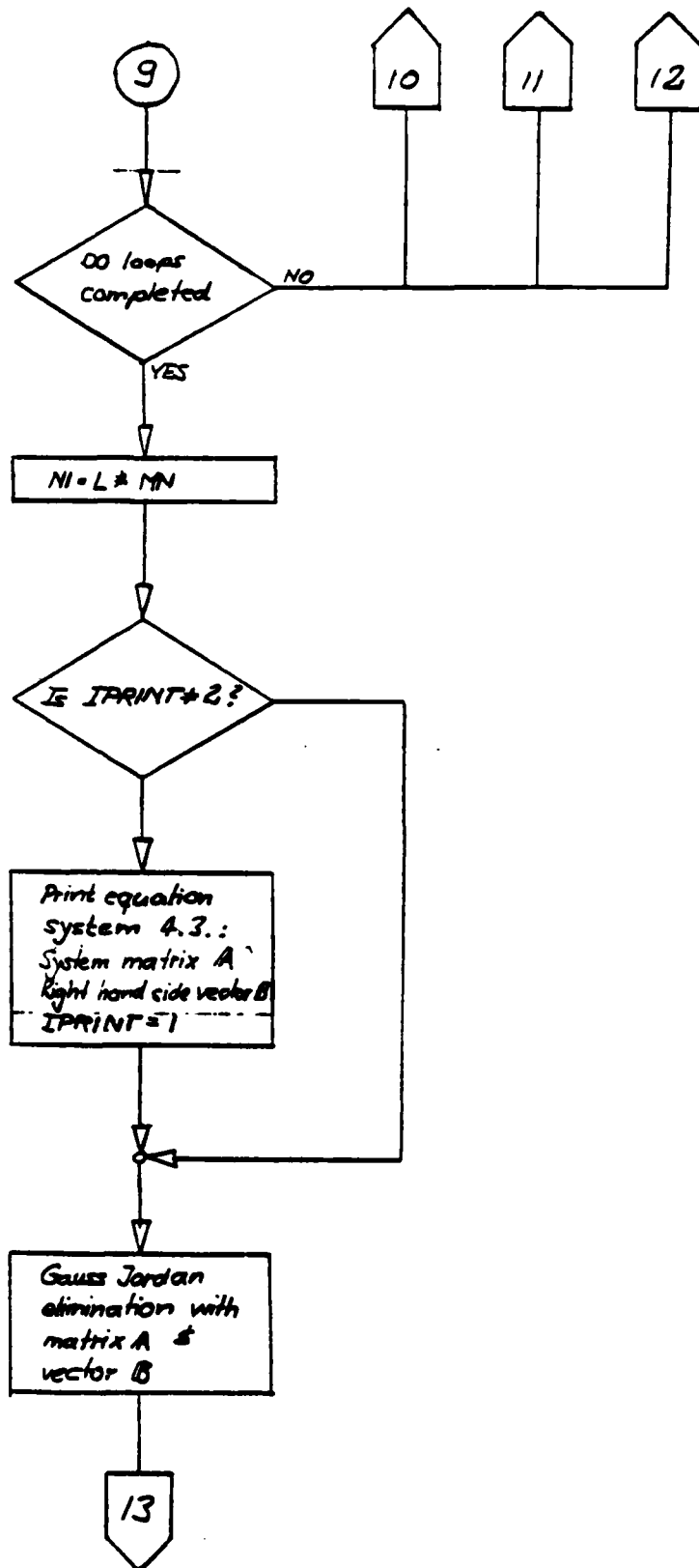


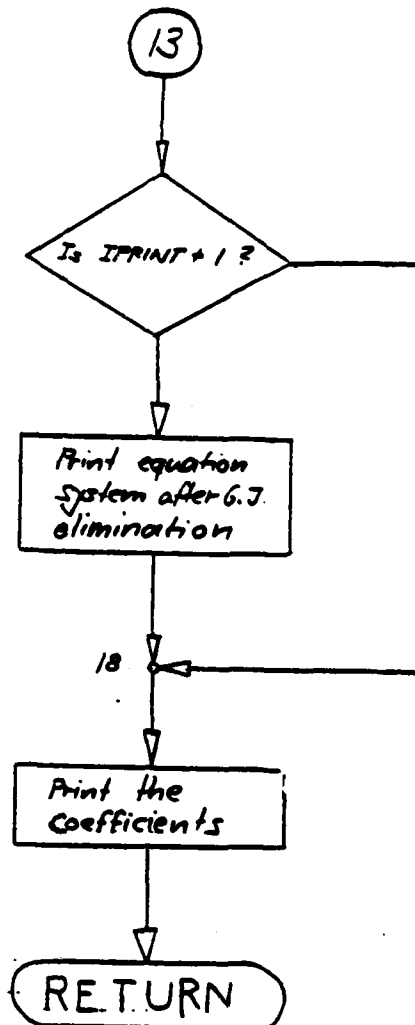




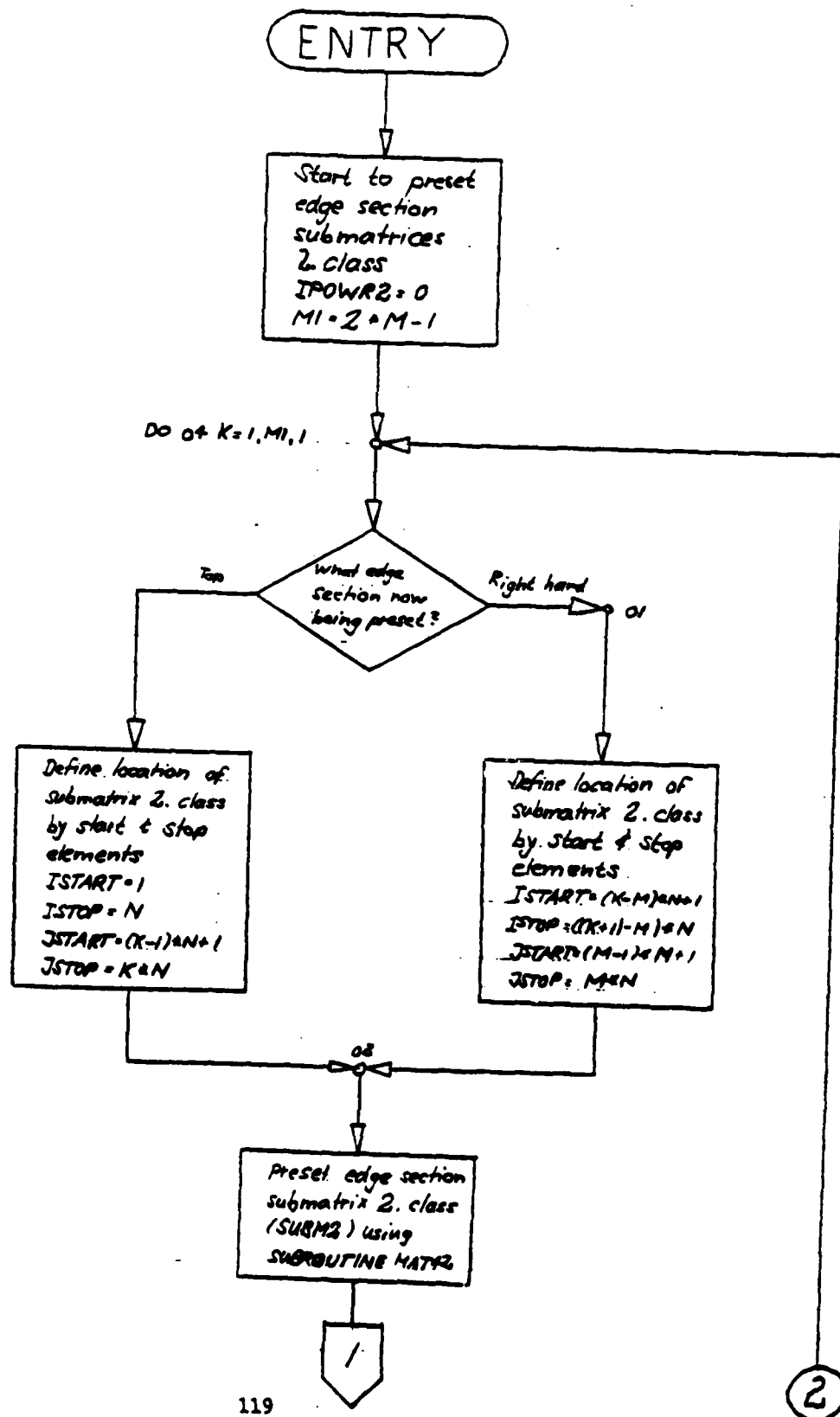


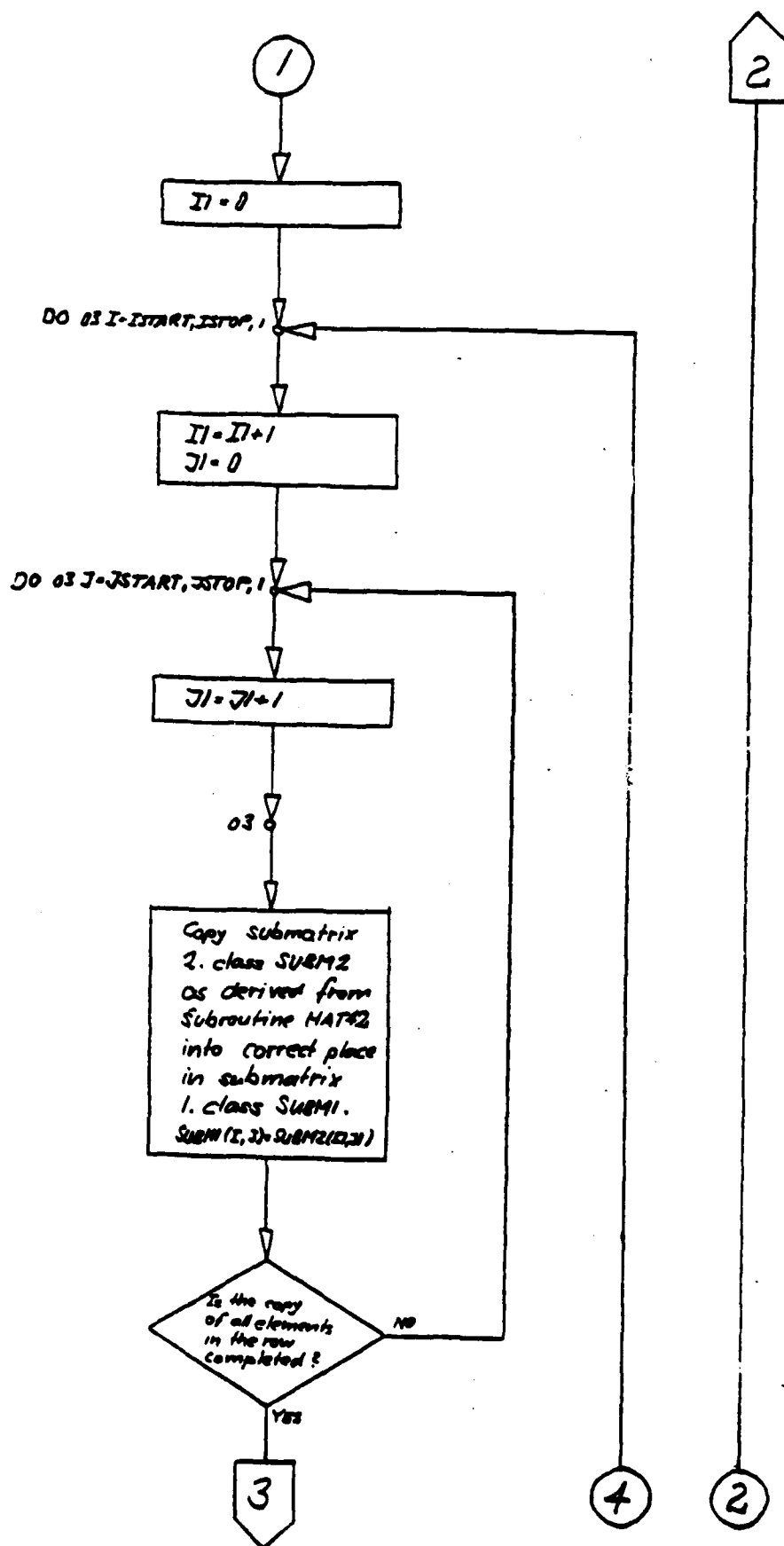


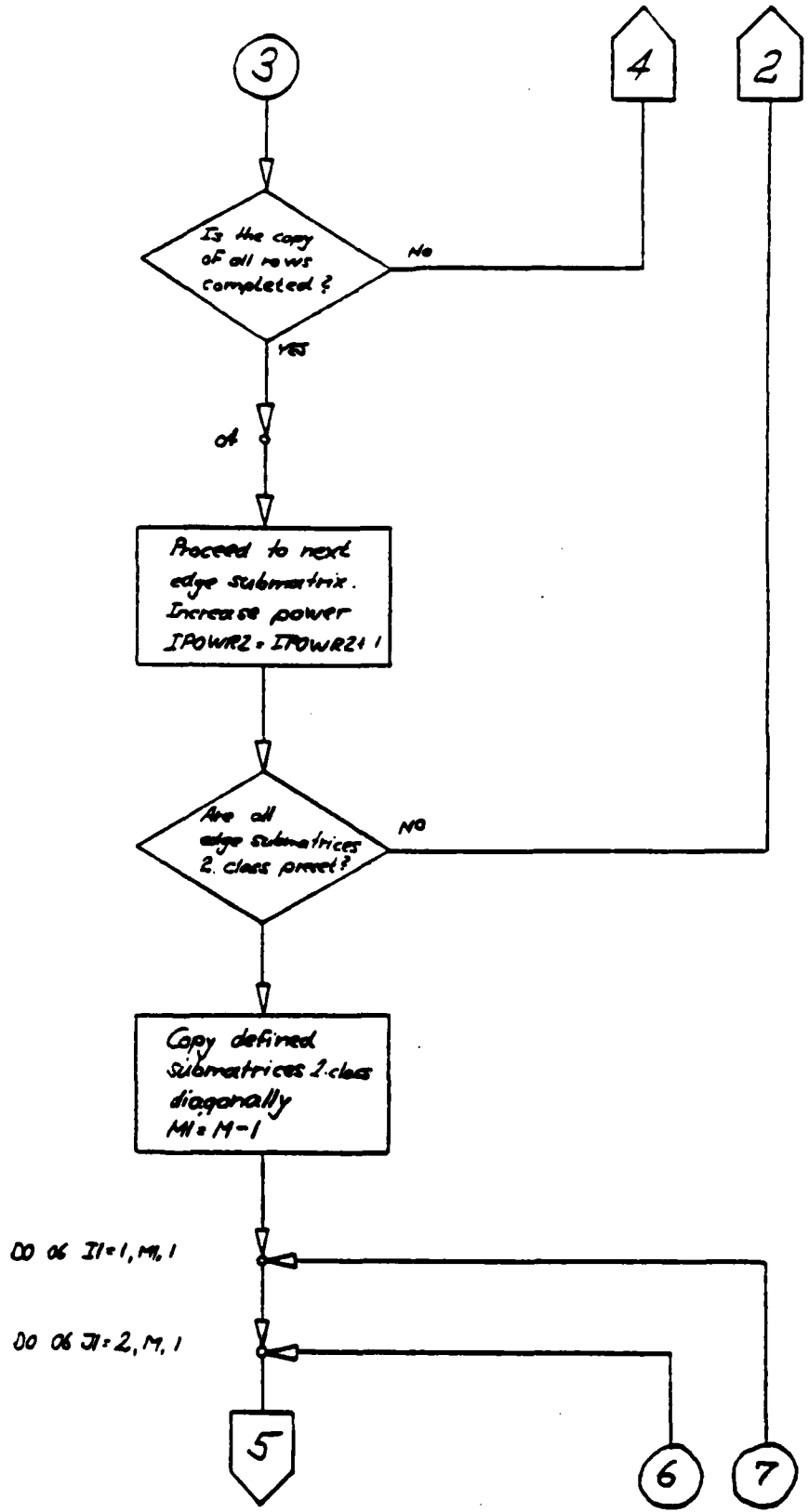




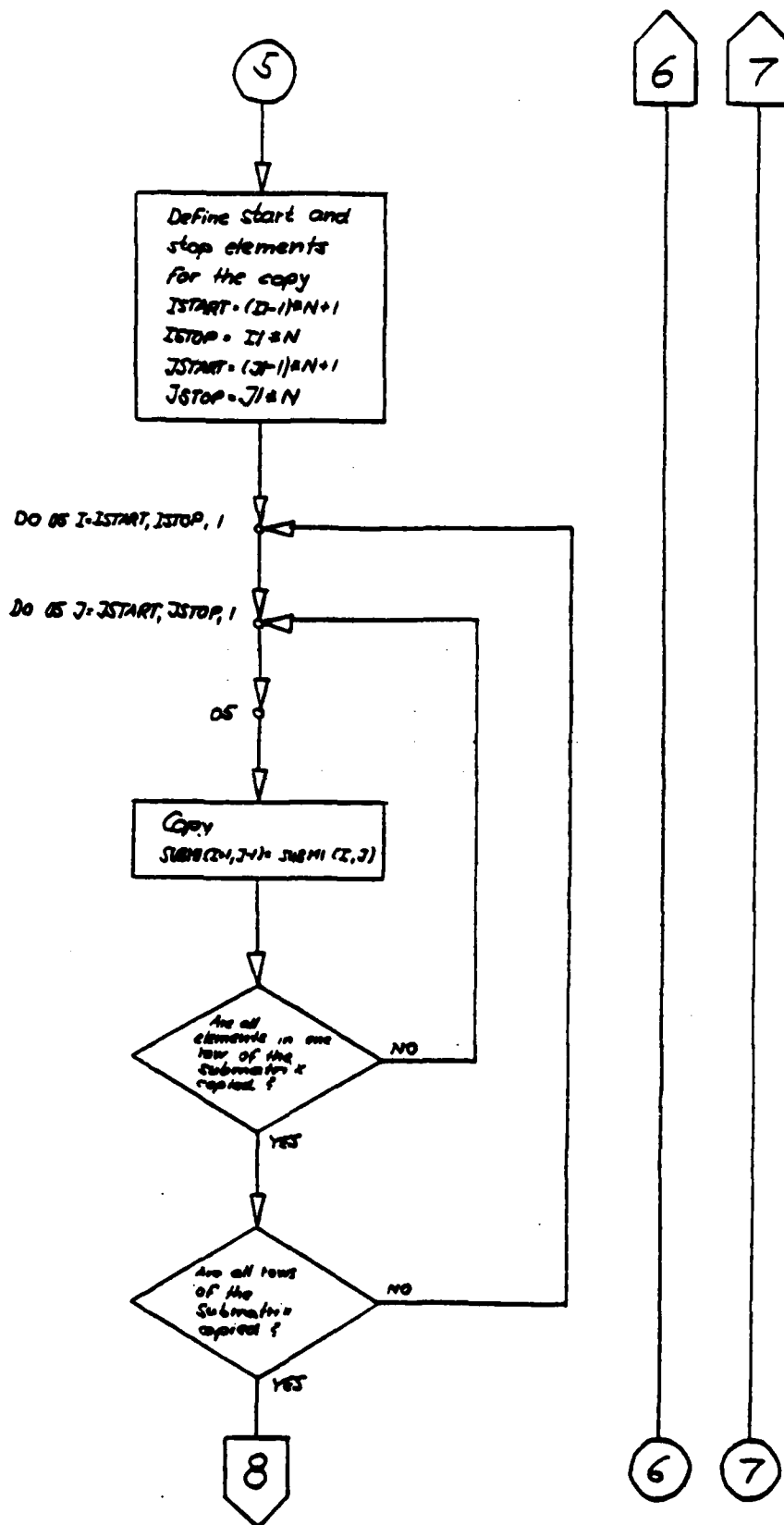
Flow chart MAT41

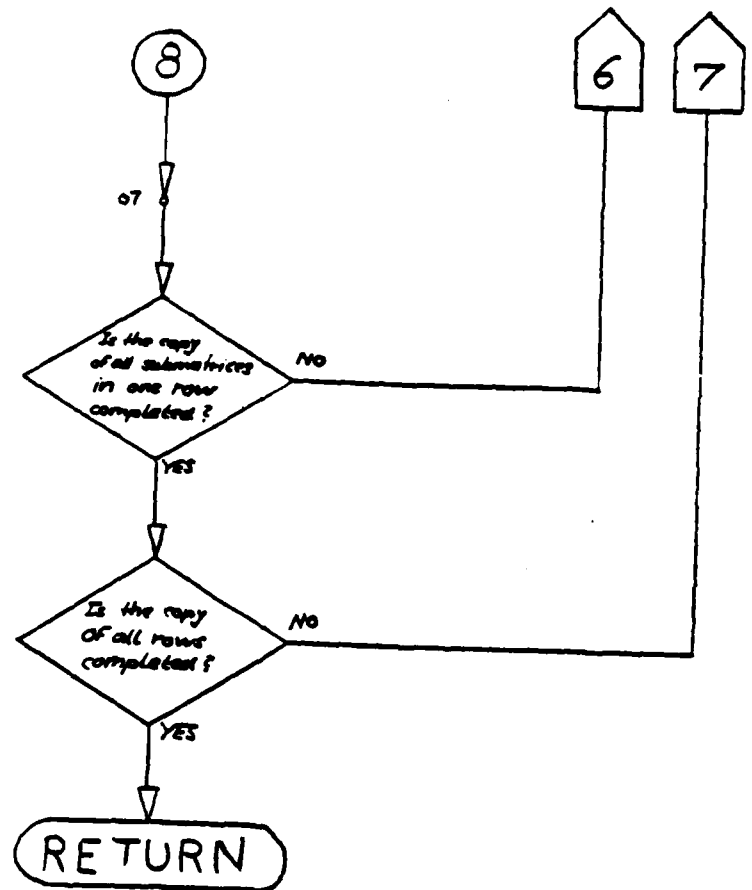




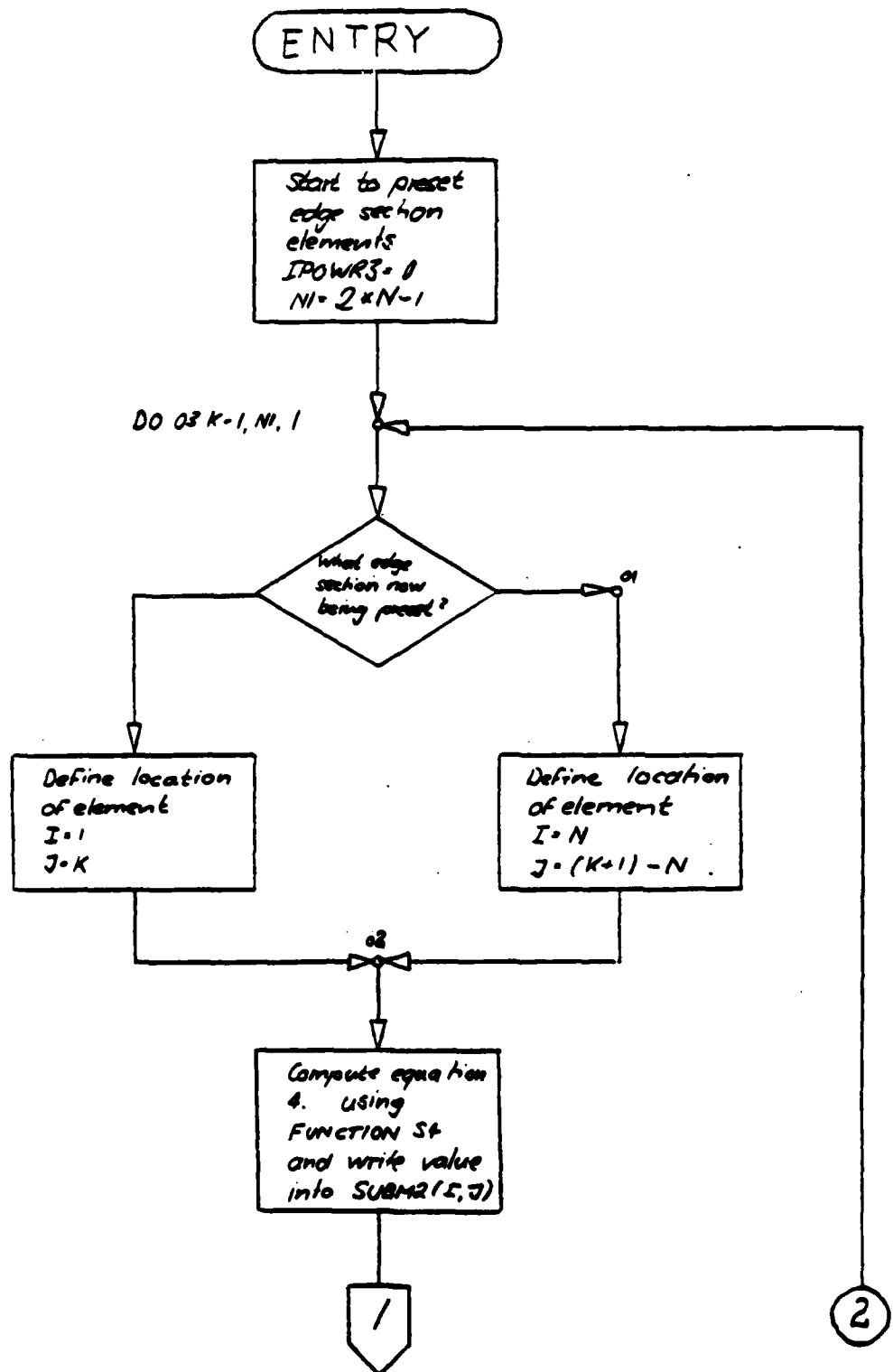


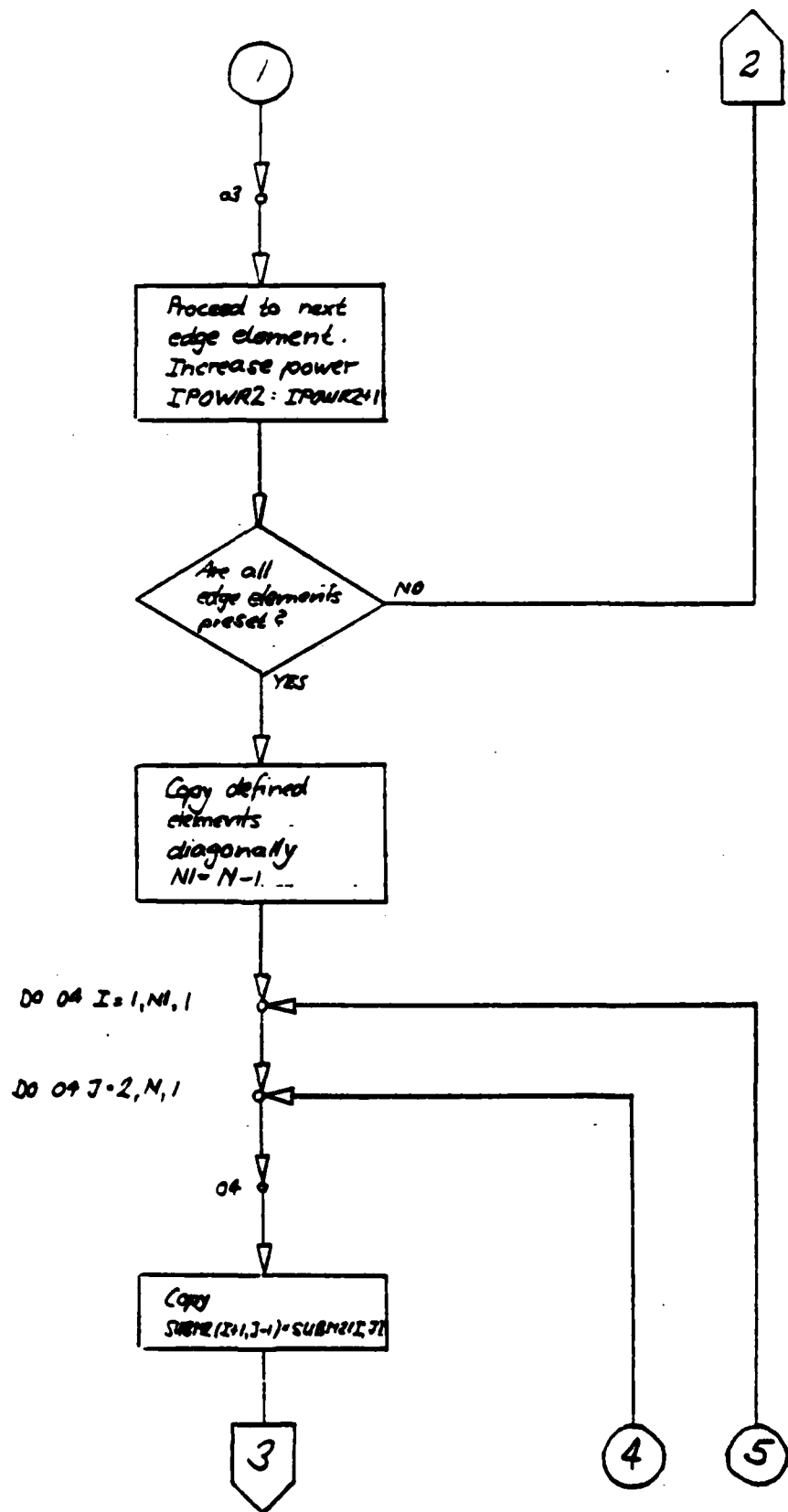


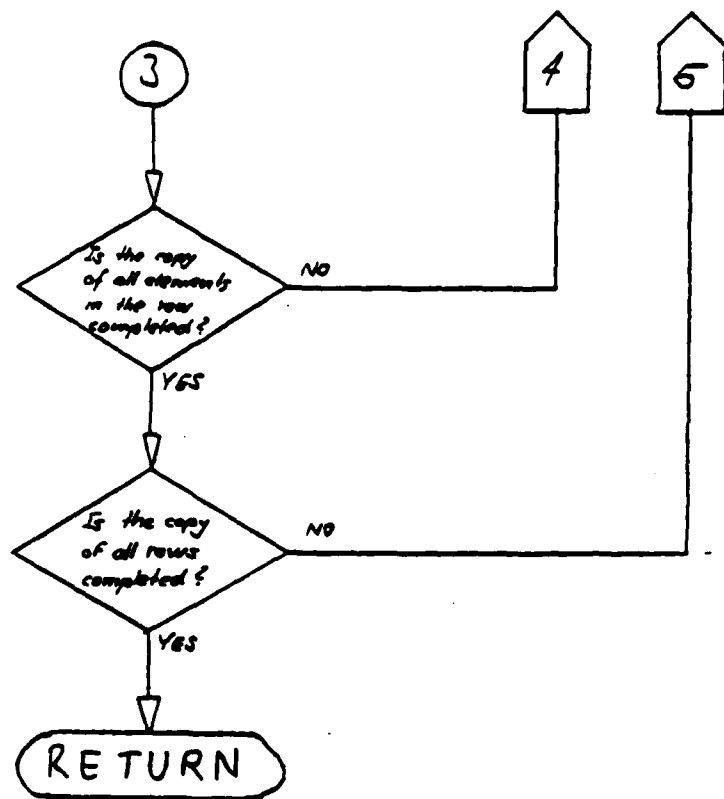




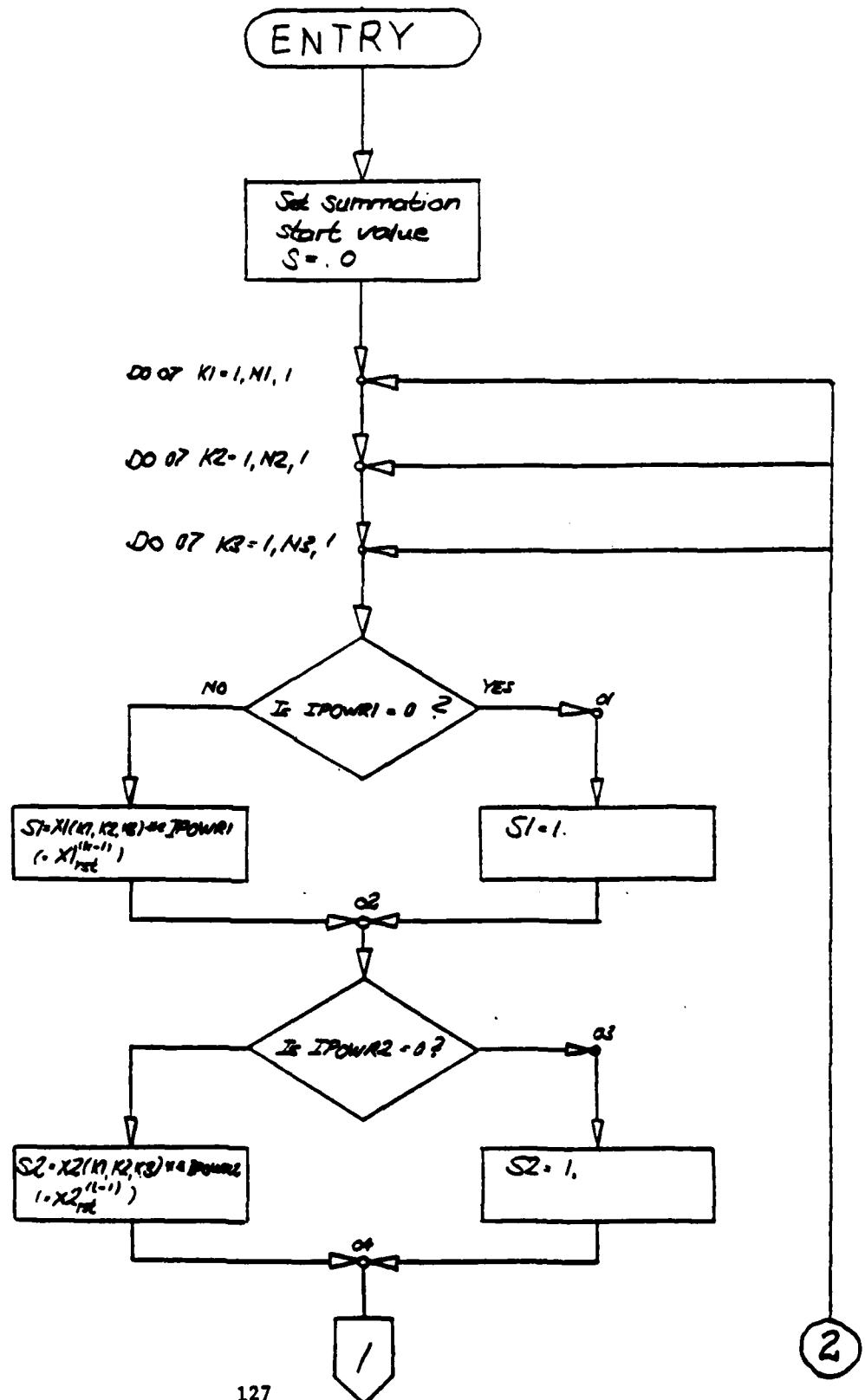
Flow chart MAT42

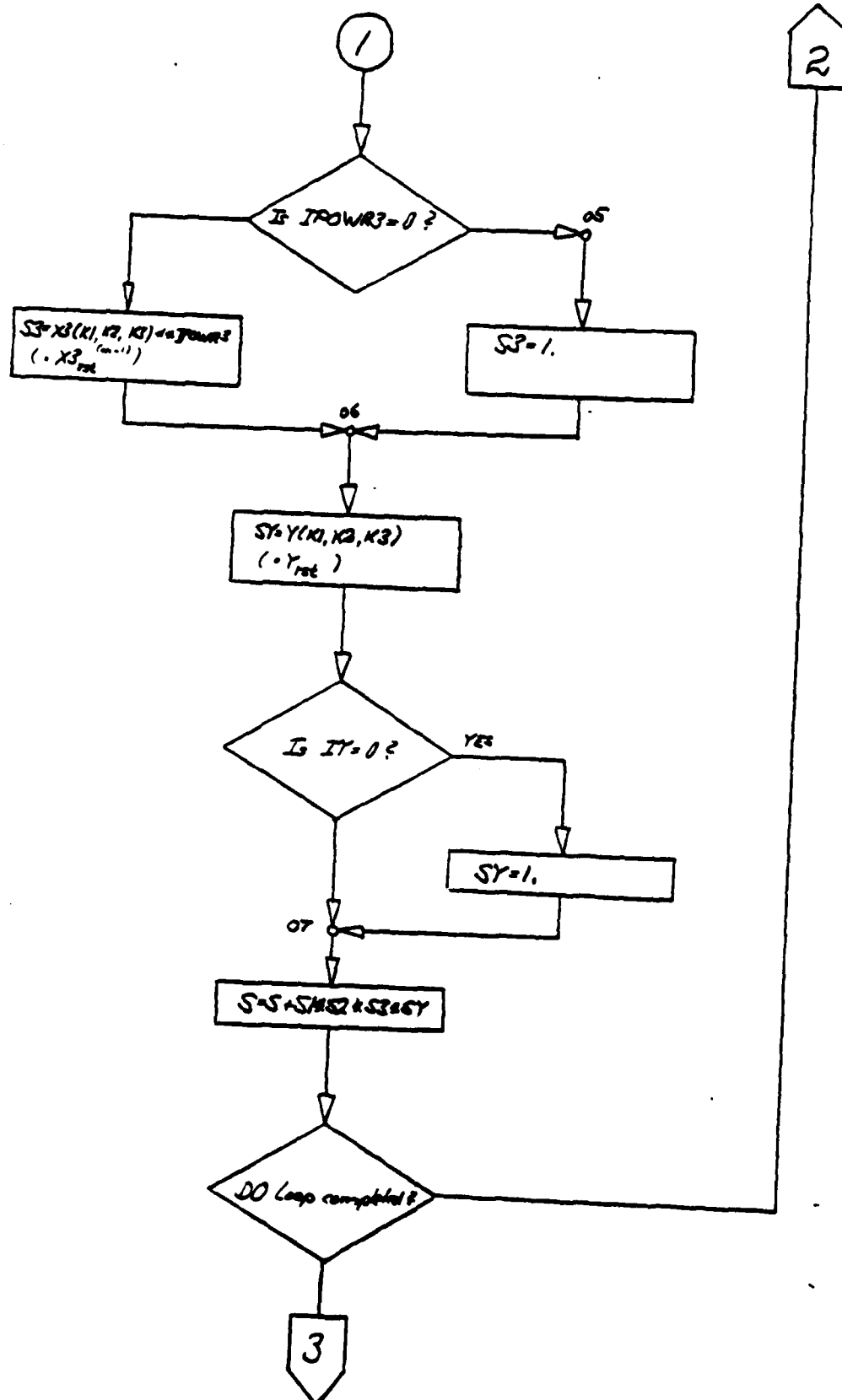


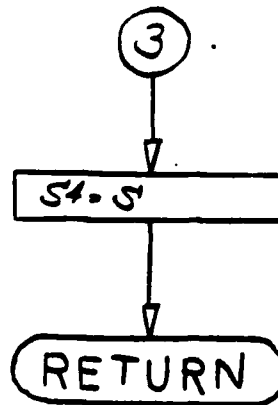




Flow chart S4









APPENDIX C: Software Description: Listings

The following are given:

<u>Listing</u>	<u>Page</u>
<u>Two-Dimensional Approximation</u>	132
BLOCK DATA AB2	132
BLOCK DATA DTA2	133
PROGRAM DEMO2	134
SUBROUTINE MAT2	137
REAL FUNCTION S2	140
INTEGER FUNCTION IEL22	141
SUBROUTINE IEL2	142
REAL FUNCTION F2	145
<u>Three-Dimensional Approximation</u>	146
BLOCK DATA AB3	146
BLOCK DATA DATA3	147
PROGRAM DEMO3	148
SUBROUTINE MAT3	151
SUBROUTINE MAT31	155
REAL FUNCTION S3	156
INTEGER FUNCTION IEL33	157
SUBROUTINE IEL3	158
REAL FUNCTION F3	161
<u>Four-Dimensional Approximation</u>	162
BLOCK DATA AB4	162
BLOCK DATA DATA4	163

Four-Dimensional Approximation (Cont'd)

PROGRAM DEMO4	164
SUBROUTINE MAT4	168
SUBROUTINE MAT41	172
SUBROUTINE MAT42	173
REAL FUNCTION S4	174
INTEGER FUNCTION IEL44	175
SUBROUTINE IEL4	176
REAL FUNCTION F4	179

BLANK DATA 102

74/175 2PT-2 00000

PTN 0.00000

06/12/29 10:02:11

PAGE

1

BLANK DATA 102  
CLIPPER ( 000 ) A+B  
100 200 300 400 500

BLOCK DATA DT42

74/175 107-2 P-000

PTN 4.00000

20/12/10 10.00.00

PAGE

1

BLOCK DATA DT42  
COMMON / DT42 / 11.7  
REAL 11(2500), 7(2500)  
END

134



1546

100

**www**

```

.....
:   TERMINATE GRAPHICS.
.....
IF (IPLBOT) CALL STOP

```

STOP WTT  
END

137



```

      LL=LOGN(ISSN)
      IPRINT=1
      IPRINT=0
      EPSLON=1.0E-07
00      PRINT APPROXIMATION PARAMETERS. IF IPRINT IS GREATER THAN 0.
      NORD=1-1
      IF (IPRINT.LT.0) GO TO 01
      WRITE (LG, 901) NORD, NPTS
      START TO PRESET EDGE SECTION ELEMENTS.
100      01 IPOINT=0
      IPOINT=1
      IF (NPTS.LT.1) GO TO 02
      IF (NPTS.LT.1) GO TO 02
      TOP EDGE SECTION ELEMENTS.
105      I=1
      GO TO 03
      RIGHT HAND EDGE SECTION ELEMENTS.
110      02 I=(N+1)-1
      PRESET EDGE SECTION ELEMENTS IN SYSTEM MATRIX USING I2.
115      03 A(I,J)=2*(NPTS-IPOINT)
      IPOINT=IPOINT+1
120      COPY DEFINED ELEMENTS DIAGONALLY.
      I=1
      GO TO 04
125      04 A(I,J)=A(I,J)
      PRESET RIGHT HAND SIDE VECTOR B.
130      05 B(I)=1-IPOINT
      PRINT SYSTEM MATRIX AND RIGHT HAND SIDE VECTOR B. IF IPRINT
135      IS SET TO 1.
      IF (IPRINT.NE.1) GO TO 06
      WRITE (LG, 902) (J=1,NPTS)
      06 B(I)=1-IPOINT
140      07 WRITE (LG, 903) (I=1,NPTS), (J=1,NPTS), B(I)
      IPRINT=1
145      - GAUSS JORDAN ELIMINATION.
      08 N1=NPTS
      EPSLON=EPSLON/2
      CALL GJELIM(TOTAL, EPSLON)
      IF (ITER.LT.0) GO TO 09
150      PRINT EQUATION SYSTEM, IF IPRINT IS SET TO 1.
      IF (IPRINT.NE.1) GO TO 10
      WRITE (LG, 904) (J=1,NPTS)
      09 WRITE (LG, 905) (I=1,NPTS), (J=1,NPTS), B(I)
155      10 WRITE (LG, 906) (J=1,NPTS)
      DEFINING COEFFICIENTS AND PRINT THEM, IF IPRINT IS GREATER THAN 0.
160      11 WRITE (LG, 907) LLCTP(1)
170      12 WRITE (LG, 907) LLCTP(1)

```

```

      C      STATISTICS, IF ISTAT ANYTHING ELSE BUT 1.
      C      13 IF (ISTAT.EQ.0) GO TO 23
      C      14 IF (ISTAT.EQ.1) GO TO 23
      C      15 IF (ISTAT.EQ.2) GO TO 23
      C      16 ISTAT=ABS(ISTAT)
      C      17 COMPUTE ABSOLUTE ERROR.
      C      18 E=APPROXIMATED VALUE - GIVEN VALUE
      C      19 STAT(I)=2(COEFF(I),NORD(I))-V(I)
      C      20 COMPUTE RELATIVE ERROR, IF ISTAT HAS LESS THAN ZERO, I.E. IERR=1.
      C      21 E=APPROXIMATED VALUE - GIVEN VALUE/(GIVEN VALUE)
      C      22 IF (IERR.NE.0) GO TO 16
      C      23 STAT(I)=STAT(I)/V(I)
      C      24 ERROR STATISTICS: COMPUTE AVERAGE ERROR AND FIND LOCATION
      C      25 OF MAXIMUM ERROR.
      C      26 ERRAVE=0
      C      27 ERRMAX=ABS(STAT(1))
      C      28 IF (IERR.EQ.0)
      C      29 GO TO 17
      C      30 IF (ABS(STAT(I))-ERRMAX) GO TO 17
      C      31 ERRMAX=ABS(STAT(I))
      C      32 ERRAVE=ERRAVE+ABS(STAT(I))
      C      33 IF (ISTAT.NE.0) GO TO 14
      C      34 PRINT 30IMATES IN DATA PRINTS AND APPROXIMATED VALUES.
      C      35 IF /ISTAT/ IS EQUAL 2.
      C      36 WRITE (LO, 000)
      C      37 WRITE (LO, 000) (I, F(I), I=1, NPTS1, 1)
      C      38 WRITE (LO, 000) (I, F(COEFF(I),NORD(I)), I=1, NPTS1, 1)
      C      39 ISTAT=1
      C      40 PRINT ERRORS, IF /ISTAT/ HAS 1 OR 2.
      C      41 IF (ISTAT.NE.1) GO TO 20
      C      42 IF (IERR.NE.0) GO TO 19
      C      43 PRINT ABSOLUTE ERRORS.
      C      44 WRITE (LO, 011)
      C      45 WRITE (LO, 000) (I, STAT(I), I=1, NPTS1, 1)
      C      46 GO TO 20
      C      47 PRINT RELATIVE ERRORS.
      C      48 IF (IERR.EQ.0)
      C      49 WRITE (LO, 012)
      C      50 WRITE (LO, 013) (I, STAT(I), I=1, NPTS1, 1)
      C      51 PRINT ERROR STATISTICS.
      C      52 IF (IERR.EQ.0) WRITE (LO, 013) ERRAVE, INAR, ERRMAX
      C      53 IF (IERR.NE.0) WRITE (LO, 013) ERRAVE, INAR, ERRMAX
      C      54 GO TO 23
      C      55 FATAL ERROR DETECTED. OUTPUT ERROR MESSAGE AND PRESET COEFFICIENTS.
      C      56 IERR=ITER
      C      57 IERR=IERR+1
      C      58 IF (IERR.EQ.1)
      C      59 GO TO 17
      C      60 IF (IERR.EQ.2, N1, 1)
      C      61 WRITE (LO, 101) (I, EPSLON
      C      62 WRITE (LO, 016) IERR, EPSLON
      C      63 GO TO 14
      C      64 RETURN.
      C      65 RETURN

```

```

1      REAL FUNCTION S2 (NPTS1,IPWRI,IV)
2      .....
3      * PERSON SUMMATIONS TO PRESET THE ELEMENTS A(I,J) IN THE
4      * SYSTEM MATRIX A AND THE ELEMENTS B(I) IN THE VECTOR B.
5      * THE VECTOR B FOR THE LINEAR EQUATION SYSTEM THAT MUST
6      * BE SOLVED FOR A POLYNOMIAL 2D-APPROXIMATION.
7      .....
8      *
9      *
10     *
11     *
12     *
13     *
14     *
15     *
16     *
17     *
18     *
19     *
20     *
21     *
22     *
23     *
24     *
25     *
26     *
27     *
28     *
29     *
30     *
31     *
32     *
33     *
34     *
35     *
36     *
37     *
38     *
39     *
40     *
41     *
42     *
43     *
44     *
45     *
46     *
47     *
48     *
49     *
50     *
51     *
52     *
53     *
54     *
55     *
56     *
57     *
58     *
59     *
60     *
61     *
62     *
63     *
64     *
65     *
66     *
67     *
68     *
69     *
70     *
71     *
72     *
73     *
74     *
75     *
76     *
77     *
78     *
79     *
80     *
81     *
82     *
83     *
84     *
85     *
86     *
87     *
88     *
89     *
90     *
91     *
92     *
93     *
94     *
95     *
96     *
97     *
98     *
99     *
100    *
101    *
102    *
103    *
104    *
105    *
106    *
107    *
108    *
109    *
110    *
111    *
112    *
113    *
114    *
115    *
116    *
117    *
118    *
119    *
120    *
121    *
122    *
123    *
124    *
125    *
126    *
127    *
128    *
129    *
130    *
131    *
132    *
133    *
134    *
135    *
136    *
137    *
138    *
139    *
140    *
141    *
142    *
143    *
144    *
145    *
146    *
147    *
148    *
149    *
150    *
151    *
152    *
153    *
154    *
155    *
156    *
157    *
158    *
159    *
160    *
161    *
162    *
163    *
164    *
165    *
166    *
167    *
168    *
169    *
170    *
171    *
172    *
173    *
174    *
175    *
176    *
177    *
178    *
179    *
180    *
181    *
182    *
183    *
184    *
185    *
186    *
187    *
188    *
189    *
190    *
191    *
192    *
193    *
194    *
195    *
196    *
197    *
198    *
199    *
200    *
201    *
202    *
203    *
204    *
205    *
206    *
207    *
208    *
209    *
210    *
211    *
212    *
213    *
214    *
215    *
216    *
217    *
218    *
219    *
220    *
221    *
222    *
223    *
224    *
225    *
226    *
227    *
228    *
229    *
230    *
231    *
232    *
233    *
234    *
235    *
236    *
237    *
238    *
239    *
240    *
241    *
242    *
243    *
244    *
245    *
246    *
247    *
248    *
249    *
250    *
251    *
252    *
253    *
254    *
255    *
256    *
257    *
258    *
259    *
260    *
261    *
262    *
263    *
264    *
265    *
266    *
267    *
268    *
269    *
270    *
271    *
272    *
273    *
274    *
275    *
276    *
277    *
278    *
279    *
280    *
281    *
282    *
283    *
284    *
285    *
286    *
287    *
288    *
289    *
290    *
291    *
292    *
293    *
294    *
295    *
296    *
297    *
298    *
299    *
300    *
301    *
302    *
303    *
304    *
305    *
306    *
307    *
308    *
309    *
310    *
311    *
312    *
313    *
314    *
315    *
316    *
317    *
318    *
319    *
320    *
321    *
322    *
323    *
324    *
325    *
326    *
327    *
328    *
329    *
330    *
331    *
332    *
333    *
334    *
335    *
336    *
337    *
338    *
339    *
340    *
341    *
342    *
343    *
344    *
345    *
346    *
347    *
348    *
349    *
350    *
351    *
352    *
353    *
354    *
355    *
356    *
357    *
358    *
359    *
360    *
361    *
362    *
363    *
364    *
365    *
366    *
367    *
368    *
369    *
370    *
371    *
372    *
373    *
374    *
375    *
376    *
377    *
378    *
379    *
380    *
381    *
382    *
383    *
384    *
385    *
386    *
387    *
388    *
389    *
390    *
391    *
392    *
393    *
394    *
395    *
396    *
397    *
398    *
399    *
400    *
401    *
402    *
403    *
404    *
405    *
406    *
407    *
408    *
409    *
410    *
411    *
412    *
413    *
414    *
415    *
416    *
417    *
418    *
419    *
420    *
421    *
422    *
423    *
424    *
425    *
426    *
427    *
428    *
429    *
430    *
431    *
432    *
433    *
434    *
435    *
436    *
437    *
438    *
439    *
440    *
441    *
442    *
443    *
444    *
445    *
446    *
447    *
448    *
449    *
450    *
451    *
452    *
453    *
454    *
455    *
456    *
457    *
458    *
459    *
460    *
461    *
462    *
463    *
464    *
465    *
466    *
467    *
468    *
469    *
470    *
471    *
472    *
473    *
474    *
475    *
476    *
477    *
478    *
479    *
480    *
481    *
482    *
483    *
484    *
485    *
486    *
487    *
488    *
489    *
490    *
491    *
492    *
493    *
494    *
495    *
496    *
497    *
498    *
499    *
500    *
501    *
502    *
503    *
504    *
505    *
506    *
507    *
508    *
509    *
510    *
511    *
512    *
513    *
514    *
515    *
516    *
517    *
518    *
519    *
520    *
521    *
522    *
523    *
524    *
525    *
526    *
527    *
528    *
529    *
530    *
531    *
532    *
533    *
534    *
535    *
536    *
537    *
538    *
539    *
540    *
541    *
542    *
543    *
544    *
545    *
546    *
547    *
548    *
549    *
550    *
551    *
552    *
553    *
554    *
555    *
556    *
557    *
558    *
559    *
560    *
561    *
562    *
563    *
564    *
565    *
566    *
567    *
568    *
569    *
570    *
571    *
572    *
573    *
574    *
575    *
576    *
577    *
578    *
579    *
580    *
581    *
582    *
583    *
584    *
585    *
586    *
587    *
588    *
589    *
590    *
591    *
592    *
593    *
594    *
595    *
596    *
597    *
598    *
599    *
600    *
601    *
602    *
603    *
604    *
605    *
606    *
607    *
608    *
609    *
610    *
611    *
612    *
613    *
614    *
615    *
616    *
617    *
618    *
619    *
620    *
621    *
622    *
623    *
624    *
625    *
626    *
627    *
628    *
629    *
630    *
631    *
632    *
633    *
634    *
635    *
636    *
637    *
638    *
639    *
640    *
641    *
642    *
643    *
644    *
645    *
646    *
647    *
648    *
649    *
650    *
651    *
652    *
653    *
654    *
655    *
656    *
657    *
658    *
659    *
660    *
661    *
662    *
663    *
664    *
665    *
666    *
667    *
668    *
669    *
670    *
671    *
672    *
673    *
674    *
675    *
676    *
677    *
678    *
679    *
680    *
681    *
682    *
683    *
684    *
685    *
686    *
687    *
688    *
689    *
690    *
691    *
692    *
693    *
694    *
695    *
696    *
697    *
698    *
699    *
700    *
701    *
702    *
703    *
704    *
705    *
706    *
707    *
708    *
709    *
710    *
711    *
712    *
713    *
714    *
715    *
716    *
717    *
718    *
719    *
720    *
721    *
722    *
723    *
724    *
725    *
726    *
727    *
728    *
729    *
730    *
731    *
732    *
733    *
734    *
735    *
736    *
737    *
738    *
739    *
740    *
741    *
742    *
743    *
744    *
745    *
746    *
747    *
748    *
749    *
750    *
751    *
752    *
753    *
754    *
755    *
756    *
757    *
758    *
759    *
760    *
761    *
762    *
763    *
764    *
765    *
766    *
767    *
768    *
769    *
770    *
771    *
772    *
773    *
774    *
775    *
776    *
777    *
778    *
779    *
780    *
781    *
782    *
783    *
784    *
785    *
786    *
787    *
788    *
789    *
790    *
791    *
792    *
793    *
794    *
795    *
796    *
797    *
798    *
799    *
800    *
801    *
802    *
803    *
804    *
805    *
806    *
807    *
808    *
809    *
810    *
811    *
812    *
813    *
814    *
815    *
816    *
817    *
818    *
819    *
820    *
821    *
822    *
823    *
824    *
825    *
826    *
827    *
828    *
829    *
830    *
831    *
832    *
833    *
834    *
835    *
836    *
837    *
838    *
839    *
840    *
841    *
842    *
843    *
844    *
845    *
846    *
847    *
848    *
849    *
850    *
851    *
852    *
853    *
854    *
855    *
856    *
857    *
858    *
859    *
860    *
861    *
862    *
863    *
864    *
865    *
866    *
867    *
868    *
869    *
870    *
871    *
872    *
873    *
874    *
875    *
876    *
877    *
878    *
879    *
880    *
881    *
882    *
883    *
884    *
885    *
886    *
887    *
888    *
889    *
890    *
891    *
892    *
893    *
894    *
895    *
896    *
897    *
898    *
899    *
900    *
901    *
902    *
903    *
904    *
905    *
906    *
907    *
908    *
909    *
910    *
911    *
912    *
913    *
914    *
915    *
916    *
917    *
918    *
919    *
920    *
921    *
922    *
923    *
924    *
925    *
926    *
927    *
928    *
929    *
930    *
931    *
932    *
933    *
934    *
935    *
936    *
937    *
938    *
939    *
940    *
941    *
942    *
943    *
944    *
945    *
946    *
947    *
948    *
949    *
950    *
951    *
952    *
953    *
954    *
955    *
956    *
957    *
958    *
959    *
960    *
961    *
962    *
963    *
964    *
965    *
966    *
967    *
968    *
969    *
970    *
971    *
972    *
973    *
974    *
975    *
976    *
977    *
978    *
979    *
980    *
981    *
982    *
983    *
984    *
985    *
986    *
987    *
988    *
989    *
990    *
991    *
992    *
993    *
994    *
995    *
996    *
997    *
998    *
999    *
1000   *
1001   *
1002   *
1003   *
1004   *
1005   *
1006   *
1007   *
1008   *
1009   *
1010   *
1011   *
1012   *
1013   *
1014   *
1015   *
1016   *
1017   *
1018   *
1019   *
1020   *
1021   *
1022   *
1023   *
1024   *
1025   *
1026   *
1027   *
1028   *
1029   *
1030   *
1031   *
1032   *
1033   *
1034   *
1035   *
1036   *
1037   *
1038   *
1039   *
1040   *
1041   *
1042   *
1043   *
1044   *
1045   *
1046   *
1047   *
1048   *
1049   *
1050   *
1051   *
1052   *
1053   *
1054   *
1055   *
1056   *
1057   *
1058   *
1059   *
1060   *
1061   *
1062   *
1063   *
1064   *
1065   *
1066   *
1067   *
1068   *
1069   *
1070   *
1071   *
1072   *
1073   *
1074   *
1075   *
1076   *
1077   *
1078   *
1079   *
1080   *
1081   *
1082   *
1083   *
1084   *
1085   *
1086   *
1087   *
1088   *
1089   *
1090   *
1091   *
1092   *
1093   *
1094   *
1095   *
1096   *
1097   *
1098   *
1099   *
1100   *
1101   *
1102   *
1103   *
1104   *
1105   *
1106   *
1107   *
1108   *
1109   *
1110   *
1111   *
1112   *
1113   *
1114   *
1115   *
1116   *
1117   *
1118   *
1119   *
1120   *
1121   *
1122   *
1123   *
1124   *
1125   *
1126   *
1127   *
1128   *
1129   *
1130   *
1131   *
1132   *
1133   *
1134   *
1135   *
1136   *
1137   *
1138   *
1139   *
1140   *
1141   *
1142   *
1143   *
1144   *
1145   *
1146   *
1147   *
1148   *
1149   *
1150   *
1151   *
1152   *
1153   *
1154   *
1155   *
1156   *
1157   *
1158   *
1159   *
1160   *
1161   *
1162   *
1163   *
1164   *
1165   *
1166   *
1167   *
1168   *
1169   *
1170   *
1171   *
1172   *
1173   *
1174   *
1175   *
1176   *
1177   *
1178   *
1179   *
1180   *
1181   *
1182   *
1183   *
1184   *
1185   *
1186   *
1187   *
1188   *
1189   *
1190   *
1191   *
1192   *
1193   *
1194   *
1195   *
1196   *
1197   *
1198   *
1199   *
1200   *
1201   *
1202   *
1203   *
1204   *
1205   *
1206   *
1207   *
1208   *
1209   *
1210   *
1211   *
1212   *
1213   *
1214   *
1215   *
1216   *
1217   *
1218   *
1219   *
1220   *
1221   *
1222   *
1223   *
1224   *
1225   *
1226   *
1227   *
1228   *
1229   *
1230   *
1231   *
1232   *
1233   *
1234   *
1235   *
1236   *
1237   *
1238   *
1239   *
1240   *
1241   *
1242   *
1243   *
1244   *
1245   *
1246   *
1247   *
1248   *
1249   *
1250   *
1251   *
1252   *
1253   *
1254   *
1255   *
1256   *
1257   *
1258   *
1259   *
1260   *
1261   *
1262   *
1263   *
1264   *
1265   *
1266   *
1267   *
1268   *
1269   *
1270   *
1271   *
1272   *
1273   *
1274   *
1275   *
1276   *
1277   *
1278   *
1279   *
1280   *
1281   *
1282   *
1283   *
1284   *
1285   *
1286   *
1287   *
1288   *
1289   *
1290   *
1291   *
1292   *
1293   *
1294   *
1295   *
1296   *
1297   *
1298   *
1299   *
1300   *
1301   *
1302   *
1303   *
1304   *
1305   *
1306   *
1307   *
1308   *
1309   *
1310   *
1311   *
1312   *
1313   *
1314   *
1315   *
1316   *
1317   *
1318   *
1319   *
1320   *
1321   *
1322   *
1323   *
1324   *
1325   *
1326   *
1327   *
1328   *
1329   *
1330   *
1331   *
1332   *
1333   *
1334   *
1335   *
1336   *
1337   *
1338   *
1339   *
1340   *
1341   *
1342   *
1343   *
1344   *
1345   *
1346   *
1347   *
1348   *
1349   *
1350   *
1351   *
1352   *
1353   *
1354   *
1355   *
1356   *
1357   *
1358   *
1359   *
1360   *
1361   *
1362   *
1363   *
1364   *
1365   *
1366   *
1367   *
1368   *
1369   *
1370   *
1371   *
1372   *
1373   *
1374   *
1375   *
1376   *
1377   *
1378   *
1379   *
1380   *
1381   *
1382   *
1383   *
1384   *
1385   *
1386   *
1387   *
1388   *
1389   *
1390   *
1391   *
1392   *
1393   *
1394   *
1395   *
1396   *
1397   *
1398   *
1399   *
1400   *
1401   *
1402   *
1403   *
1404   *
1405   *
1406   *
1407   *
1408   *
1409   *
1410   *
1411   *
1412   *
1413   *
1414   *
1415   *
1416   *
1417   *
1418   *
1419   *
1420   *
1421   *
1422   *
1423   *
1424   *
1425   *
1426   *
1427   *
1428   *
1429   *
1430   *
1431   *
1432   *
1433   *
1434   *
1435   *
1436   *
1437   *
1438   *
1439   *
1440   *
1441   *
1442   *
1443   *
1444   *
1445   *
1446   *
1447   *
1448   *
1449   *
1450   *
1451   *
1452   *
1453   *
1454   *
1455   *
1456   *
1457   *
1458   *
1459   *
1460   *
1461   *
1462   *
1463   *
1464   *
1465   *
1466   *
1467   *
1468   *
1469   *
1470   *
1471   *
1472   *
1473   *
1474   *
1475   *
1476   *
1477   *
1478   *
1479   *
1480   *
1481   *
1482   *
1483   *
1484   *
1485   *
1486   *
1487   *
1488   *
1489   *
1490   *
1491   *
1492   *
1493   *
1494   *
1495   *
1496   *
1497   *
1498   *
1499   *
1500   *
1501   *
1502   *
1503   *
1504   *
1505   *
1506   *
1507   *
1508   *
1509   *
1510   *
1511   *
1512   *
1513   *
1514   *
1515   *
1516   *
1517   *
1518   *
1519   *
1520   *
1521   *
1522   *
1523   *
1524   *
1525   *
1526   *
1527   *
1528   *
1529   *
1530   *
1531   *
1532   *
1533   *
1534   *
1535   *
1536   *
1537   *
1538   *
1539   *
1540   *
1541   *
1542   *
1543   *
1544   *
1545   *
1546   *
1547   *
1548   *
1549   *
1550   *
1551   *
1552   *
1553   *
1554   *
1555   *
1556   *
1557   *
1558   *
1559   *
1560   *
1561   *
1562   *
1563   *
1564   *
1565   *
1566   *
1567   *
1568   *
1569   *
1570   *
1571   *
1572   *
1573   *
1574   *
1575   *
1576   *
1577   *
1578   *
1579   *
1580   *
1581   *
1582   *
1583   *
1584   *
1585   *
1586   *
1587   *
1588   *
1589   *
1590   *
1591   *
1592   *
1593   *
1594   *
1595   *
1596   *
1597   *
1598   *
1599   *
1600   *
1601   *
1602   *
1603   *
1604   *
1605   *
1606   *
1607   *
1608   *
1609   *
1610   *
1611   *
1612   *
1613   *
1614   *
1615   *
1616   *
1617   *
1618   *
1619   *
1620   *
1621   *
1622   *
1623   *
1624   *
1625   *
1626   *
1627   *
1628   *
1629   *
1630   *
1631   *
1632   *
1633   *
1634   *
1635   *
1636   *
1637   *
1638   *
1639   *
1640   *
1641   *
1642   *
1643   *
1644   *
1645   *
1646   *
1647   *
1648   *
1649   *
1650   *
1651   *
1652   *
1653   *
1654   *
1655   *
1656   *
1657   *
1658   *
1659   *
1660   *
1661   *
1662   *
1663   *
1664   *
1665   *
1666   *
1667   *
1668   *
1669   *
1670   *
1671   *
1672   *
1673   *
1674   *
1675   *
1676   *
1677   *
1678   *
1679   *
1680   *
1681   *
1682   *
1683   *
1684   *
1685   *
1686   *
1687   *
1688   *
1689   *
1690   *
1691   *
1692   *
1693   *
1694   *
1695   *
1696   *
1697   *
1698   *
1699   *
1700   *
1701   *
1702   *
1703   *
1704   *
1705   *
1706   *
1707   *
1708   *
1709   *
1710   *
1711   *
1712   *
1713   *
1714   *
1715   *
1716   *
1717   *
1718   *
1719   *
1720   *
1721   *
1722   *
1723   *
1724   *
1725   *
1726   *
1727   *
1728   *
1729   *
1730   *
1731   *
1732   *
1733   *
1734   *
1735   *
1736   *
1737   *
1738   *
1739   *
1740   *
1741   *
1742   *
1743   *
1744   *
1745   *
1746   *
1747   *
1748   *
1749   *
1750   *
1751   *
1752   *
1753   *
1754   *
1755   *
1756   *
1757   *
1758   *
1759   *
1760   *
1761   *
1762   *
1763   *
1764   *
1765   *
1766   *
1767   *
1768   *
1769   *
1770   *
1771   *
1772   *
1773   *
1774   *
1775   *
1776   *
1777   *
1778   *
1779   *
1780   *
1781   *
1782   *
1783   *
1784   *
1785   *
1786   *
1787   *
1788   *
1789   *
1790   *
1791   *
1792   *
1793   *
1794   *
1795   *
1796   *
1797   *
1798   *
1799   *
1800   *
1801   *
1802   *
1803   *
1804   *
1805   *
1806   *
1807   *
1808   *
1809   *
1810   *
1811   *
1812   *
1813   *
1814   *
1815   *
1816   *
1817   *
1818   *
1819   *
1820   *
1821   *
1822   *
1823   *
1824   *
1825   *
1826   *
1827   *
1828   *
1829   *
1830   *
1831   *
1832   *
1833   *
1834   *
1835   *
1836   *
1837   *
1838   *
1839   *
1840   *
1841   *
1842   *
1843   *
1844   *
1845   *
1846   *
1847   *
1848   *
1849   *
1850   *
1851   *
1852   *
1853   *
1854   *
1855   *
1856   *
1857   *
1858   *
1859   *
1860   *
1861   *
1862   *
1863   *
1864   *
1865   *
1866   *
1867   *
1868   *
1869   *
1870   *
1871   *
1872   *
1873   *
1874   *
1875   *
1876   *
1877   *
1878   *
1879   *
1880   *
1881   *
1882   *
1883   *
1884   *
1885   *
1886   *
1887   *
188
```

C

142

• 2405

143

1944/45 - 1945/46

144

FUNCTION F2

7/175 OPT-2 PROG

RTN 440000

CL/22/20 10-4000

FORM

```

1      REAL FUNCTION F2 (CJEP, XI, NJORDI)
2
3      .....
4      COMPUTE FUNCTION VALUE Y=F(XI) OF A 2D-POLYNOMIAL USING
5      THE HORNED SCHEME IN ORDER TO INCREASE SPEED AND ACCURACY.
6
7      AUTHOR: JAMES ZERNER
8      DATE: DECEMBER 20, 1988
9
10     VARIABLE EXPLANATION
11     CJEP(1) 2D-POLYNOMIAL COEFFICIENTS
12     XI       X-VALUE WHERE Y WILL BE COMPUTED.
13     NJORDI   ORDER OF THE POLYNOMIAL IN THE
14             XI DIRECTION
15     F2       ORDINATE OF THE 2D-POLYNOMIAL.
16
17     REAL CJEP(1)
18     REAL F2
19     REAL XI
20     REAL NJORDI
21
22     F2 = 0.0
23     DO 10 I = 1, NJORDI
24       F2 = F2 + CJEP(I) * XI ** I
25     RETURN
26     END

```



BLK DATA 483

70/175 JPT-2 P1000

FTN 0000000

20/22/270 100 0000

Page

1

BLK DATA 483  
COMMON / 483 / 483  
REAL 24000000, 00000000

BLOCK DATA DATAS 76/175 3PT=2 P4000

FIN 400000

00/16/200 100.0000

P80

1

BLOCK DATA DATAS  
CHROM 7 00123 7 11022-7  
FILE 11100101021100101071100101  
END



```

      READ MEASURED DATA FROM TAPE 9 AND SIMULTANEOUSLY SEARCH FOR
      MINIMUM AND MAXIMUM VALUE OF EACH (X, Y AND Z) (RECORDED T)
      GET THE PLOT WITHIN THE BOUNDARIES OF A SHEET OF PAPER.
    .....
90  READ (LD, 901) VAL, IRAN
    .....
95  READ (LD, 902) VAL, IRAN
    .....
100  READ (LD, 903) VAL, IRAN
    .....
105  READ (LD, 904) VAL, IRAN
    .....
110  READ (LD, 905) VAL, IRAN
    .....
115  WRITE (LI, 103)
    .....
120  .....
125  .....
130  .....
135  .....
140  .....
145  .....
150  .....
155  .....
160  .....
165  .....
170  .....

```

```

C .....
179 00 00 .....
04 00 00 .....
180 00 00 .....
05 00 00 .....
WRITE (1, 106)

189 .....
C .....
190 .....
06 .....
READ (5, *) NORM1, NORM2, IPRINT, ISTAT
NORM1 = NORM1
NORM2 = NORM2
IPRINT = 0
ISTAT = 0
NORM1 = 1
NORM2 = 1
IF (L-1) 1, 2, 3, 4
IF (L-1) 1, 2, 3, 4
CALL MATR (IPRINT, ISTAT, SUB1, ISTAT, SUB1, ISTAT)
WRITE (1, 106)

210 .....
C .....
219 .....
07 .....
08 .....
09 .....
220 .....
00 .....
225 .....
09 .....
230 .....
C .....
239 .....
IF (IPRINT) CALL STOP

240 STOP 0577
END

```

151

```

012 FORMAT (///35H RELATIVE ERROR AT EACH POINT IN 1/1
013 FORMAT (///35H STATISTICS: 1. MAXIMUM ERROR, 2. AVERAGE ERROR, 3. MAX. 1/ST
014 FORMAT (///35H STATISTICS: 4. MIN. 1/ST, 5. MAX. 2/ST, 6. MAX. 3/ST
015 FORMAT (///35H MAXIMUM ERROR AT 1. 1/ST, 2. 2/ST, 3. 3/ST, 4. 4/ST, 5. 5/ST
C
016 FORMAT (///35H AFTER 1000 STEPS NO REDUCTION WITHIN THE ACCURACY
017 FORMAT (///35H QUIT)
C
018 FORMAT (///35H)
019
020
021
022
023
024
025
026
027
028
029
030
031
032
033
034
035
036
037
038
039
040
041
042
043
044
045
046
047
048
049
050
051
052
053
054
055
056
057
058
059
060
061
062
063
064
065
066
067
068
069
070
071
072
073
074
075
076
077
078
079
080
081
082
083
084
085
086
087
088
089
090
091
092
093
094
095
096
097
098
099
100
101
102
103
104
105
106
107
108
109
110
111
112
113
114
115
116
117
118
119
120
121
122
123
124
125
126
127
128
129
130
131
132
133
134
135
136
137
138
139
140
141
142
143
144
145
146
147
148
149
150
151
152
153
154
155
156
157
158
159
160
161
162
163
164
165
166
167
168
169
170
171
172
173
174
175
176
177
178
179
180
181
182
183
184
185
186
187
188
189
190
191
192
193
194
195
196
197
198
199
200
201
202
203
204
205
206
207
208
209
210
211
212
213
214
215
216
217
218
219
220
221
222
223
224
225
226
227
228
229
230
231
232
233
234
235
236
237
238
239
240
241
242
243
244
245
246
247
248
249
250
251
252
253
254
255
256
257
258
259
260
261
262
263
264
265
266
267
268
269
270
271
272
273
274
275
276
277
278
279
280
281
282
283
284
285
286
287
288
289
290
291
292
293
294
295
296
297
298
299
300
301
302
303
304
305
306
307
308
309
310
311
312
313
314
315
316
317
318
319
320
321
322
323
324
325
326
327
328
329
330
331
332
333
334
335
336
337
338
339
340
341
342
343
344
345
346
347
348
349
350
351
352
353
354
355
356
357
358
359
360
361
362
363
364
365
366
367
368
369
370
371
372
373
374
375
376
377
378
379
380
381
382
383
384
385
386
387
388
389
390
391
392
393
394
395
396
397
398
399
400
401
402
403
404
405
406
407
408
409
410
411
412
413
414
415
416
417
418
419
420
421
422
423
424
425
426
427
428
429
430
431
432
433
434
435
436
437
438
439
440
441
442
443
444
445
446
447
448
449
450
451
452
453
454
455
456
457
458
459
460
461
462
463
464
465
466
467
468
469
470
471
472
473
474
475
476
477
478
479
480
481
482
483
484
485
486
487
488
489
490
491
492
493
494
495
496
497
498
499
500
501
502
503
504
505
506
507
508
509
510
511
512
513
514
515
516
517
518
519
520
521
522
523
524
525
526
527
528
529
530
531
532
533
534
535
536
537
538
539
540
541
542
543
544
545
546
547
548
549
550
551
552
553
554
555
556
557
558
559
560
561
562
563
564
565
566
567
568
569
570
571
572
573
574
575
576
577
578
579
580
581
582
583
584
585
586
587
588
589
590
591
592
593
594
595
596
597
598
599
600
601
602
603
604
605
606
607
608
609
610
611
612
613
614
615
616
617
618
619
620
621
622
623
624
625
626
627
628
629
630
631
632
633
634
635
636
637
638
639
640
641
642
643
644
645
646
647
648
649
650
651
652
653
654
655
656
657
658
659
660
661
662
663
664
665
666
667
668
669
670
671
672
673
674
675
676
677
678
679
680
681
682
683
684
685
686
687
688
689
690
691
692
693
694
695
696
697
698
699
700
701
702
703
704
705
706
707
708
709
710
711
712
713
714
715
716
717
718
719
720
721
722
723
724
725
726
727
728
729
730
731
732
733
734
735
736
737
738
739
740
741
742
743
744
745
746
747
748
749
750
751
752
753
754
755
756
757
758
759
760
761
762
763
764
765
766
767
768
769
770
771
772
773
774
775
776
777
778
779
780
781
782
783
784
785
786
787
788
789
790
791
792
793
794
795
796
797
798
799
800
801
802
803
804
805
806
807
808
809
810
811
812
813
814
815
816
817
818
819
820
821
822
823
824
825
826
827
828
829
830
831
832
833
834
835
836
837
838
839
840
841
842
843
844
845
846
847
848
849
850
851
852
853
854
855
856
857
858
859
860
861
862
863
864
865
866
867
868
869
870
871
872
873
874
875
876
877
878
879
880
881
882
883
884
885
886
887
888
889
890
891
892
893
894
895
896
897
898
899
900
901
902
903
904
905
906
907
908
909
910
911
912
913
914
915
916
917
918
919
920
921
922
923
924
925
926
927
928
929
930
931
932
933
934
935
936
937
938
939
940
941
942
943
944
945
946
947
948
949
950
951
952
953
954
955
956
957
958
959
960
961
962
963
964
965
966
967
968
969
970
971
972
973
974
975
976
977
978
979
980
981
982
983
984
985
986
987
988
989
990
991
992
993
994
995
996
997
998
999
1000

```

```

175      WRITE (LO, 602) (J=1,NM1)
180      WRITE (LO, 603) (J=1,NM1)
190      WRITE (LO, 604) (J=1,NM1)
200      WRITE (LO, 605) (J=1,NM1)
210      WRITE (LO, 606) (J=1,NM1)
220      WRITE (LO, 607) (J=1,NM1)
230      WRITE (LO, 608) (J=1,NM1)
240      WRITE (LO, 609) (J=1,NM1)
250      WRITE (LO, 610) (J=1,NM1)
260      WRITE (LO, 611) (J=1,NM1)
270      WRITE (LO, 612) (J=1,NM1)
280      WRITE (LO, 613) (J=1,NM1)
290      WRITE (LO, 614) (J=1,NM1)
300      WRITE (LO, 615) (J=1,NM1)
310      WRITE (LO, 616) (J=1,NM1)
320      WRITE (LO, 617) (J=1,NM1)
330      WRITE (LO, 618) (J=1,NM1)
340      WRITE (LO, 619) (J=1,NM1)
350      WRITE (LO, 620) (J=1,NM1)
360      WRITE (LO, 621) (J=1,NM1)
370      WRITE (LO, 622) (J=1,NM1)
380      WRITE (LO, 623) (J=1,NM1)
390      WRITE (LO, 624) (J=1,NM1)
400      WRITE (LO, 625) (J=1,NM1)
410      WRITE (LO, 626) (J=1,NM1)
420      WRITE (LO, 627) (J=1,NM1)
430      WRITE (LO, 628) (J=1,NM1)
440      WRITE (LO, 629) (J=1,NM1)
450      WRITE (LO, 630) (J=1,NM1)
460      WRITE (LO, 631) (J=1,NM1)
470      WRITE (LO, 632) (J=1,NM1)
480      WRITE (LO, 633) (J=1,NM1)
490      WRITE (LO, 634) (J=1,NM1)
500      WRITE (LO, 635) (J=1,NM1)
510      WRITE (LO, 636) (J=1,NM1)
520      WRITE (LO, 637) (J=1,NM1)
530      WRITE (LO, 638) (J=1,NM1)
540      WRITE (LO, 639) (J=1,NM1)
550      WRITE (LO, 640) (J=1,NM1)
560      WRITE (LO, 641) (J=1,NM1)
570      WRITE (LO, 642) (J=1,NM1)
580      WRITE (LO, 643) (J=1,NM1)
590      WRITE (LO, 644) (J=1,NM1)
600      WRITE (LO, 645) (J=1,NM1)
610      WRITE (LO, 646) (J=1,NM1)
620      WRITE (LO, 647) (J=1,NM1)
630      WRITE (LO, 648) (J=1,NM1)
640      WRITE (LO, 649) (J=1,NM1)
650      WRITE (LO, 650) (J=1,NM1)
660      WRITE (LO, 651) (J=1,NM1)
670      WRITE (LO, 652) (J=1,NM1)
680      WRITE (LO, 653) (J=1,NM1)
690      WRITE (LO, 654) (J=1,NM1)
700      WRITE (LO, 655) (J=1,NM1)
710      WRITE (LO, 656) (J=1,NM1)
720      WRITE (LO, 657) (J=1,NM1)
730      WRITE (LO, 658) (J=1,NM1)
740      WRITE (LO, 659) (J=1,NM1)
750      WRITE (LO, 660) (J=1,NM1)
760      WRITE (LO, 661) (J=1,NM1)
770      WRITE (LO, 662) (J=1,NM1)
780      WRITE (LO, 663) (J=1,NM1)
790      WRITE (LO, 664) (J=1,NM1)
800      WRITE (LO, 665) (J=1,NM1)
810      WRITE (LO, 666) (J=1,NM1)
820      WRITE (LO, 667) (J=1,NM1)
830      WRITE (LO, 668) (J=1,NM1)
840      WRITE (LO, 669) (J=1,NM1)
850      WRITE (LO, 670) (J=1,NM1)
860      WRITE (LO, 671) (J=1,NM1)
870      WRITE (LO, 672) (J=1,NM1)
880      WRITE (LO, 673) (J=1,NM1)
890      WRITE (LO, 674) (J=1,NM1)
900      WRITE (LO, 675) (J=1,NM1)
910      WRITE (LO, 676) (J=1,NM1)
920      WRITE (LO, 677) (J=1,NM1)
930      WRITE (LO, 678) (J=1,NM1)
940      WRITE (LO, 679) (J=1,NM1)
950      WRITE (LO, 680) (J=1,NM1)
960      WRITE (LO, 681) (J=1,NM1)
970      WRITE (LO, 682) (J=1,NM1)
980      WRITE (LO, 683) (J=1,NM1)
990      WRITE (LO, 684) (J=1,NM1)

```



SUBROUTINE YAT3

16/179 NPT=2 P4000

PTN 4.00000

20/12/20 10.0000

PAGE

```

200      DO 11 I=1,NPNTS2,1
201      11 10, 0001 11(P3(COEF,X1(I),I2),X2(I),I2),MOROR1,MOROR2),12=
202      11(PNTS2,1)
203      11 10, 0001 11(I2,I2=1,NPNTS2,1)
204      PRINT ERRORS, (P /ESTAT/ WAS 1 20 2.
205      22 IF (ESTAT.M0.1) GO TO 20
206      IF (ERROR.M0.1) GO TO 24
207      PRINT ABSOLUTE ERRORS.
208      WRITE (10, 011) (I2,I2=1,NPNTS2,1)
209      DO 23 I=1,NPNTS2,1
210      23 10, 0001 11(P3(COEF,X1(I),I2),X2(I),I2),MOROR1,MOROR2),12=
211      11(PNTS2,1)
212      10, 0001 11(I2,I2=1,NPNTS2,1)
213      PRINT RELATIVE ERRORS.
214      WRITE (10, 011) (I2,I2=1,NPNTS2,1)
215      DO 25 I=1,NPNTS2,1
216      25 10, 0001 11(P3(COEF,X1(I),I2),X2(I),I2),MOROR1,MOROR2),12=
217      11(PNTS2,1)
218      10, 0001 11(I2,I2=1,NPNTS2,1)
219      PRINT ERROR STATISTICS.
220      IF (ERROR.M0.1) WRITE (10, 011) ERRORS=MAX(JN0,ERR000)
221      IF (ERROR.M0.1) WRITE (10, 011) ERRORS=MAX(JN0,ERR000)
222      GO TO 20
223      FATAL ERROR DETECTED. OUTPUT ERROR MESSAGE AND PRESET COEFFICIENTS.
224      27 10, 0001 11(P3(COEF,X1(I),I2),X2(I),I2),MOROR1,MOROR2),12=
225      11(PNTS2,1)
226      10, 0001 11(I2,I2=1,NPNTS2,1)
227      10, 0001 11(P3(COEF,X1(I),I2),X2(I),I2),MOROR1,MOROR2),12=
228      11(PNTS2,1)
229      10, 0001 11(I2,I2=1,NPNTS2,1)
230      RETURN.
231      29 RETURN
232      END

```

SUBROUTINE MAT31

76/175 3PT=2 #400P

RTN 000000

007.000000

PAUL

```

1      SUBROUTINE MAT31 (NPNTS1,NPNTS2,M(IPMR1),SUBM1)
2      .....
3      PRESET SUBMATRIX 1, CLASS SUBM1 IN THE SYSTEM MATRIX A FOR A
4      3D-POLYNOMIAL APPROXIMATION.
5      .....
6      AUTHOR: JAMES LEANER
7      DATE: DECEMBER 31, 1980
8      .....
9      VARIABLE      EXPLANATION      TYPE
10     *-----* *-----* *-----*
11     * NPNTS1      NUMBER OF DATA POINTS AT CONSTANT 11      INTEGER
12     * NPNTS2      NUMBER OF DATA POINTS AT CONSTANT 12      INTEGER
13     * M           NUMBER OF COEFFICIENTS FOR THE POLY-      INTEGER
14     *             NOMIAL IN THE DIRECTION
15     * IPMR1       POINTS FOR THE 11 VALUES IN THE      INTEGER
16     *             SUBMATRIX
17     * SUBM1(M)    SUBMATRIX 1, CLASS IN THE SYSTEM      REAL
18     *-----* *-----* *-----*
19     REAL SUBM1(M)
20     DIMENSION M(1)
21     IF (M(1).EQ.0)
22     THEN
23     GO TO 01
24     IF (M(1).GT.0) GO TO 01
25     TOP EDGE SECTION ELEMENTS.
26     I=1
27     J=1
28     GO TO 02
29     RIGHT HAND EDGE SECTION ELEMENTS.
30     01 I=(M-1)+1
31     J=1
32     PRESET EDGE SECTION ELEMENTS IN SUBMATRIX 1, CLASS USING 53.
33     02 SUBM1(I,J)=53*(NPNTS1,NPNTS2,(IPMR1,IPMR2,0))
34     IF (NPNTS2-IPMR2).EQ.1
35     THEN
36     COPY DEFINED ELEMENTS DIAGONALLY.
37     M=M-1
38     DO 03 I=1,M+1
39     DO 03 J=1,M+1
40     SUBM1(I,J)=SUBM1(I,J)
41     RETURN
42     END

```



## INTEGER FUNCTION IEI3 (N)

GAUSS JORDAN ELIMINATION WITH MATRIX A AND VECTOR B. THE  
VARIABLE IEI3 IS SET TO 1 IF THE ELIMINATION IS COMPLETED.  
IEI3 LESS THAN 0 INDICATES AN ERROR.

AUTHOR: HANS ZERNER  
DATE: DECEMBER 11, 1960

VARIABLE	EXPLANATION	TYPE
N	NUMBER OF EQUATIONS	INTEGER
IEI3	CONTROL PARAMETER, RETURNED TO THE CALLING ROUTINE	INTEGER
	IF GAUSS JORDAN ELIMINATION WAS PERFORMED CORRECTLY, ERROR INDICATED DURING GAUSS JORDAN ELIMINATION.	

(149:49) SYSTEM MATRIX A REAL  
B(49) RIGHT HAND SIDE VECTOR B REAL

IMPORTANT: MATRIX A AND VECTOR B HAVE TO BE PROVIDED THROUGH  
THE LABELS COMMON BLOCK A93. BOTH A AND B WILL  
BE LARGER CHANGES DURING EXECUTION, I.E. THE  
ELIMINATION.

COMMON / A93 / A,B

REAL A(49:49),B(49)

```

10  IEI3=1
11  IEI3=IEI3
12  IF IEI3.EQ.0 GO TO 30
13  DO 10 I=1,N
14  DO 10 J=1,N
15  IF (A(I,J).EQ.0) GO TO 17
16  CONST=A(I,J)/A(I,I)
17  DO 10 J=1,N
18  A(I,J)=A(I,J)-CONST*A(I,I)
19  IF (J.EQ.1) A(I,J)=0
20  CONTINUE
21  A(I,I)=CONST*A(I,I)
22  CONTINUE
23  CONST=A(I,I)
24  DO 10 J=1,N
25  A(I,J)=A(I,J)/CONST
26  IF (J.EQ.1) A(I,J)=1
27  IF (J.EQ.1) GO TO 30
28  IF (A(I,J).GT.EPS) GO TO 24
29  IF (A(I,J).GT.EPS) GO TO 33
30  IF (A(I,J).GT.EPS) GO TO 37
31  CONTINUE
32  GO TO 30
33  IEI3=IEI3
34  RETURN
35  END

```

158

159

741175 3P1-2 24090

FTN 0040474

30146129 49,50,99

246

[illegible]

FUNCTION F3

7/6/75 1PT=2 P10NP

FIN 4-00-000

CL/12/4V 14-14-44

PAGE

```

1      REAL FUNCTION F3 (COEF,X1,X2,NORDR1,N3PCF2)
2      .....
3      COMPUTE FUNCTION VALUE Y=F(X1,X2) OF A 3D-POLYNOMIAL USING
4      THE NORMER SCHEME IN ORDER TO INCREASE SPEED AND ACCURACY.
5      .....
6      AUTHOR: JAMES J. JENSEN
7      DATE: OCTOBER 31, 1969
8      .....
9      VARIABLE      EXPLANATION      TYPE
10     COEFF1      3D-POLYNOMIAL COEFFICIENTS      REAL
11     X1          X1-VALUE, WHERE Y WILL BE COMPUTED.      REAL
12     X2          X2-VALUE, WHERE Y WILL BE COMPUTED.      REAL
13     NORDR1      ORDER OF THE POLYNOMIAL IN THE X1 DIRECTION.      INTEGER
14     NORDR2      ORDER OF THE POLYNOMIAL IN THE X2 DIRECTION.      INTEGER
15     F3          FUNCTION VALUE OF THE 3D-POLYNOMIAL.      REAL
16     .....
17     REAL COEF1(40),X1(40)
18     NORDR1=NORDR1
19     NORDR2=NORDR2
20     .....
21     COMPUTE COEFFICIENTS D BASED ON COEFFICIENTS COEF AND X2.
22     .....
23     DO 02 I=1,NCOEF1,1
24     IC=I+NCOEF2
25     Y=COEF1(I)
26     DO 01 J=1,NORDR2,1
27     JNCOEF2=J
28     IC=I+JNCOEF2
29     Y=Y+COEF1(IC)*X2**J
30     01 CONTINUE
31     02 CONTINUE
32     Y=Y
33     .....
34     COMPUTE Y BASED ON COEFFICIENTS D AND X1
35     .....
36     Y=COEF1(1)
37     DO 03 I=1,NORDR1,1
38     JNCOEF1=I
39     Y=Y+COEF1(JNCOEF1)*X1**I
40     03 CONTINUE
41     RETURN
42     END

```



BLANK DATA 484

74/173 JPT-2 77000

STN 4.00000

00/00/00 10/00/00

Page 4

1

BLANK DATA 484  
CROSSING / 484  
JPT 2100001.01001

BLOCK DATA DATA

74/175 JPT-2 PRONO

PTN 4.00000

06/12/29 10:17:11

PAGE

1

1

BLOCK DATA DATA  
FROM / DATA / 11:12:12-Y  
REAL 11(9.7.31.2210.7.51.4210.7.51.719.7.51)  
END

164





PROGRAM CEN06 74/175 J07-2 PH000

PTM 4.00000

00/00/00 10:00:00

Page

IF (IPL07) CALL ST006

200

STOP 0077



[illegible]







```

1 SUBROUTINE NAT41 (NPNTS1,NPNTS2,NPNTS3,N,M,IPOM1,SUB1,4,SUB2)
2 .....
3 PRESBY SUBMATRICES 1. CLASS 1.001 IN THE SYSTEM MATRIX A FOR A
4 40-DEGREE APPROXIMATION.
5
6 AUTHOR: JAMES E. JENSEN
7 DATE: DECEMBER 31, 1980
8
9 VARIABLE EXPLANATION TYPE
10 .....
11 NPNTS1 NUMBER OF DATA POINTS AT CONSTANT 1 INTEGER
12 NPNTS2 NUMBER OF DATA POINTS AT CONSTANT 2 INTEGER
13 NPNTS3 NUMBER OF DATA POINTS AT CONSTANT 3 INTEGER
14 M NUMBER OF INTERSECTIONS FOR THE POLY- INTEGER
15 N NUMBER OF CYCLES FOR THE POLY- INTEGER
16 IPOM1 POINT ON THE 11 VALUES IN THE INTEGER
17 SUB1 SUBMATRICES 1. CLASS IN THE SYSTEM REAL
18 SUB2 SUBMATRICES 2. CLASS IN THE SYSTEM REAL
19 .....
20 .....
21 .....
22 .....
23 .....
24 .....
25 .....
26 .....
27 .....
28 .....
29 .....
30 .....
31 .....
32 .....
33 .....
34 .....
35 .....
36 .....
37 .....
38 .....
39 .....
40 .....
41 .....
42 .....
43 .....
44 .....
45 .....
46 .....
47 .....
48 .....
49 .....
50 .....
51 .....
52 .....
53 .....
54 .....
55 .....
56 .....
57 .....
58 .....
59 .....
60 .....
61 .....
62 .....
63 .....
64 .....
65 .....
66 .....
67 .....
68 .....
69 .....
70 .....
71 .....
72 .....
73 .....
74 .....
75 .....
76 .....
77 .....
78 .....
79 .....
80 .....
81 .....
82 .....
83 .....
84 .....
85 .....
86 .....
87 .....
88 .....
89 .....
90 .....
91 .....
92 .....
93 .....
94 .....
95 .....
96 .....
97 .....
98 .....
99 .....
100 .....

```



174

```

1  INTEGER FUNCTION IEL4* (N)
2  .....
3  GAUSS JORDAN ELIMINATION WITH MATRIX A AND VECTOR B. THE
4  VARIABLE IEL4 IS SET TO 1 IF THE ELIMINATION IS COMPLETED.
5  IEL4 LESS THAN 0 INDICATES AN ERROR.
6  .....
7  AUTHOR: JAMES ZERNER
8  DATE: OCTOBER 11, 1968
9  .....
10 VARIABLE EXPLANATION TYPE
11 .....
12 N NUMBER OF EQUATIONS INTEGER
13 IEL4 CONTROL PARAMETER RETURNED TO THE CALLING ROUTINE INTEGER
14 .....
15 *A GAUSS JORDAN ELIMINATION WAS
16 *PERFORMED CORRECTLY.
17 *NO ERROR ENCOUNTERED DURING GAUSS
18 *JORDAN ELIMINATION.
19 .....
20 A(100,001) SYSTEM MATRIX A REAL
21 B(100,001) RIGHT HAND SIDE VECTOR B REAL
22 .....
23 IMPORTANT: MATRIX A AND VECTOR B HAVE TO BE PROVIDED THROUGH
24 THE LABELED COMMON BLOCK LAB. BOTH A AND B WILL
25 WILL UNDERGO CHANGES DURING EXECUTION. I.E. THE
26 ELIMINATION.
27 .....
28 COMMON / LAB / A,B
29 REAL A(100,001),B(100,001)
30 .....
31 IEL4=1
32 EPS=1.00E-030
33 DO 01 I=1,N
34 DO 02 J=1,N
35 IF (A(I,J).EQ.0) GO TO 02
36 CONST=-A(I,J)/A(I,I)
37 DO 01 J=1,N
38 A(I,J)=A(I,J)+CONST*A(I,I)
39 IF (I.EQ.1) A(I,J)=0
40 .....
41 01 CONTINUE
42 B(I)=B(I)+CONST*B(I)
43 02 CONTINUE
44 CONST=A(I,I)
45 DO 03 J=1,N
46 A(I,J)=A(I,J)/CONST
47 B(I)=B(I)/CONST
48 DO 02 J=1,N
49 .....
50 IF (ABS(A(I,I))-GT.EPS) GO TO 34
51 ERROR=ERROR+1
52 .....
53 04 CONTINUE
54 IF (ABS(B(I))-GT.EPS) GO TO 39
55 ERROR=ERROR+1
56 IF (I.EQ.N) GO TO 37
57 05 CONTINUE
58 GO TO 05
59 IEL4=1
60 RETURN
61 END

```

176

292

177



```

      R=SQRT(E)
      IF (R.LT.EPSILON) GO TO 28
      IF (1/2*E*ITERMAX) GO TO 33
      24 R=1/2*E*ITERMAX
      PRINT, SYSTEM MATRIX A AND RIGHT HAND SIDE VECTOR B AFTER THE
      PIVOTIZATION.
      100 IF (PRINT) WRITE (L3, 501) L3
      101 IF (PRINT) WRITE (L3, 502) (J,J=1,N+1)
      25 IF (PRINT) WRITE (L3, 503) (J,J=1,N+1)
      105 IF (PRINT) WRITE (L3, 504) (J,J=1,N+1)
      26 V(J)=A(J,I)
      STOP LOOP TO FIND SOLUTION.
      100 27 CONTINUE
      REARRANGE SOLUTION IN THE ARRAY V.
      105 28 CONTINUE
      29 V(I)=B(I)
      PRINT SOLUTION, IF PRINT IS SET .TRUE. .
      200 30 IF (PRINT) WRITE (L3, 505) I
      31 IF (PRINT) WRITE (L3, 506) (J,J=1,N+1)
      205 32 IF (PRINT) WRITE (L3, 507) (J,J=1,N+1)
      210 33 IF (PRINT) WRITE (L3, 508) (J,J=1,N+1)
      RETURN
      215 34 ERROR ENCOUNTERED. STOP.
      220 35 IF (ITER=1) GO TO 100
      36 IF (1/2*E*ITERMAX) GO TO 33
      RETURN
      225 37 NO CONVERGENCE OF THE SOLUTION.
      38 IF (ITER=ITERMAX) GO TO 33
      RETURN
      END

```

```

1      REAL FUNCTION F4 (CJEP,X1,X2,X3,NOROR1,NOROR2,NOROR3)
2      .....
3      COMPUTE FUNCTION VALUE F=P(X1,X2,X3) OF A 40-POLYNOMIAL USING
4      THE HARNER SCHEME IN ORDER TO INCREASE SPEED AND ACCURACY.
5      .....
6      AUTHOR:  NAME ZERNER
7      DATE:    OCTOBER 11, 1980
8      .....
9      VARIABLE EXPLANATION                                TYPE
10     CJEP(10)  ARGUMENTS: 10 COEFFICIENTS OF THE 40-POLYNOMIAL.  REAL
11     X1         X1-VALUE: UNDER Y WILL BE COMPUTED.              REAL
12     X2         X2-VALUE: UNDER Y WILL BE COMPUTED.              REAL
13     X3         X3-VALUE: UNDER Y WILL BE COMPUTED.              REAL
14     NOROR1     ORDER OF THE POLYNOMIAL IN THE X1 DIRECTION.      INTEGER
15     NOROR2     ORDER OF THE POLYNOMIAL IN THE X2 DIRECTION.      INTEGER
16     NOROR3     ORDER OF THE POLYNOMIAL IN THE X3 DIRECTION.      INTEGER
17     F4         FUNCTION VALUE OF THE 40-POLYNOMIAL.              REAL
18     .....
19     REAL CJEP(10),X1,X2,X3,NOROR1,NOROR2,NOROR3
20     NOROR1=NOROR1+1
21     NOROR2=NOROR2+1
22     NOROR3=NOROR3+1
23     .....
24     COMPUTE COEFFICIENTS D BASED ON COEFFICIENTS COEF AND X3
25     .....
26     DO 10 I=1,NOROR1+1
27     10  D(I)=1
28     DO 20 J=1,NOROR2+1
29     20  D(J)=1
30     DO 30 K=1,NOROR3+1
31     30  D(K)=1
32     DO 40 L=1,NOROR4+1
33     40  D(L)=1
34     DO 50 M=1,NOROR5+1
35     50  D(M)=1
36     DO 60 N=1,NOROR6+1
37     60  D(N)=1
38     DO 70 O=1,NOROR7+1
39     70  D(O)=1
40     DO 80 P=1,NOROR8+1
41     80  D(P)=1
42     DO 90 Q=1,NOROR9+1
43     90  D(Q)=1
44     DO 100 R=1,NOROR10+1
45     100 D(R)=1
46     DO 110 S=1,NOROR11+1
47     110 D(S)=1
48     DO 120 T=1,NOROR12+1
49     120 D(T)=1
50     DO 130 U=1,NOROR13+1
51     130 D(U)=1
52     DO 140 V=1,NOROR14+1
53     140 D(V)=1
54     DO 150 W=1,NOROR15+1
55     150 D(W)=1
56     DO 160 X=1,NOROR16+1
57     160 D(X)=1
58     DO 170 Y=1,NOROR17+1
59     170 D(Y)=1
60     DO 180 Z=1,NOROR18+1
61     180 D(Z)=1
62     DO 190 AA=1,NOROR19+1
63     190 D(AA)=1
64     DO 200 AB=1,NOROR20+1
65     200 D(AB)=1
66     DO 210 AC=1,NOROR21+1
67     210 D(AC)=1
68     DO 220 AD=1,NOROR22+1
69     220 D(AD)=1
70     DO 230 AE=1,NOROR23+1
71     230 D(AE)=1
72     DO 240 AF=1,NOROR24+1
73     240 D(AF)=1
74     DO 250 AG=1,NOROR25+1
75     250 D(AG)=1
76     DO 260 AH=1,NOROR26+1
77     260 D(AH)=1
78     DO 270 AI=1,NOROR27+1
79     270 D(AI)=1
80     DO 280 AJ=1,NOROR28+1
81     280 D(AJ)=1
82     DO 290 AK=1,NOROR29+1
83     290 D(AK)=1
84     DO 300 AL=1,NOROR30+1
85     300 D(AL)=1
86     DO 310 AM=1,NOROR31+1
87     310 D(AM)=1
88     DO 320 AN=1,NOROR32+1
89     320 D(AN)=1
90     DO 330 AO=1,NOROR33+1
91     330 D(AO)=1
92     DO 340 AP=1,NOROR34+1
93     340 D(AP)=1
94     DO 350 AQ=1,NOROR35+1
95     350 D(AQ)=1
96     DO 360 AR=1,NOROR36+1
97     360 D(AR)=1
98     DO 370 AS=1,NOROR37+1
99     370 D(AS)=1
100    DO 380 AT=1,NOROR38+1
101    380 D(AT)=1
102    DO 390 AU=1,NOROR39+1
103    390 D(AU)=1
104    DO 400 AV=1,NOROR40+1
105    400 D(AV)=1
106    .....
107    COMPUTE COEFFICIENTS E BASED ON COEFFICIENTS D AND X2.
108    .....
109    DO 410 I=1,NOROR1+1
110    410  E(I)=1
111    DO 420 J=1,NOROR2+1
112    420  E(J)=1
113    DO 430 K=1,NOROR3+1
114    430  E(K)=1
115    DO 440 L=1,NOROR4+1
116    440  E(L)=1
117    DO 450 M=1,NOROR5+1
118    450  E(M)=1
119    DO 460 N=1,NOROR6+1
120    460  E(N)=1
121    DO 470 O=1,NOROR7+1
122    470  E(O)=1
123    DO 480 P=1,NOROR8+1
124    480  E(P)=1
125    DO 490 Q=1,NOROR9+1
126    490  E(Q)=1
127    DO 500 R=1,NOROR10+1
128    500  E(R)=1
129    DO 510 S=1,NOROR11+1
130    510  E(S)=1
131    DO 520 T=1,NOROR12+1
132    520  E(T)=1
133    DO 530 U=1,NOROR13+1
134    530  E(U)=1
135    DO 540 V=1,NOROR14+1
136    540  E(V)=1
137    DO 550 W=1,NOROR15+1
138    550  E(W)=1
139    DO 560 X=1,NOROR16+1
140    560  E(X)=1
141    DO 570 Y=1,NOROR17+1
142    570  E(Y)=1
143    DO 580 Z=1,NOROR18+1
144    580  E(Z)=1
145    DO 590 AA=1,NOROR19+1
146    590  E(AA)=1
147    DO 600 AB=1,NOROR20+1
148    600  E(AB)=1
149    DO 610 AC=1,NOROR21+1
150    610  E(AC)=1
151    DO 620 AD=1,NOROR22+1
152    620  E(AD)=1
153    DO 630 AE=1,NOROR23+1
154    630  E(AE)=1
155    DO 640 AF=1,NOROR24+1
156    640  E(AF)=1
157    DO 650 AG=1,NOROR25+1
158    650  E(AG)=1
159    DO 660 AH=1,NOROR26+1
160    660  E(AH)=1
161    DO 670 AI=1,NOROR27+1
162    670  E(AI)=1
163    DO 680 AJ=1,NOROR28+1
164    680  E(AJ)=1
165    DO 690 AK=1,NOROR29+1
166    690  E(AK)=1
167    DO 700 AL=1,NOROR30+1
168    700  E(AL)=1
169    DO 710 AM=1,NOROR31+1
170    710  E(AM)=1
171    DO 720 AN=1,NOROR32+1
172    720  E(AN)=1
173    DO 730 AO=1,NOROR33+1
174    730  E(AO)=1
175    DO 740 AP=1,NOROR34+1
176    740  E(AP)=1
177    DO 750 AQ=1,NOROR35+1
178    750  E(AQ)=1
179    DO 760 AR=1,NOROR36+1
180    760  E(AR)=1
181    DO 770 AS=1,NOROR37+1
182    770  E(AS)=1
183    DO 780 AT=1,NOROR38+1
184    780  E(AT)=1
185    DO 790 AU=1,NOROR39+1
186    790  E(AU)=1
187    DO 800 AV=1,NOROR40+1
188    800  E(AV)=1
189    .....
190    COMPUTE Y BASED IN COEFFICIENTS E AND X1.
191    .....
192    DO 810 I=1,NOROR1+1
193    810  Y(I)=1
194    DO 820 J=1,NOROR2+1
195    820  Y(J)=1
196    DO 830 K=1,NOROR3+1
197    830  Y(K)=1
198    DO 840 L=1,NOROR4+1
199    840  Y(L)=1
200    DO 850 M=1,NOROR5+1
201    850  Y(M)=1
202    DO 860 N=1,NOROR6+1
203    860  Y(N)=1
204    DO 870 O=1,NOROR7+1
205    870  Y(O)=1
206    DO 880 P=1,NOROR8+1
207    880  Y(P)=1
208    DO 890 Q=1,NOROR9+1
209    890  Y(Q)=1
210    DO 900 R=1,NOROR10+1
211    900  Y(R)=1
212    DO 910 S=1,NOROR11+1
213    910  Y(S)=1
214    DO 920 T=1,NOROR12+1
215    920  Y(T)=1
216    DO 930 U=1,NOROR13+1
217    930  Y(U)=1
218    DO 940 V=1,NOROR14+1
219    940  Y(V)=1
220    DO 950 W=1,NOROR15+1
221    950  Y(W)=1
222    DO 960 X=1,NOROR16+1
223    960  Y(X)=1
224    DO 970 Y=1,NOROR17+1
225    970  Y(Y)=1
226    DO 980 Z=1,NOROR18+1
227    980  Y(Z)=1
228    DO 990 AA=1,NOROR19+1
229    990  Y(AA)=1
230    DO 1000 AB=1,NOROR20+1
231    1000 Y(AB)=1
232    DO 1010 AC=1,NOROR21+1
233    1010 Y(AC)=1
234    DO 1020 AD=1,NOROR22+1
235    1020 Y(AD)=1
236    DO 1030 AE=1,NOROR23+1
237    1030 Y(AE)=1
238    DO 1040 AF=1,NOROR24+1
239    1040 Y(AF)=1
240    DO 1050 AG=1,NOROR25+1
241    1050 Y(AG)=1
242    DO 1060 AH=1,NOROR26+1
243    1060 Y(AH)=1
244    DO 1070 AI=1,NOROR27+1
245    1070 Y(AI)=1
246    DO 1080 AJ=1,NOROR28+1
247    1080 Y(AJ)=1
248    DO 1090 AK=1,NOROR29+1
249    1090 Y(AK)=1
250    DO 1100 AL=1,NOROR30+1
251    1100 Y(AL)=1
252    DO 1110 AM=1,NOROR31+1
253    1110 Y(AM)=1
254    DO 1120 AN=1,NOROR32+1
255    1120 Y(AN)=1
256    DO 1130 AO=1,NOROR33+1
257    1130 Y(AO)=1
258    DO 1140 AP=1,NOROR34+1
259    1140 Y(AP)=1
260    DO 1150 AQ=1,NOROR35+1
261    1150 Y(AQ)=1
262    DO 1160 AR=1,NOROR36+1
263    1160 Y(AR)=1
264    DO 1170 AS=1,NOROR37+1
265    1170 Y(AS)=1
266    DO 1180 AT=1,NOROR38+1
267    1180 Y(AT)=1
268    DO 1190 AU=1,NOROR39+1
269    1190 Y(AU)=1
270    DO 1200 AV=1,NOROR40+1
271    1200 Y(AV)=1
272    .....
273    RETURN
274    END

```

# DISTRIBUTION LIST

	<u>No. of Copies</u>
1. Defense Documentation Center Cameron Station Alexandria, Virginia 22314	1
2. Library Code 0212 Naval Postgraduate School Monterey, California 93940	2
3. Office of Research Administration Code 012A Naval Postgraduate School Monterey, California 93940	1
4. Chairman Code 67 Department of Aeronautics Naval Postgraduate School Monterey, California 93940	1
5. Director, Turbopropulsion Laboratory Department of Aeronautics Naval Postgraduate School Monterey, California 93940	30
6. Dr. Gerhard Heiche Naval Air Systems Command Code AIR-310 Navy Department Washington, D.C. 20360	1
7. Mr. Karl H. Guttman Naval Air Systems Command Code AIR-330 Navy Department Washington, D.C. 20360	1
8. Mr. James R. Patton, Jr. Power Program, Code 473 Office of Naval Research Arlington, Virginia 22218	1
9. Dr. A. D. Wood Office of Naval Research Eastern/ Central Regional Office 666 Summer Street Boston, Massachusetts 02210	1

10. Commanding Officer 1  
Naval Air Propulsion Test Center  
Attn: Mr. Vernon Lubosky  
Trenton, New Jersey 08628
11. National Aeronautics & Space Administration 1  
Lewis Research Center (Library)  
2100 Brookpark Road  
Cleveland, Ohio 44135
12. CAG Library 1  
The Boeing Company  
Seattle, Washington 98124
13. Library 1  
General Electric Company  
Aircraft Engine Technology Division  
DTO Mail Drop H43  
Cincinnati, Ohio 45215
14. Library 1  
Pratt and Whitney Aircraft  
Post Office Box 2691  
West Palm Beach, Florida 33402
15. Library 1  
Pratt and Whitney Aircraft  
East Hartford, Connecticut 06108
16. Chief, Fan and Compressor Branch 1  
Mail Stop 5-9  
NASA Lewis Research Center  
2100 Brookpark Road  
Cleveland, Ohio 44135
17. Prof. D. Adler 1  
Technion Israel Institute of Technology  
Department of Mechanical Engineering  
Haifa 32000  
Israel
18. Director, Whittle Laboratory 1  
Department of Engineering  
Cambridge University  
England

19. Prof. F. A. E. Breugelmans 1  
 Institut von Karman de la Dynamique  
 des Fluides  
 72 Chaussee de Waterloo  
 1640 Rhode-St. Genese  
 Belgium
  
20. Library 1  
 Air Research Mfg. Corporation  
 Division of Garrett Corporation  
 402 South 36th Street  
 Phoenix, Arizona 85034
  
21. Dr. Robert P. Dring 1  
 United Technologies Research Labs  
 400 Main Street  
 Hartford, Connecticut 06108
  
22. Mr. Jean Fabri 1  
 ONERA  
 29, Ave. de la Division Leclerc  
 92 Chatillon  
 France
  
23. Prof. Dr. Ing Heinz E. Gallus 1  
 Lehrstuhl und Institut fur Strahlantriebe  
 und Turboarbeitsmaschinen  
 Rhein.-Westf. Techn. Hochschule Aachen  
 Templergraben 55  
 5100 Aachen, Germany
  
24. Mr. R. A. Langworthy 1  
 Army Aviation Material Laboratories  
 Department of the Army  
 Fort Eustis, Virginia 23604
  
25. Dr. W. Schlachter 1  
 Brown, Boveri & Company Ltd.  
 Department T-T  
 Haselstrasse  
 CH-5401 Baden, Switzerland
  
26. Prof. T. H. Okiishi 1  
 Professor of Mechanical Engineering  
 208 Mechanical Engineering Building  
 Iowa State University  
 Ames, Iowa 50011

27. Dr. Leroy H. Smith, Jr. 1  
Manager, Compressor & Fan Technology Operation  
General Electric Company  
Aircraft Engine Technology Division  
DTO Mail Drop H43  
Cincinnati, Ohio 45215
28. Dr. Arthur J. Wannerstrom 1  
ARL/LF  
Wright-Patterson AFB  
Dayton, Ohio 45433
29. Dr. H. Weyer 1  
DFVLR  
Linder Hohe  
505 Porz-Wahn  
Germany
30. Mr. P. F. Yaggy 1  
Director  
U.S. Army Aeronautical Research Laboratory  
AMES Research Center  
Moffett Field, California 94035
31. Prof. C. H. Wu 1  
P.O. Box 2706  
Beijing 100080  
China
32. Director 1  
Gas Turbine Establishment  
P.O. Box 305  
Jiangyou County  
Sichuan Province  
China

END

DATE  
FILMED

3-82

DTIC